

# **Product Description Version 15**

Ingenieurgesellschaft für  
technische Software mbH



© **INTES GmbH**, September 2014 (rev. 15.02)

The words in **green** colour are collected in the index at the end of this document.

The finite element model of a tractor transmission on the frontpage appears by courtesy of ZF Friedrichshafen AG in Friedrichshafen, Germany.

Address: **INTES GmbH**  
**Schulze-Delitzsch-Str. 16**  
**D-70565 Stuttgart**

Phone: **+49 711 784 99 - 0**  
 Fax: **+49 711 784 99 - 10**

E-mail: **[info@intes.de](mailto:info@intes.de)**  
 WWW: **<http://www.intes.de>**

Abaqus is a registered trademark of Dassault Systèmes Simulia Corp., Providence, RI, USA.

Adams is a registered trademark of the MSC.Software Corporation, Santa Ana, USA.

CATIA is a registered trademark of of DASSAULT SYSTEMS, and in Canada, IBM is the registered user under No. RU 81167.

COMREL is a registered trademark of the RCP GmbH, München, Germany .

DADS is a registered trademark of LMS CADSI, Coralville, USA.

Excite is a registered trademark of AVL List, Graz, Austria.

HyperMesh, HyperView, H3D and MotionSolve are registered trademarks of the Altair Engineering Inc., Big Beaver, USA.

I-DEAS is a registered trademark of SIEMENS PLM Software Inc., Plano, USA.

MEDINA and CAE-Datenschienen are registered trademarks of the T-Systems ITS GmbH, Stuttgart, Germany .

MATLAB is a registered trademark of The Mathworks Inc., Natick, MA, USA.

MpCCI is a registered trademark of FhG SCAI, St. Augustin, Germany .

NASTRAN is a registered trademark of the National Aeronautics and Space Administration (NASA).

MSC.Patran is a registered trademark of the MSC.Software Corporation, Santa Ana, USA.

**PERMAS** is a registered trademark of **INTES** Ingenieurgesellschaft für technische Software mbH, Stuttgart, Germany .

SIMPACT is a registered trademark of SIMPACK AG, Gilching, Germany.

STAR CD is a registered trademark of Computational Dynamics Ltd., London, England.

TOSCA is a registered trademark of FE-DESIGN GmbH, Karlsruhe, Germany .

VAO is a registered trademark of CDH AG, Ingolstadt, Germany .

Virtual.Lab is a registered trademark of LMS International, Leuven, Belgium .

**VisPER** is a registered trademark of **INTES** Ingenieurgesellschaft für technische Software mbH, Stuttgart, Germany .

The use of registered names or trademarks does not imply, even in the absence of further specific statements, that such names are free for general use.

## Contents

	Page		
<b>INTES</b>	<b>5</b>	Error estimator	56
Company Profile	5	Material Properties	57
Services	5	Sets	57
<b>PERMAS</b>	<b>7</b>	Mathematical Functions	58
Overview	7	Loads	58
Introduction to PERMAS	7	Model Verification	58
Benefits of PERMAS	8	Interfaces	60
What's New in PERMAS Version 15	9	Matrix Models	61
What's New in VisPER Version 4	14	Combination of Results	62
Universal Features	15	Transformation of Results	62
Available VisPER Modules	16	Comparison of Results	62
Available PERMAS Modules	16	XY Result Data	62
Performance Aspects	17	Cutting Forces	63
Parallelization	17	Restarts	63
Areas of Application	18	Open Software System	63
Reliability	18	Direct Coupled Analyses	63
Quality Assurance	19	Coupling with CFD	64
<b>Applications</b>	<b>21</b>	<b>PERMAS Analysis Modules</b>	<b>65</b>
Car Body Analysis	21	PERMAS-MQA – Model Quality Assurance	65
Engine Analysis	24	PERMAS-LS – Linear Statics	65
Part Connections	26	PERMAS-CA – Contact Analysis	66
Brake Squeal Analysis	28	PERMAS-CAX – Extended Contact Analysis	69
Rotating Systems	30	PERMAS-CAU – Contact Geometry Update	69
Analysis of Machine Tools	31	PERMAS-NLS – Nonlinear Statics	70
Actively Controlled Systems	35	PERMAS-NLSMAT – Extended Material Laws	72
Robust Optimum Design	36	PERMAS-BA – Linear Buckling	73
<b>VisPER</b>	<b>39</b>	PERMAS-DEV – Dynamic Eigenvalues	73
The VisPER History	39	PERMAS-DEVX – Extended Mode Analysis	74
VisPER – A Short Introduction	39	PERMAS-MLDR – Eigenmodes with MLDR	76
VisPER-BAS – Basic Module	40	PERMAS-DRA – Dynamic Response	76
VisPER-TOP – Topology Optimization	42	PERMAS-DRX – Extended Dynamics	79
VisPER-OPT – Design Optimization	42	PERMAS-FS – Fluid-Structure Acoustics	79
VisPER-FS – Fluid-Structure Coupling	44	PERMAS-NLD – Nonlinear Dynamics	81
VisPER-CA – Contact Analysis	46	PERMAS-HT – Heat Transfer	81
Substructuring	47	PERMAS-NLHT – Nonlinear Heat Transfer	82
Evaluation of Spotwelds	48	PERMAS-OPT – Design Optimization	83
<b>PERMAS Basic Functions</b>	<b>49</b>	PERMAS-TOPO – Layout Optimization	86
Substructuring	49	PERMAS-AOS – Advanced Optim. Solvers	88
Submodeling	49	PERMAS-RA – Reliability Analysis	89
Variant Analysis	50	PERMAS-EMS – Electro- and Magneto-Statics	91
Cyclic Symmetry	51	PERMAS-EMD – Electrodynamics	91
Surface and Line Description	51	<b>PERMAS Special Modules</b>	<b>93</b>
Automated Coupling of Parts	52	PERMAS-LA – Laminate Analysis	93
Automated Spotweld Modeling	53	PERMAS-WLDS – Refined Weldspot Model	93
Kinematic Constraints	53	PERMAS-XPU – GPU accelerator	93
Handling of Singularities	55	<b>Interfaces</b>	<b>95</b>
Element Library	55	PERMAS-MEDI – MEDINA Door	95
Standard Beam Cross Sections	56	PERMAS-PAT – PATRAN Door	95
Design Elements for Optimization	56	PERMAS-ID – I-DEAS Door	95
		PERMAS-AD – ADAMS Interface	96
		PERMAS-DADS – DADS Interface	96
		PERMAS-EXCI – EXCITE Interface	96

PERMAS-SIM – SIMPACK Interface	96
PERMAS-HMS – MotionSolve Interface	97
PERMAS-H3D – HYPERVIEW Interface	97
PERMAS-VAO – VAO Interface	97
PERMAS-VLAB – Virtual.Lab Interface	97
PERMAS-MAT – MATLAB Interface	97
PERMAS-NAS – NASTRAN Door	97
PERMAS-ABA – ABAQUS-Door	98
PERMAS-CCL – MpCCI Coupling	99
More Interfaces	99
<b>Installation and beyond</b>	<b>101</b>
Supported Hardware Platforms	101
Licensing	101
Maintenance and Porting	101
User Support	102
MEDINA + PERMAS	102
Additional Tools	102
Documentation	103
Training	103
Future Developments	103
Additional Information	104
<b>Index</b>	<b>105</b>

---



## INTES



Figure 1: INTES Head Office in Stuttgart

### Company Profile

INTES company was founded as an FE technology enterprise in 1984. Its competence in every aspect of Finite Element (FE) technology is provided by INTES to its clients not only through the high-end software system PERMAS. The full range of development know-how of INTES is also made available to its clients by the provision of top-notch services and expert consultancy. INTES activities mainly concentrate on the

- development and distribution of the FE solver PERMAS and the joint pre- and post-processor VisPER,
- development of new and efficient numerical and graphical methods,
- development of software for new hardware architectures (such as parallel computers),
- coupling of PERMAS and VisPER with other software systems (such as pre- and postprocessors and MBS systems),
- consultancy and training of users,
- consulting by analysis projects.

The international support of PERMAS clients is supported in France by INTES France and in Japan by INTES Japan. In addition, partners are supporting and distributing the software in other countries.

For all of its customers, INTES wants to be a competent partner in all respects regarding the Finite Element Method. Above all, satisfaction of the customers with all the software and services is of prime importance to the company.

### Services

INTES offers a number of services to its customers including:

- Developments for PERMAS and VisPER:
  - Interfaces to other software packages,
  - New modeling processes,
  - New analysis capabilities,
  - New finite elements,
  - Customer specific developments.
- Installation of PERMAS and VisPER on new hardware platforms as well as consultancy concerning the optimum hardware configuration,
- Software maintenance,
- FEM training,
- FEM research and development,
- Configuration and installation of add-on software products,
- Engineering:
  - modeling with VisPER, MEDINA,
  - simulation with PERMAS,
- Introduction of FE analysis in enterprises, continuous consultation service (hotline), and support on current projects.

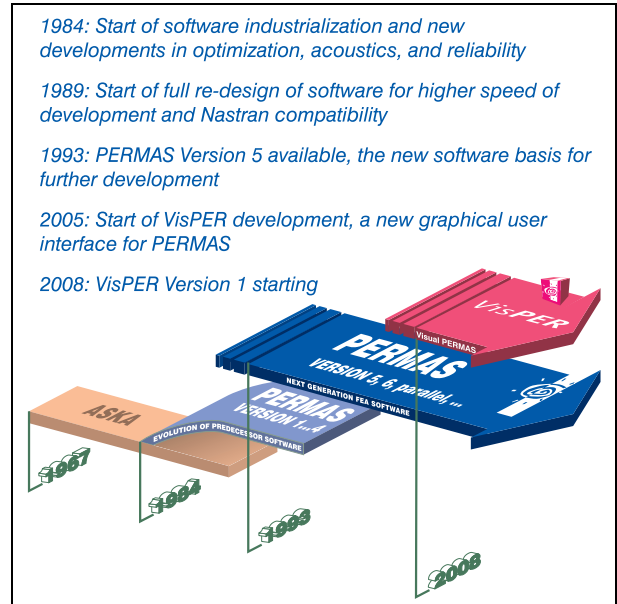


Figure 2: Synopsis of PERMAS history



## PERMAS

### Overview

This product description provides information on all essential characteristics of PERMAS and its application. Therefore, the description is organized into seven parts set forth below:

- The **introduction** gives some good reasons for the application of the Finite-Element-Method (FEM) and PERMAS. The particular **benefits** of PERMAS are presented on pages 8 to 19.
- **Applications** using several functional modules are illustrated on pages 21 to 37
- The features of **VisPER** are described on pages 39 to 48.
- The **universal features** of PERMAS, which are not related to a single module, are explained on pages 50 to 64.
- The available **functional modules** are described on pages 65 to 91.
- The **interfaces** are collected on pages 95 to 99.
- Additional information about the **installation and further aspects** of PERMAS is given on pages 101 to 104.

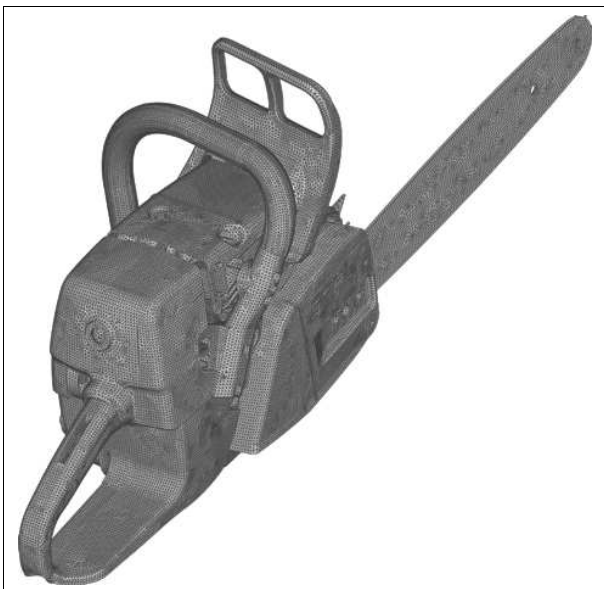


Figure 3: Model of a chain saw  
Andreas Stihl AG & Co. KG, Waiblingen, Germany.

### Introduction to PERMAS

**PERMAS** is a general purpose software system to perform complex calculations in engineering using the finite element method (FEM), and to optimize the analyzed structures and models. It has been developed by INTES and is available to engineers as an analysis tool worldwide.

**PERMAS** enables the engineer to perform comprehensive analyses and simulations in many fields of applications like stiffness analysis, stress analysis, determination of natural modes, dynamic simulations in the time and frequency domain, determination of temperature fields, acoustic fields, and electromagnetic fields, analysis of anisotropic material like fibre-reinforced composites.

**PERMAS** computes a large number of results during the course of these analyses, which may be used in the assessment of the structural behavior like deflections, stresses and strains, natural frequencies and mode shapes, strain energy distribution, sound vibration power density, time history and interaction with other parts of the structure.

Independent of the area of application, these results provide a lot of valuable information for the design and development process. A number of essential **benefits** can be derived from the early use of the FEM:

- Safe accomplishment of customer requirements.
- Reduction of expensive manufacturing and testing of prototypes.
- Simulation of extreme conditions.
- Shorter development and design cycles.
- Significant suggestions for design optimization:
  - topology optimization,
  - sizing optimization,
  - shape optimization,
  - parameter studies by sampling.
- Improvement of structural reliability.
- Analysis in case of malfunction of a structure during operation.
- Long term quality improvements.

In view of today's increasing requirements for short design cycles and high quality products, the finite element analysis becomes an indispensable tool for the daily development work. Moreover, complex products are often developed in distributed struc-

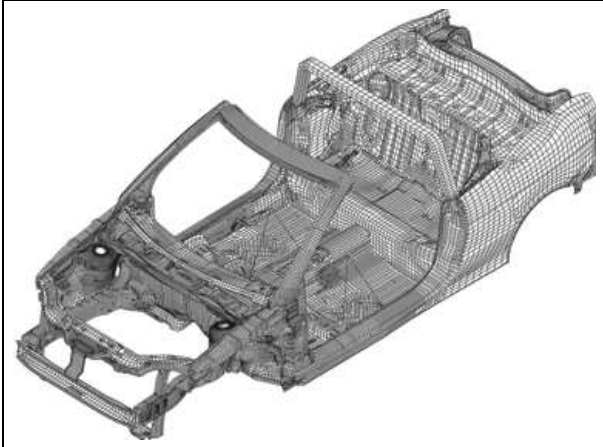


Figure 4: Car body model  
Wilhelm Karmann GmbH, Osnabrück, Germany.

tured companies. This makes interdependencies between different components of the product visible in time only if they are simulated and analysed on the computer. At the same time, the quality assurance of analysis results is of great importance. Hence, the choice of the right analysis tool is of crucial significance.

## Benefits of PERMAS

PERMAS is an internationally established FE analysis system with users in many countries. INTES has developed the system and, additionally, offers **individual consultation and user support** and all training required. The consultation covers all requests regarding the use of the software but also basic questions regarding the idealization and physical modeling.

The benefits arising from the use of PERMAS can be characterized by the following points:

- As a general purpose software package PERMAS provides for **powerful capabilities**, which cover a wide range of applications from mechanics to heat transfer, fluid structure acoustics and electrodynamics.
- Integrated optimization algorithms allow PERMAS not only to analyze models but also to determine optimized parts which fulfil many different conditions. The optimization methods include topology optimization, sizing and shape optimization, and reliability analysis to take into

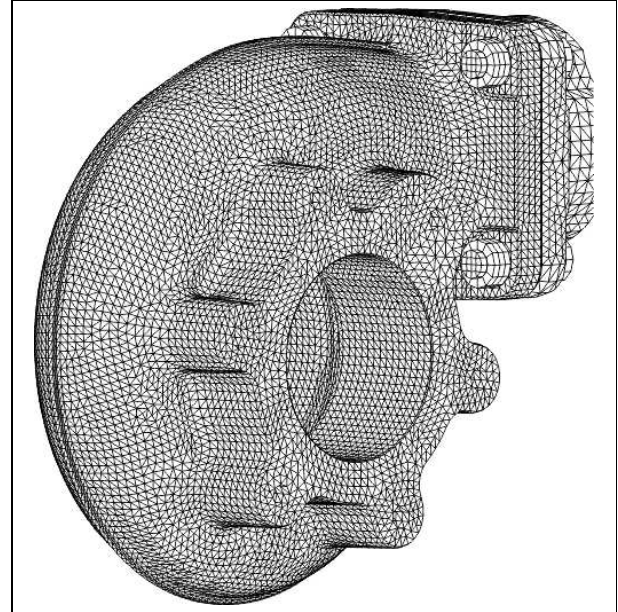


Figure 5: Turbocharger housing  
BorgWarner Turbo Systems Engineering GmbH,  
Kirchheimbolanden, Germany.

account uncertain model parameters.

- The graphical user interface VisPER supports the user in verifying his models and in evaluation of the analysis results. Moreover, VisPER provides advanced modeling features, e.g. for generation of fluid meshes, and for the set-up of optimization models in particular.
- Efficient equation solvers and optimized data storage schemes provide PERMAS with **ultimate computing power** with low resource consumption. Moreover, the software is continually adapted to the most advanced and powerful computers.
- PERMAS, a well-proven and mature software, has been available for many years and in numerous structural analysis departments. There, the **reliability of the software** is appreciated above all.

On the subsequent pages all these points are specified in more detail.

PERMAS is an advanced software package with up-to-date user conveniences. The PERMAS development aims to implement future-oriented functionalities in close cooperation with the users and to provide currently most advanced algorithms. In this way, PERMAS today faces the requirements of tomorrow.



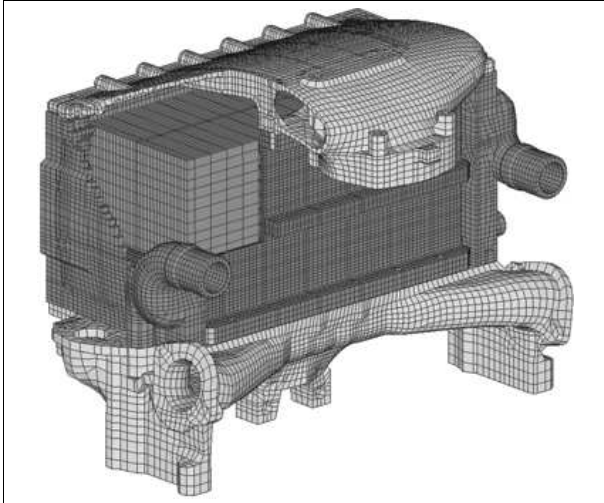


Figure 6: Charge air cooler  
Behr GmbH & Co., Stuttgart, Germany.

## What's New in PERMAS Version 15

The new Version 15 of PERMAS is the result of about 24 months of development work since the shipment of the predecessor version 14. For the regular reader of our Product Description of PERMAS, a rough overview summarizes the main changes in the new version. Of course, a complete and detailed Software Release Note is available with Version 15 in addition.

Great effort has been spent in the past years to provide *VisPER* (i.e. Visual PERMAS) as a dedicated tool to improve pre- and post-processing for special PERMAS functions. The fourth regular VisPER Version 4 is released at the same time as PERMAS Version 15. The new features of VisPER Version 4 will be introduced in the next section (see page 14).

PERMAS Version 15 offers again improved computing performance:

- The reordering process has been improved and parallelized which reduces the run time for large models significantly.
- Improved algorithm for direct FS coupled frequency response analysis (see Fig. 8).
- Performance improvements are made for contact analysis in case of friction (Module CAX). A better convergence in frictional force iterations can lead to a significant run time reduction dependent on the model characteristics.

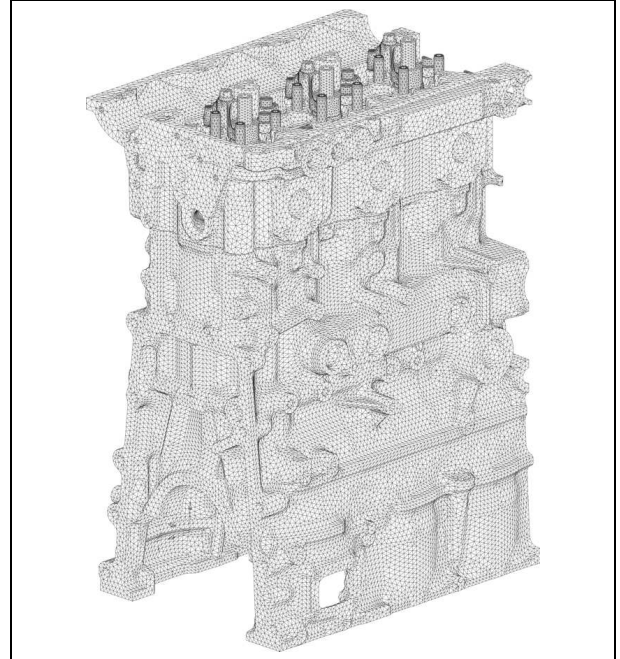


Figure 7: Half model of a six-cylinder Diesel engine  
Daimler Trucks North America in Detroit, Michigan, USA.

- Further acceleration of compute-bound analyses by utilizing a GPU of Nvidia (Tesla K20c or better) in addition to standard SMP parallelization (see Fig. 8 and below paragraph on new module XPU).

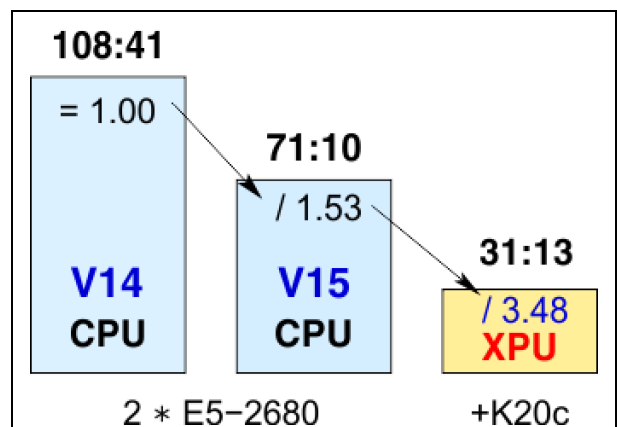


Figure 8: Direct frequency response  
for coupled fluid-structure model with 11.2 Million DOF  
on 16 cores plus Tesla K20c

The list of major software extensions in PERMAS is as follows:

- **New modules:**
  - The new module XPU (see module XPU on page 93) supports Nvidia GPU (Tesla K20c or better with CUDA Version 6) for acceleration of compute-bound analyses by a seamless

integration in PERMAS parallelization concept. The functionality is offered together with the S64 versions of PERMAS (see the list of PERMAS operating modes on page 101), since an efficient use of the GPU requires a sufficient core memory of the host machine ( $\geq 32$  GB). The acceleration will be most beneficial for compute-intensive analyses, such as eigenvalue analysis with a high number of modes, a direct fluid-structure coupled response analysis, or large normal contact problems.

- **Major extensions:**

- Extensions to basic module (module MQA, page 65):
  - \* In order to investigate the influence of certain parameters of a model on the results, a new *sampling* analysis feature has been integrated. Sampling is the repeated execution of analyses where the values of design or basic variables are specified explicitly. Sampling works analogous to optimization and reliability analysis. So, all possible design or basic variables can be used for sampling and all analysis steps supported by optimization and reliability analysis can be used for sampling as well. For all desired results, sampling collects this data and prepares special XY data for convenient post-processing. Some examples for suitable variables are geometric parameters, Young's modulus for stiffness considerations, or different load parameters.
- Extensions to contact analysis (modules CA, CAX, and CAU from page 66):
  - \* *Self contact* is now supported by contact analysis (see Fig. 9), i.e. contact models where a surface may get in touch with itself, e.g. due to warping or other forms of large deformations, and where contacting regions are not known in advance (see module CA, page 66).
  - \* Contacting bodies are often only supported by contact forces. To provide an elastic foundation to such bodies, a new element family of *compensation springs* (also called *zero force springs*) has been introduced, where the spring force is compensated by an additional contact

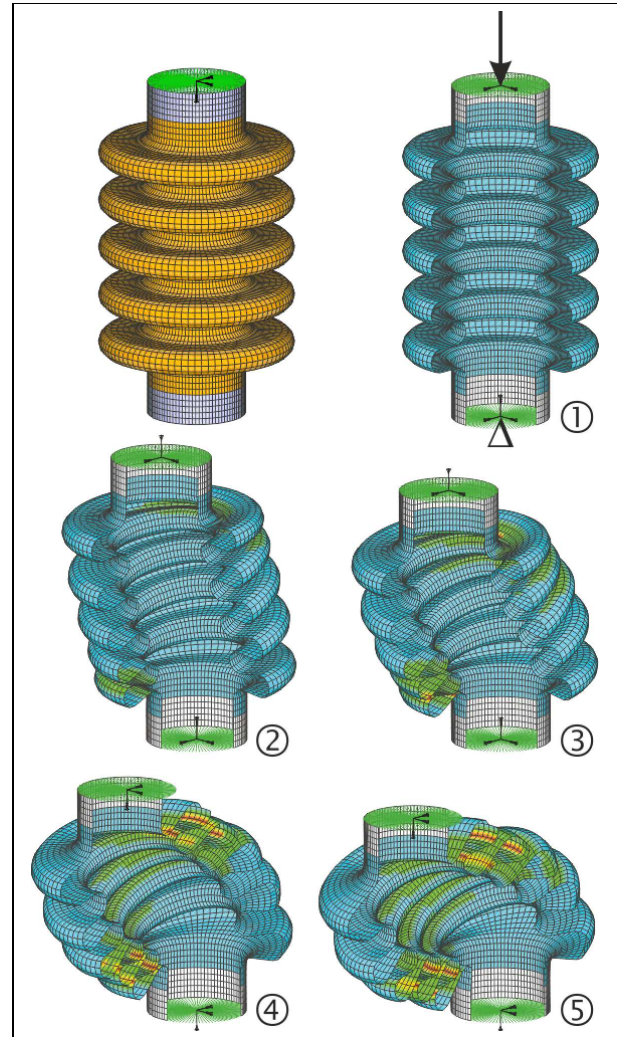


Figure 9: Self contact of a bellow

force (see Fig. 10 and below paragraph about new elements). By this means, a *force-guided contact* is facilitated. In addition, the new *RBM Assistant* in VisPER helps to place and properly define such elements (see next chapter on VisPER news).

- \* For the linearization of contacts to perform a subsequent linear analysis (like vibration analysis), various thresholds for the locking of contacts are available (see *contact locking* on page 68). E.g., the locking can be made dependent on a threshold of the contact pressure. Beyond this pressure value, a coupling is performed, below this threshold no coupling takes place. Frequently, contact analysis and subsequent linear analysis (like vibration analysis) are made in sep-

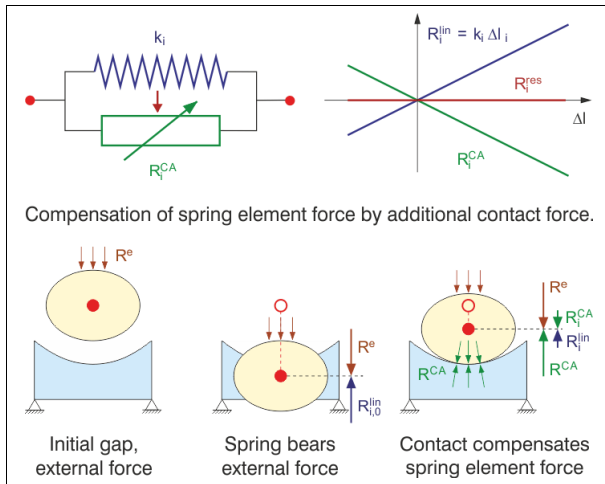


Figure 10: Force-guided contact with the aid of compensation spring

arate runs. To support this, contact status files from contact analysis can be used by subsequent runs to generate the locking of contacts (see also about contact status files on page 68).

- \* Contact geometry update has been made more stable and faster. One measure was to extend the contact status files to take geometry update into account (see Fig. 11).
- Extensions to nonlinear static analysis (module NLS, page 70):
  - \* PERMAS provides two different solvers for nonlinear static analysis according to the degree of non-linearities involved: NLMATERIAL/NLINERTIA and STATIC/INERTIA. Now, STATIC/INERTIA has been extended to take temperature-dependent material into account, too. In addition, special algorithmic improvements give STATIC/INERTIA in those cases a high potential for run time savings compared to NLMATERIAL/NLINERTIA (see engine example in Fig. 12).
  - \* A new *post-buckling* feature has been integrated (see Fig. 13). This allows the static analysis beyond buckling. In addition, after nonlinear analysis a linear buckling can be performed to study all possible buckling modes at this point of loading.
  - \* Geometrically nonlinear calculations

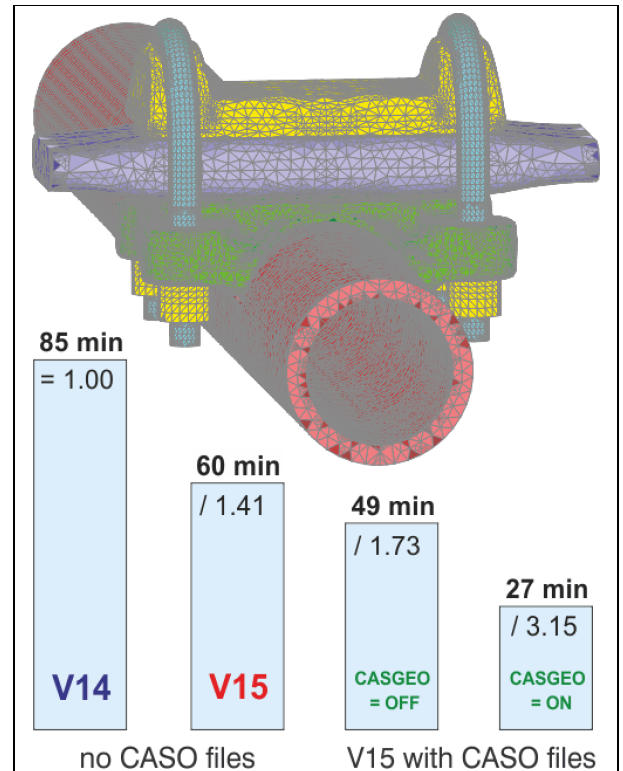


Figure 11: Contact geometry update with contact status files

can now be also made for composite and sandwich materials (with elements SHELL3 and SHELL4 for composites and elements TRIA3S and QUAD4S for sandwich shells). This includes buckling and post-buckling capabilities.

- Extensions to dynamic eigenvalue analysis (module DEV, page 73):
  - \* Nonstructural masses can be assigned to solid elements. Beside absolute mass values, mass per volume can also be specified.
- Extensions to dynamic eigenvalue analysis with MLDR (module MLDR, page 76):
  - \* Additional matrices like geometrical stiffness can now also be taken into account.
  - \* Additional static mode shapes can now be additionally taken into account, when results are requested only in the top component.
- Extensions to the extended dynamic eigenvalue analysis (module DEVX, page 74):
  - \* Backtransformation after complex eigenvalue analysis can be reduced to a subset of modes.



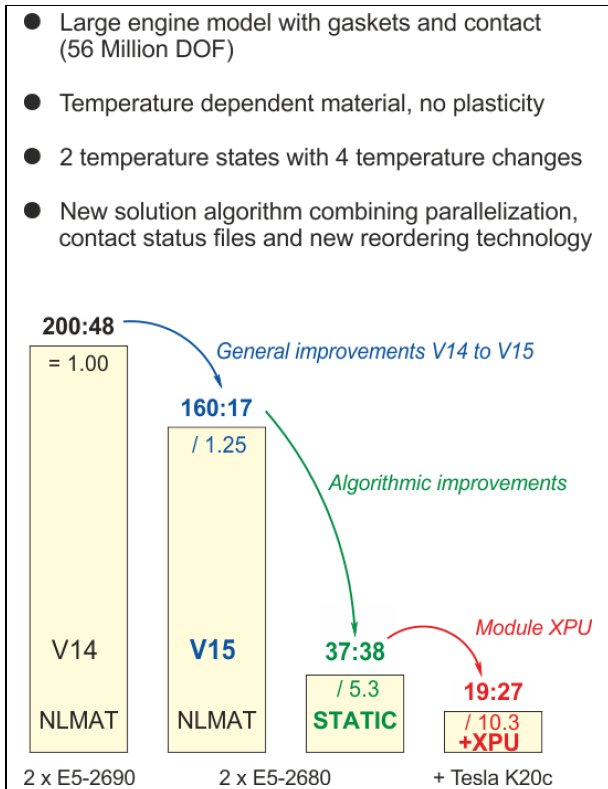


Figure 12: Run time reduction in engine analysis for temperature-dependent material

- Extensions to dynamic response analysis (module DRA, page 76):
  - \* Modal response analysis in time and frequency domain can be used to provide loading matrices to enhance matrix models (see about matrix models on page 61).
- Extensions to fluid-structure coupled analysis (module FS, page 79):
  - \* Coupled vibration analysis and modal coupled frequency response analysis support automatic substructuring and results in the top component.
  - \* Additional static mode shapes can now be taken into account.
- Extensions to design optimization (module OPT, page 83):
  - \* A new *non-parametric optimization* feature has been integrated which allows a *free-form shape optimization* of structures for minimizing stresses by homogenization or for limiting stresses at minimum weight (see Fig. 14). This opens the most easy way to define shape optimization of free-form geometries. The

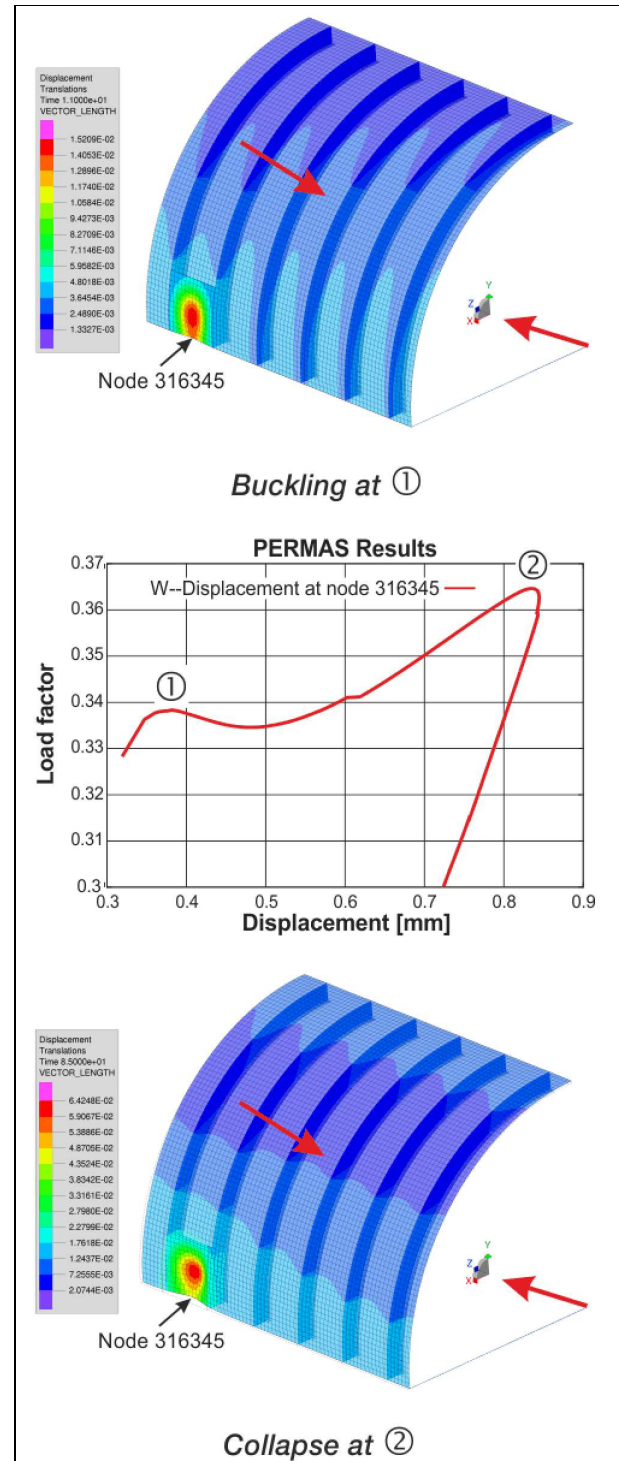


Figure 13: Elastic-plastic buckling with post-buckling behavior

set-up of this optimization is supported by a new freeshape wizard in VisPER (see next section on VisPER news on page 14).

- \* *Design constraint linking* has been introduced as powerful tool to build constraint



equations for multiple result values.

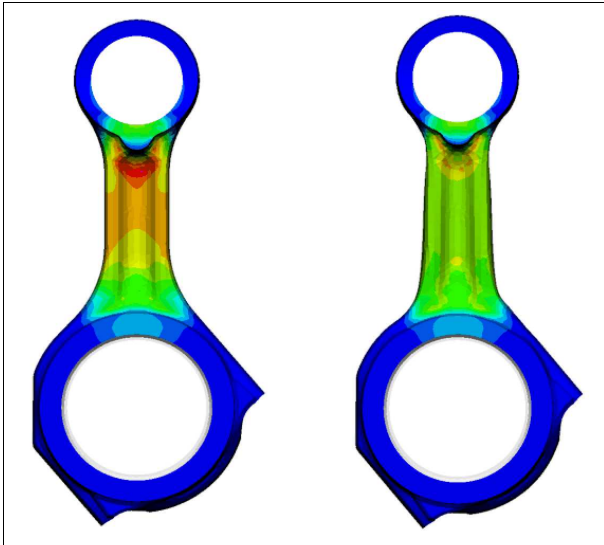


Figure 14: Stress homogenization of conrod

- Extensions to topology optimization (module TOPO, page 86):
  - \* Topology optimization is now able to provide a solution with clear separation of filled and void elements. So, elements will show filling ratios near 0 or near 1. This feature avoids misinterpretation of topology optimization results and facilitates the use of the result directly for further analysis and design steps (see Fig. 15).
- Extensions to advanced optimization solvers (module AOS, page 88):
  - \* Optimization is now also possible for complex eigenvalue analysis (including rotating structures).
  - \* *Design constraint linking* has been introduced as powerful tool to build constraint equations for multiple result values.
- Extensions to reliability analysis (module RA, page 89):
  - \* Reliability analysis is now also possible for complex eigenvalue analysis (including rotating structures).
- **New elements:**
  - An element family of *zero force springs* (also denoted as *compensation springs*) has been introduced to be solely applied in conjunction with contact analysis. These elements can compensate rigid body modes finally supported by contact forces without introducing

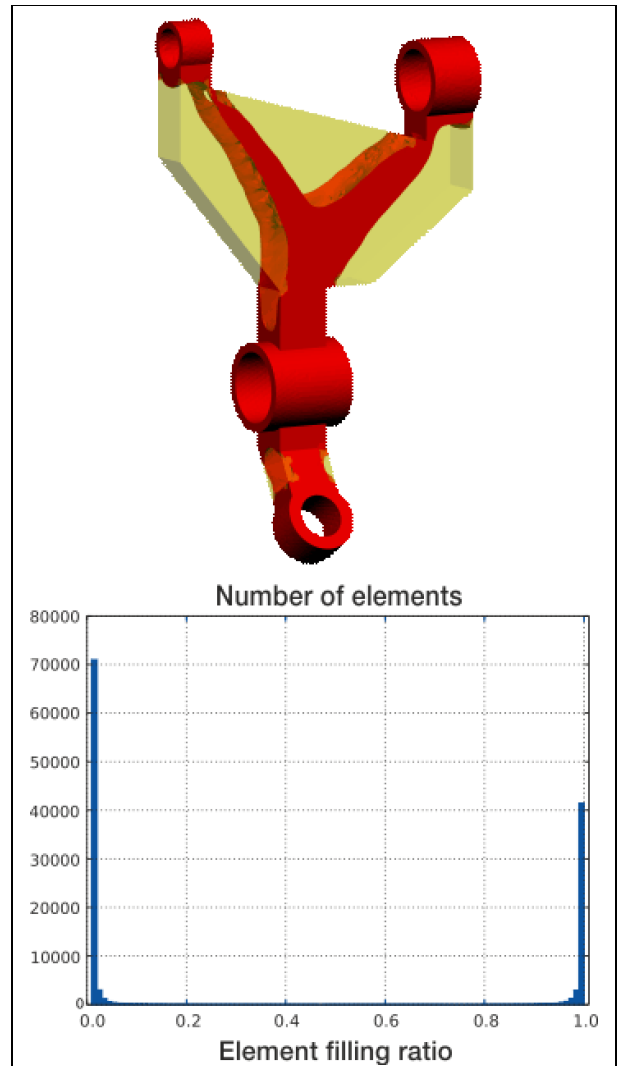


Figure 15: Fully converged topology

additional spring forces (see Fig. 10).

- Linear load elements (see page 56) and linear 3D shell elements (see page 55) now provide pressure stiffness as additional stiffness effect.
- **New kinematic constraint conditions:**
  - For the linearization of surfaces which consists of element faces with quadratic shape functions, a new kinematic constraint (ISURFLIN) is available. This feature can be used e.g. to linearize surfaces in case of contact.
  - For the interpolation of values in a volume, a new kinematic constraint is available (IVOLUME). This can be used e.g. for the generation of results at points where no element nodes are available.

Many smaller extensions of almost all functional

modules had been performed in addition. Moreover, all interfaces were updated and adapted to the new functionalities. Major interface enhancements are:

- **MEDINA (MEDI)** (see page 95):
  - Support of MEDINA 8.4.2,
  - Self contact definition enabled,
  - Export of new PERMAS Version 15 results.
- **ABAQUS (ABA)** (see page 98):
  - Extended interface for mass, beam, membrane and connector elements,
  - Support of pre-tension section,
  - Additional loads: centrifugal, gravity and surface pressure,
  - Support of steady-state heat transfer analysis, natural frequency analysis, steady-state dynamics analysis.
- **NASTRAN (NAS)** (see page 97):
  - An option is provided to generate QUAMS elements instead of QUAD elements,
  - Support of pyramid elements,
  - Extended support of OUTPUT2 format.
- **HyperView (H3D)** (see page 97):
  - Support of HyperView 11.0 Libraries,
  - Post-processing on HyperView 8.0 libraries possible.
  - Export of new PERMAS Version 15 results.

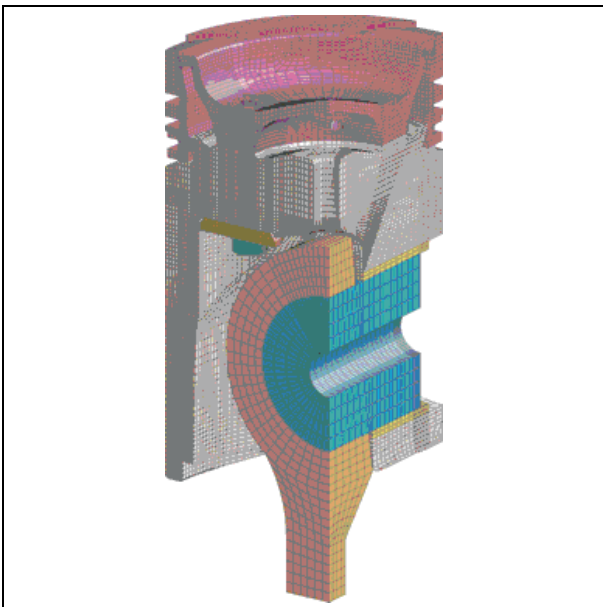


Figure 16: Ship engine piston  
Mahle GmbH, Stuttgart, Germany.

For all system platforms an update to the current release of the operating system had been performed (see Page 101).

## What's New in VisPER Version 4

Great effort has been spent in the past years to provide **VisPER** (i.e. Visual PERMAS) as a dedicated tool to improve pre- and post-processing for special PERMAS functions. VisPER Version 4 is released at the same time as PERMAS Version 15. More information on VisPER can be found on page 39.

Because the list of extensions in Version 4 is extra long, the following overview just lists the most important extensions. More information can be found in the VisPER Users Manual.

The list of major software extensions in VisPER is divided in four sections as follows:

- **Model completion:**
  - Multi-point Constraints (MPC) can be created, verified and visualized.
  - Extended **element generator** functions for assignment of parts or element sets, converting and splitting of elements.
  - Element quality is checked and plain text diagnostics are shown with an automatic look-at and annotation.
  - Sets of nodes and elements can be modified and set bins (sets of sets) are also supported.
  - A dialog bar is available to specify rigid body modes explicitly.
  - System variants can now be specified by a dialog bar for inserting other variants, adding other matrix contributions (like geometrical stiffness) or specifying effects of temperature-dependent material or loads dependent on previously calculated results (e.g. for geometrical stiffness).
  - A dialog bar is available to support subcomponent positioning via a coordinate system.
  - A new **RBM Assistant** (Rigid Body Mode Assistant) is available which provides an efficient workflow to generate elastic supports for static models. In particular for contact models, the new compensation springs can be comfortably generated by this assistant (see page 68).
  - The **mapping** of node temperature loadings is supported by a special workflow.
- **Wizards:**
  - A new Pretension Wizard allows an easy description of complex bolt definitions according to PERMAS pretension technology. See

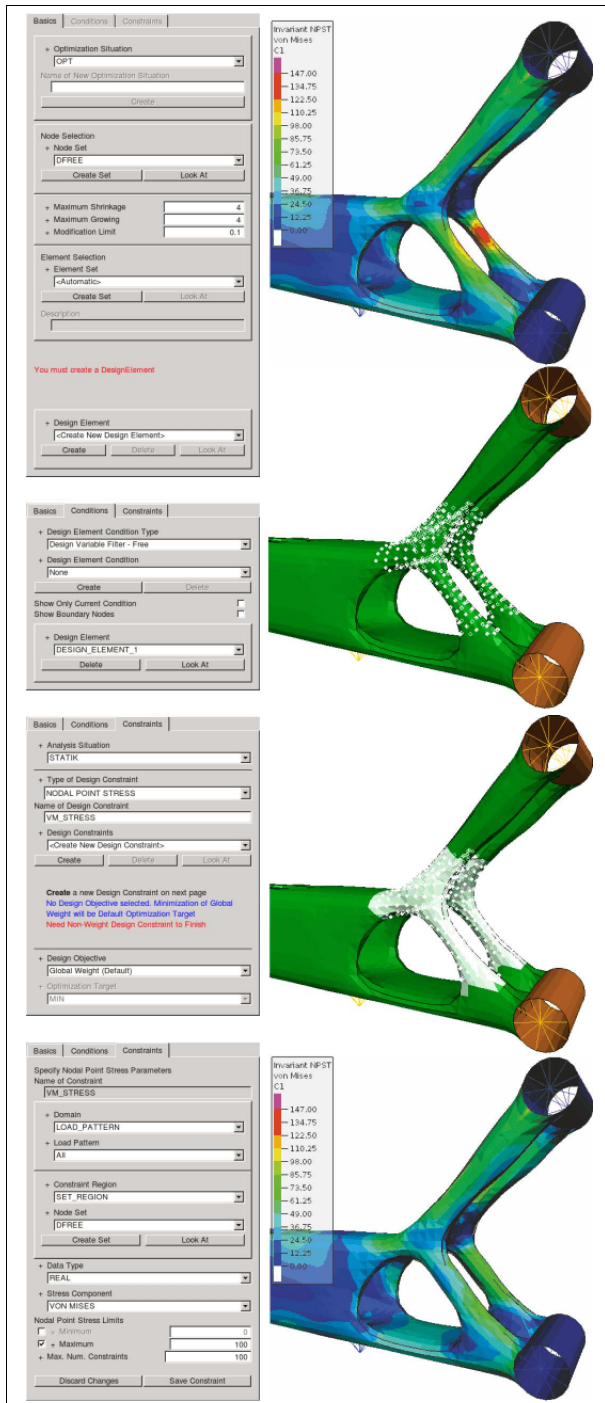


Figure 17: Pre- and post-processing for free-form optimization:

- Initial shape with too high stresses
- Definition of surface node set to specify design area
- Automatic selection of design space in thickness direction
- Modified shape after free-form optimization with reduced stresses

module VCA on page 46.

- A new Freeshape Wizard has been devel-

oped for a parameter-free shape optimization to reduce stress concentrations. See module VOPT on page 42.

- A new Substructure Wizard has been developed as a guide to build substructure models. See page 40.

The already existing wizards have been updated to cover the current state of the PERMAS functions.

#### • Post-processing:

- Complex mode shapes can be visualized taking the phase angle into account.
- Direct creation of animated GIF images for animations.
- Extended selection of components for visibility by show or hide elements.
- Gasket pressure-closure curves can be plotted now.
- Improved handling of active and inactive clip planes.

#### • Tools:

- A *message tree* is provided to summarize diagnostics and to provide additional information (like tool tips). The dialog provides parsing of the PERMAS protocol file. There is a direct relation to model entities mentioned in the diagnostics.
- Information about the model is provided by a tree view of all processed input files.
- VisPER can be configured to submit a PERMAS batch job.
- Imported results and the related post-processing model can be exported to other formats (like MEDINA or HyperView).

## Universal Features

The outstanding mostly module-independent basic features of PERMAS are as follows (see pages 49 to 63):

- Hierarchical substructuring, with automatic sub-component insertion (see page 49)
- Submodeling (see page 49)
- Variant analysis (see page 50)
- Cyclic symmetry (see page 51)
- Surface and Line Description (see page 51)
- Automated coupling of parts (see page 52)

- Automated spotweld modeling (see page 53)
- Multiple kinematic constraints (see page 53)
- Automatic detection of singularities (see page 55)
- Same elements for different analysis types (Element library, see page 55)
- Standard beam cross sections (Seite 56)
- Design elements for optimization (page 56)
- Error estimation and refinement indicator (page 56)
- General material description (see page 57)
- Node and element sets (see page 57)
- Mathematical functions (see page 58)
- All kinds of loading (see page 58)
- Model verification (see page 58)
- Integrated interfaces to pre- and post-processors (see page 60)
- Input and Output of Data Objects and matrices (see page 61)
- Combination, transformation, and comparison of results (see page 62)
- Output of XY result data (see page 62)
- Calculation of cutting forces (see page 63)
- Restart facility (see page 63)
- Open software through Fortran and C interfaces (see page 63)
- Direct coupling of different analysis types (see page 63)
- Coupling with CFD (see page 64)

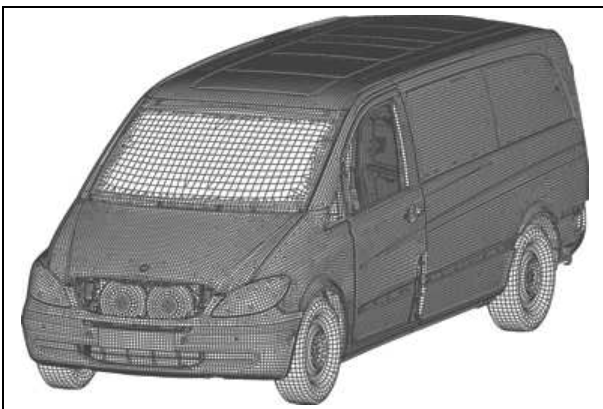


Figure 18: Model of a transport vehicle  
courtesy of Daimler AG,  
Commercial Vehicle Division in Stuttgart

## Available VisPER Modules

The below listed functional modules are explained in more detail on pages 40 to 46:

- |                            |        |
|----------------------------|--------|
| • Basic module             | (VBAS) |
| • Topology optimization    | (VTOP) |
| • Design optimization      | (VOPT) |
| • Fluid-structure coupling | (VFS)  |
| • Contact analysis         | (VCA)  |

## Available PERMAS Modules

The below listed functional modules are explained in more detail on pages 65 to 99:

- |                                              |          |
|----------------------------------------------|----------|
| • Model Quality Assurance                    | (MQA)    |
| • Linear Statics                             | (LS)     |
| • Refined Weldspot Model                     | (WLDS)   |
| • Contact Analysis                           | (CA)     |
| • Extended Contact Analysis                  | (CAX)    |
| • Contact Geometry Update                    | (CAU)    |
| • Nonlinear Statics                          | (NLS)    |
| • Extended Nonlinear Material Laws           | (NLSMAT) |
| • Buckling Analysis                          | (BA)     |
| • Dynamic Eigenvalue Analysis                | (DEV)    |
| • Extended Dynamic Eigenvalue Analysis       | (DEVX)   |
| • Eigenmodes with MLDR                       | (MLDR)   |
| • Dynamic Response Analysis                  | (DRA)    |
| • Extended Dynamic Response Analysis         | (DRX)    |
| • Fluid-Structure Acoustics                  | (FS)     |
| • Nonlinear Dynamics                         | (NLD)    |
| • Heat Transfer                              | (HT)     |
| • Nonlinear Heat Transfer                    | (NLHT)   |
| • Laminate Analysis                          | (LA)     |
| • Design Optimization                        | (OPT)    |
| • Layout Optimization                        | (TOPO)   |
| • Advanced optimization solvers              | (AOS)    |
| • Reliability Analysis                       | (RA)     |
| • Steady-state electromagnetics              | (EMS)    |
| • Electrodynamics                            | (EMD)    |
| • Use of GPU                                 | (XPU)    |
| • Interfaces to various pre-/post-processors |          |
| – MEDINA                                     | (MEDI)   |
| – PATRAN                                     | (PAT)    |
| – I-DEAS                                     | (ID)     |
| • Interfaces to other analysis packages      |          |
| – ADAMS                                      | (AD)     |



– DADS	(DADS)
– SIMPACK	(SIM)
– EXCITE	(EXCI)
– MOTIONSOLVE	(HMS)
– HYPERVIEW	(H3D)
– VAO	(VAO)
– Virtual.Lab	(VLAB)
– MATLAB	(MAT)
– NASTRAN	(NAS)
– ABAQUS	(ABA)
– MpCCI	(CCL)

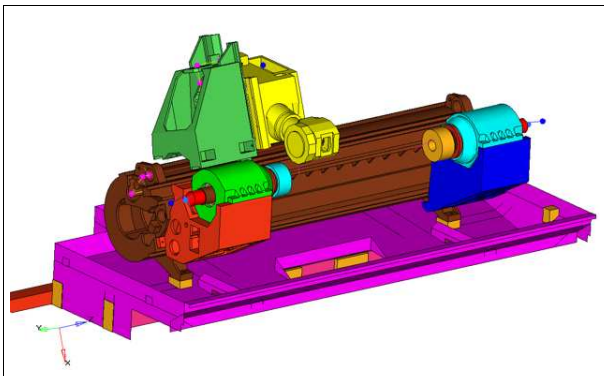


Figure 19: Machine tool  
INDEX-Werke GmbH & Co. KG, Esslingen, Germany

## Performance Aspects

By ongoing further developments of the equation solvers PERMAS achieves a very high computation speed. Both, direct and iterative solvers, are continuously optimized.

- Very good multitasking behavior due to a high degree of computer utilization and a low demand for central memory.
- The central memory size used can be freely configured – without any limitation on the model size.
- The disk space used can be partitioned on several disks – without any logical partitioning (e.g. optimum disk utilization in a workstation network).
- There are practically no limits on the model size and no explicit limits exist within the software. Even models with many million degrees of freedom can be handled.
- By using well-established libraries like BLAS for matrix and vector operations, PERMAS is

adapted to the specific characteristics of hardware platforms and thus provides a very high efficiency.

- Another increase of computing power has been achieved by an overall **parallelization** of the software.
- By simultaneous use of several disks (so-called disk striping) the I/O performance can be raised beyond the characteristics of the single disks.
- Disk I/O can be avoided at all, if large memories are used. The memory size may readily exceed 256 GB.

## Parallelization

PERMAS is also fully available for parallel computers. A general parallelization approach allows the parallel processing of all time-critical operations without being limited to equation solvers. There is only **one** software version for both sequential and parallel computers.

PERMAS supports the parallelization on shared memory computers. There, the parallelization is based on POSIX Threads, i.e. PERMAS is executed in several parallel processes, which all use the same memory area. This avoids additional communication between the processors, which fully corresponds with the overall architecture of such systems.

In addition, PERMAS allows asynchronous I/O on this architecture, which realizes better performance by overlapping CPU and I/O times.

Parallelization does not change the sequence of numerical operations in PERMAS, i.e. **the results of a sequential analysis and a parallel analysis of the same model on the same machine are identical** (if all other parameters remain unchanged).

PERMAS is able to work with constant and pre-fixed memory for each analysis. This also holds for a parallel execution of PERMAS. So, several simultaneous sequential jobs as well as several simultaneous parallel jobs or any mix of sequential and parallel jobs are possible.

The parallelization is based on a mathematical approach, which allows the automatic parallelization of sequentially programmed software. So, PERMAS

remains generally portable and the main goal has been achieved: *One single* PERMAS version for all platforms.

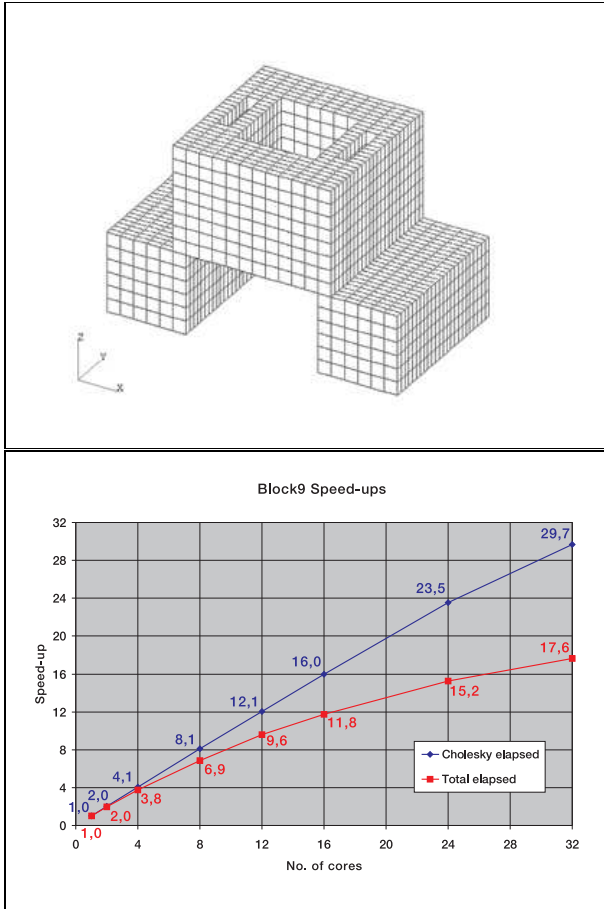


Figure 20: Static analysis with 3 loading cases  
1.5M nodes, 176k HEXE27, 4.4M Dof  
run time on Intel Boxboro

Parallel PERMAS is available for all platforms, where a sequential version is supported, too.

The parallelization on several cores can be extended by the use of a GPU (Graphical Processing Unit) of Nvidia, where a Tesla K20c or better is supported. For compute bound solution steps, the GPU can essentially accelerate the analysis run (see also module XPU on page 93).

The parallel execution of PERMAS is very simple. Because there are no special commands necessary, a sequential run of PERMAS does not differ from a parallel one - except for the shorter run time. Only the number of parallel processes or processors for the PERMAS run has to be defined in advance.

## Areas of Application

Presently, PERMAS is used in the following branches of industry:

- Automotive industry
- Aerospace industry
- Ship building industry
- Mechanical engineering
- Offshore- and power plant engineering
- Plant- and equipment engineering

## Reliability

Nowadays, not all results of FE analyses can be proven by experiments. They are often directly used in the development process. Moreover, the models become more and more complex and the results have to be produced faster and faster. Early detection of possible modeling errors and their elimination means a great challenge to the analysis software. To this end, PERMAS and VisPER make a substantial contribution.

- **Robustness of the software:** Low system error rate due to advanced software engineering methods and intensive software testing.
- **Model verification:** The basic PERMAS-MQA module provides tools for model quality assurance (see page 65). Beside automatic model testing, many quantities and model properties can be exported for visualization and checking in a postprocessor (see section Model Verification on page 58). In addition, VisPER provides a model verification environment for a growing number of modeling parameters (see page 40).
- **Safe use:** Expensive faulty runs are avoided by the task scanning concept of PERMAS-MQA. Firstly, these give an estimation of the necessary computer resources, which allow for a more reliable planning of large model analyses. In addition, numerous modeling deficiencies can be detected, which directly improves the reliability and quality of the subsequent analysis.
- **Correctness of results:** The quality of results is ensured by comprehensive and continuous verification (using the tests of NAFEMS and SFM).

Above all, the application of well-proven algorithms and esteemed development tools results in the high quality of the software.

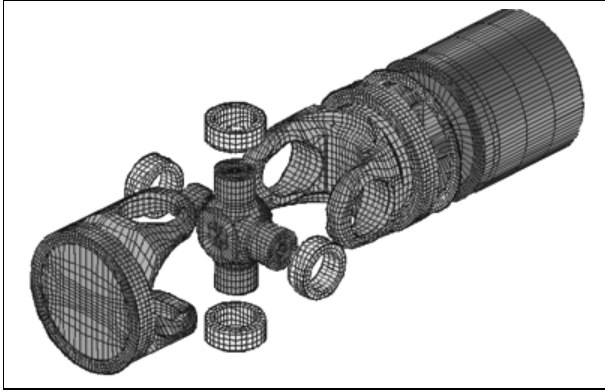


Figure 21: Model of a cardan shaft

Voith Turbo GmbH & Co. KG, Heidenheim, Germany.

A broad traditional PERMAS user base from different branches of industry essentially contributes to the reliability of the software.

---

## Quality Assurance

INTES develops high quality software und offers all related services. All phases of the software development are performed on the basis of established standards and appropriate tools in order to achieve a maximum of product quality.

Some important aspects of quality assurance are:

- Especially developed for the management and administration of the software, a development tool provides for a safe software database, which includes all modifications and new sub-routines and manages them in a unique and apprehensible way.
  - A problem report management system gathers all messages regarding software problems and development requests as well as other user requests together with the subsequently elaborated solutions and responses. A 'Technical Newsletter' issued regularly informs the users about all inquiries made and the pertinent solutions.
  - An ever growing library of software test runs daily ensures the equally high quality of the software. Problem cases extracted from the problem report management system lead to an extension of the test library in order to preclude the re-occurrence of problems handled in the past.
-





## Applications

### Car Body Analysis

Finite Element Analysis of car bodies comprise a broad variety of modeling levels from *BIW (body-in-white)* to trimmed bodies and acoustic models taking into account enclosed and even surrounding air. This variety of structural variants corresponds to different targets from simple stiffness issues up to complex comfort tasks. Therefore, a lot of different methods are applied in car body analysis ranging from linear static analysis up to fluid-structure coupled acoustics.

A typical characteristic of car body models is the use of shell elements. Most frequently, quadrangular linear shell elements are used (together with triangular shell elements). Dependent on the mesh size, up to several million shell elements are used to model car bodies. A car body consists of a larger number of structural parts (typically 50 to 100) which are joined by different techniques like spot welding (see page 53 and module WLDS on page 93), bonding, laser welding. In order to generate the meshes of all parts efficiently, *incompatible meshing* (see page 52) is used for independent meshing.

A special feature in VisPER supports post-processing of spotwelds (see page 48) in very large body structures.

#### Static analysis

For computations of static stiffness of a car body, linear static analysis is used. For some load cases like towing or light impact calculations of *inertia relief* (see page 66) are applied.

To check the force flow through any structural member, *cutting forces* (e.g. through a column or sill, see page 63) can easily be derived and a summary of the forces and moments is exported (and printed).

#### Dynamic analysis

It is an important issue in dynamic analysis that all masses are taken into account. The matching of masses between the real structure and the simulation model is very important. Masses and moments of inertia can be calculated by the simulation and compared to the expected values.

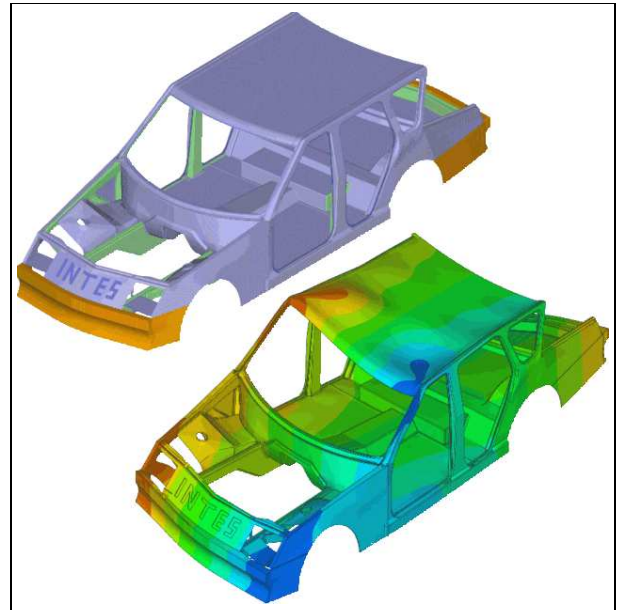


Figure 22: Workshop example INTEScar under torsional loading

An eigenvalue analysis is performed as a basis for subsequent response analysis. Because cars are not supported on ground, a free-free vibration analysis has to be performed. A check on the rigid body modes is highly recommended and supported by corresponding printed information. The frequency range for the eigenvalue analysis depends on the intended frequency range of the subsequent response analysis. A certain factor (2 to 3) on the intended frequency range is frequently applied in order to get good response results over the full frequency range.

Flexible bodies are often incorporated in *MBS (Multi-Body Systems)* models. Usually, this is done on the basis of modal models. PERMAS supports a number of interfaces to export flexible bodies in special formats (see page 96).

Due to the cut of eigenfrequencies beyond the frequency range, response results can be insufficient in the quasi-static range (between zero frequency and first eigenfrequency). This quasi-static response can be improved by taking relevant static mode shapes which are computed automatically from given static load cases (see page 78).

Structural modifications of the car body (BIW) are usually done for only a few parts, e.g. the front of the car. Then, there is no need to repeat the full analysis of the car from scratch but the rear car can be

reduced by dynamic condensation (see page 74). Using dynamic condensation so-called *matrix models* (see page 61) are generated which represent the reduced part of structure. These matrix models are used in each analysis of the remaining structure. In this way, run time for variants (e.g. of the front car) is reduced drastically.

For the subsequent response analysis (see page 76), there are methods for the frequency domain (i.e. frequency response analysis) and for the time domain (i.e. time-history response analysis). These methods are available as modal methods (based on previously determined eigenfrequencies and mode shapes) and as direct methods (based on full system matrices). For realistic models, the direct methods are much more time consuming than modal methods. But the direct methods are very accurate and can be used on a case-by-case basis to check the accuracy of the modal models.

The dynamic loading (or excitation) can be specified by forces (and moments) or prescribed displacements (or rotations) and a frequency or time function which describes the course of the excitation dependent on frequency or time.

- In frequency domain, the discretization of the excitation frequency range is an important accuracy parameter for the resulting response graphs. In particular, the discretization of peaks is important and this is supported by generation of clusters of excitation frequencies around eigenfrequencies.
- If a time function is provided by measurements, beside a time-history response an alternative approach is also available to get a periodic response result. An internal *FFT (Fast Fourier transformation)* is available to detect the main excitation frequencies. For each of these frequencies a frequency response can be performed (with just one excitation frequency). The result of all these harmonic response results can then be superimposed in the time domain to get the periodic response (or *steady-state response*). Fig. 88 shows an example.
- In time domain, the sampling rate should be related to the time characteristics of the excitation function.

For response analysis, the specification of damping is very important. There are a lot of ways to specify damping (see page 78). In particular, trimmed bod-

ies require a detailed and accurate modeling of all additional springs, masses, and dampers.

The results from a frequency response analysis are any complex primary result (displacements, velocities, or accelerations) and secondary result (e.g. stresses, strains, sound radiation power density) for all nodes at any excitation frequency. Frequently, so-called *transfer functions* are more important than the full fields of result quantities. Transfer functions describe the relation between the excitation points and any target point of interest (by a unit excitation) for all excitation frequencies.

In order to reduce computational effort for response analysis the user can specify the requested results in advance. In case of requested transfer functions, the response analysis can be restricted to just a node set.

### Fluid-structure dynamics

Coupled simulation of structure and air is seen as natural extension of structural dynamics. This extension is needed, because noise in a car is a combination of structural-borne and air-borne noise. Noise at the driver's ear is important for the comfort and the acoustic quality of a car.

As a first step the interior of the car is modeled by so-called fluid elements which are classical volume elements but with a pressure degree of freedom. In order to model the coupling between structure and air physically, there are additional coupling (or interface) elements which contain both the displacement and pressure degrees of freedom and represent the physical compatibility condition between structure and air.

To facilitate the two modeling steps for fluid and coupling elements of the car interior, VisPER contains an easy-to-use wizard starting from the structural mesh and generating the fluid mesh and the coupling elements step by step in an almost automatic way (see page 44). Typically, the coupling elements are compatible with structural elements of the interior surface, but the fluid elements representing the enclosed air are incompatibly meshed, because the mesh for the air is usually much coarser than for the structure. The wizard derives the appropriate element edge length from the requested frequency range.

The fluid may contribute to the damping by so-called

volumetric drag which represents the absorption in a fluid volume. The coupling elements contribute to the damping by surface absorption which represents a normal impedance of the coupling surface.

After completing the fluid-structure model, the analysis steps are very similar to structural dynamics of cars as described above (see also page 79 for more functional details).

- A coupled eigenvalue analysis is available to derive the coupled eigenfrequencies and mode shapes. The mode shapes consist of two corresponding parts, a displacement mode shape of the structure and a pressure mode shape of the fluid.
- Excitations can now also be specified in the fluid by a pressure signal.
- Based on coupled eigenfrequencies and mode shapes, modal frequency response analysis and modal time-history response analysis can be performed in the same way as for the sole structure.

In addition to modal methods, also a direct frequency response is available for fluid-structure coupled analysis.

From the coupled response results, all results as described for structural response calculations can be obtained. In addition, the pressure field in the air and transfer functions from structural points to pressure points are available (and vice versa). Moreover, sound particle velocities (as vector field or magnitudes) can be derived from the pressure field.

In addition to enclosed air in a car, the surrounding air can also be modeled and coupled to the structure. This feature can be used to calculate noise transition through the structure (from the road or from air flow induced noise to the driver's ear).

### High performance

Continuous effort is spent in improving and accelerating the speed of algorithms. In car body analysis emphasis is put on the following achievements:

- For large models (millions of degrees of freedom) and many modes (thousands of modes), eigenvalue analysis is made much faster by *MLDR (Multi-Level Dynamic Reduction)*. Details can be found on page 76. This method is available for both structural dynamics and coupled fluid-structure dynamics.

- In frequency response analysis many different dynamic load cases (several hundreds) are often applied. So-called *assembled situations* (see page 78) are used to solve these load cases simultaneously instead of one after the other.
- In frequency response analysis the equation solving can be made much faster (for a high number of modes and many excitation frequencies) using an iterative solver.

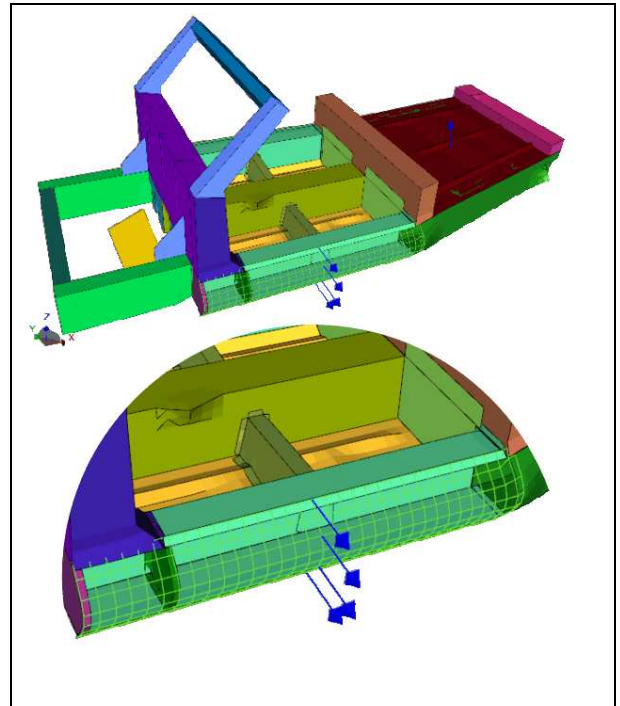


Figure 23: Shape optimization of a sill with transition to neighbored parts

### Optimization

Supported by VisPER and PERMAS, optimization tasks for the car body can be solved in an integrated way. So, the optimization model is part of the model description and can easily use all available references to existing model parts like node and element sets. Although all available optimization types (as described on pages 83 to 89) can be used for car bodies, the most important ones are as follows:

- **Sizing:** This is used to optimize element properties like shell thickness, beam cross section, spring stiffness, and damper properties.
- **Shaping:** This is used to optimize geometry of parts by modifying node coordinates (also possible with incompatible meshes).
- **Bead design:** This is used to position and



shape beads in shell structures (see example in Fig. 24).

All these optimization types can be combined in one optimization project. Static and dynamic analysis can be used simultaneously for optimization tasks. The optimization modeling is fully supported by VisPER (see details on page 42). Even post-processing of optimization results can be made with VisPER.

Optimization of transfer functions due to sizing, shaping, and bead design is of major importance in dynamic analysis. This frequency response optimization can be used with an objective transfer function (i.e. a frequency dependent limit of amplitudes).

If the objective transfer function is derived from experimental results, then the optimization process is named *model updating*. By this process selected model parameters are modified in order to fit the simulation transfer function to the experimental one.

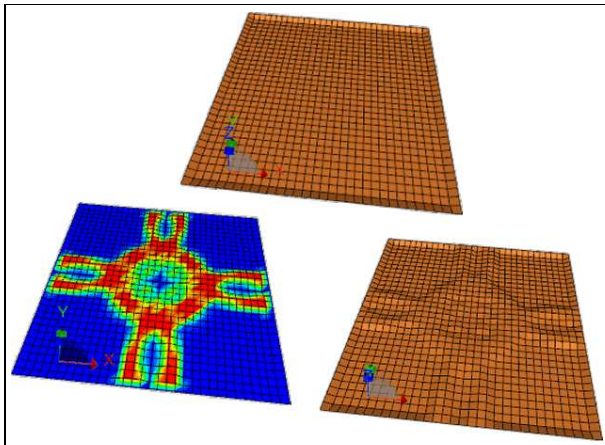


Figure 24: Bead design of a plate with positioning and height of beads

## Engine Analysis

Many physical effects play an important role during a mechanical analysis of combustion engines. In static analysis such effects are leak tightness and durability under changing temperature conditions and in dynamic analysis there are sound radiation and frequency responses of complex engine assemblies. At least in static analysis the influence of temperature requires a coupled analysis taking heat transfer into account. Modeling the mounting

of an engine requires the consideration of bolt loading conditions where the correct sequence of bolt pre-stressing and operating loads is of major importance. In addition, nonlinear material behavior has to be considered.

These and other effects are important for engine analysis.

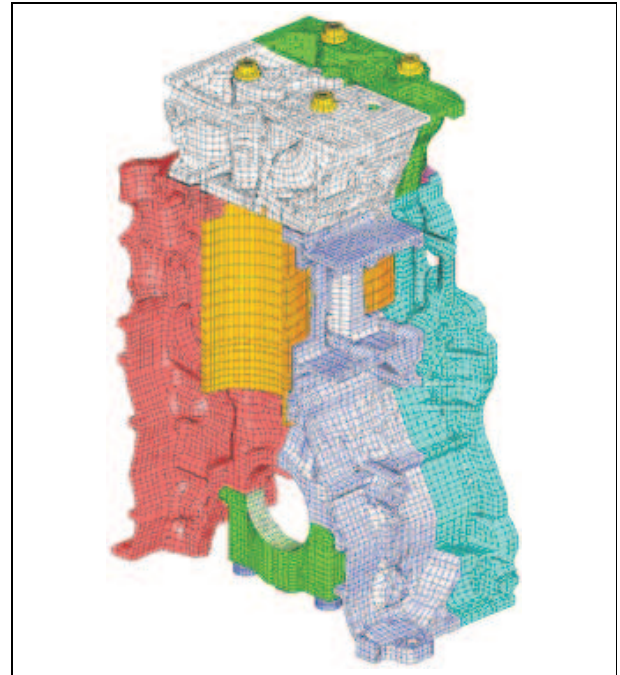


Figure 25: Simple engine model  
Daimler AG, Commercial Vehicle Division)

## Heat Transfer

Applications are e.g. the analysis of operating temperatures and the aging in an oil bath by simulating the cooling down process. The following features are available:

- Nonlinear material behavior with temperature-dependent conductivity and heat capacity,
- Temperature-dependent heat convection for the modeling of heat exchange with the surrounding,
- **Automatic solution method** for nonlinear heat transfer with **automatic step control** and several convergence criteria, i.e. an automatic load stepping for steady-state analyses and an automatic time stepping for transient analyses,
- Convenient and very detailed specification possible for loading steps and points in time where results have to be obtained,
- Full coupling to subsequent static analysis (steady-state and transient),

- Heat exchange by radiation can be included, if this makes a relevant effect on the temperature field.
- If temperature fields are available for cylinder head and engine block, then other parts may not yet have temperatures, like gaskets or bolts. Then *temperature mapping* procedures using the submodel technique are available to provide all parts with proper temperatures (see submodeling on page 49).

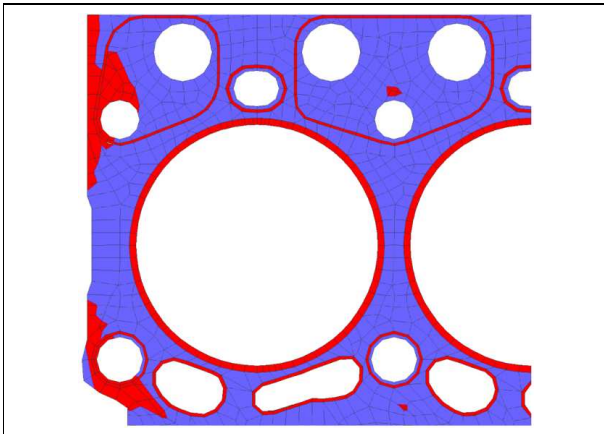


Figure 26: Contact status for gasket elements

## Statics

Static deformations are calculated under various loads with linear and nonlinear material behavior:

- Nonlinear material models:
  - plastic deformation,
  - nonlinear elastic,
  - creep,
  - cast iron with different material behavior under tension and compression.
- *Gasket elements*:
  - for convenient simulation of sealings,
  - the behavior of sealings is described by measured pressure-closure curves,
  - input of many unloading curves possible.
- Contact analysis:
  - many contacts possible ( $> 100,000$ ),
  - unrivaled short run times,
  - most advanced solver technology,
  - friction can be taken into account with transitions between sticking and sliding,
  - bolt conditions can be applied in one step,
  - specification of a realistic loading history,
  - If an engine has many parts, which are connected only by contact, the RBM assistant

in VisPER helps to avoid rigid body modes by applying compensation springs (see page 40).

- contact results: contact pressure, contact status, contact forces, saturation, etc..
- *Submodeling*:
  - for subsequent local mesh refinements,
  - automatic interpolation of displacements to get kinematic boundary conditions for a finer mesh,
  - then, a local analysis is performed e.g. to achieve more accurate stresses.

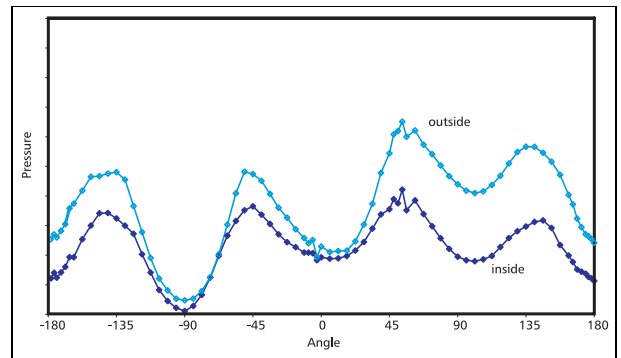


Figure 27: Pressure distribution at stopper over the angle

## High performance

Due to typically large models in engine analysis all analysis methods are oriented towards highest possible performance. The following points can be highlighted:

- outstanding performance through special algorithms for large models with nonlinear material and contact,
- contact algorithms have been strictly designed to meet the needs of large models with many contacts,
- unrivaled fast method for linear material and contact.
- Gasket elements can be handled as integral part of the contact iteration instead of a feature in nonlinear material analysis (i.e. *CCNG analysis*, Contact Controlled Nonlinear Gasket analysis). If no other material nonlinearities are present in the model, run time reduction factors can be higher than 10 (e.g. for analysis of combustion engines with pretension, temperature loads, and cylinder pressures). In cases, where other material nonlinearities are present in the model, a run time reduction by a factor of about 2 can still

be achieved.

- An additional speed-up can be obtained, if a contact analysis is repeated. The resulting contact status of a contact analysis is stored in so-called *contact status files*. These contact status files may be used as starting point for the subsequent contact analysis. In case of small changes, this will essentially reduce the run time of the new contact analysis.
- If several temperature fields are used several times in an engine analysis, e.g. to calculate several load cycles with different temperatures like in a cold and hot engine, then a special algorithm can be used to accelerate the analysis significantly (see Fig. 12).

## Dynamics

By using the same software for dynamic and static simulations only one structural model is necessary. All dynamic methods are available for engine analysis (see pages 73 to 79). Some important points are:

- Eigenvalues and mode shapes for large solid models can be calculated using MLDR (see page 76).
- Fast *dynamic condensation* methods support the efficient analysis of engines with many attached parts (DEVX, see page 74).
- By using *dry condensation* (page 74) even fluids can be integrated in a dynamic model without taking along pressure degrees of freedom (e.g. in an oil pan).
- Calculation of sound particle velocity is supported for the evaluation of noise emission of engines.

In order to facilitate the transition from static analysis with contact to dynamic analysis, a *contact locking* feature is provided (see page 68). By using this feature, the results of any loading state in static analysis of an engine can be used for a subsequent dynamic analysis. A contact pressure dependent threshold value for the locking of contacts is available to fit dynamic results to experiments, if necessary.

## Part Connections

The modeling of part connections essentially determine the quality of simulation results. On the other hand modeling of connection details is sometimes complex and time consuming. Hence, analysts want to have simplified models for various connections giving satisfactory simulation results. Consequently, part connection is a typical modeling feature in the area of tension between modeling effort and result quality.

There are two different classes of part connections which will be subsequently described in more detail:

- Structural connections,
- Connection elements.

### Structural Connections

Parts can be coupled at their surfaces in two different ways:

- Tight coupling (i.e. coupling remains under both tension and compression). This is typically achieved by kinematic constraints (see page 53).
- Contact (i.e. connection can open and close during loading). This is the topic of contact analysis (see page 66).

In both cases, the surfaces of coupled parts can be meshed compatibly or incompatibly (see the part coupling on page 52). The latter is an advantageous feature reducing modeling effort, because parts can be meshed independently.

Tight coupling can be used with all analysis types (like static and dynamic analysis). But in case of contact, e.g. a subsequent dynamic analysis needs one additional analysis step. This step includes a contact analysis where the final contact status is locked (i.e. *contact locking*, see page 68). In this way, a linearization of the contact problem is achieved. Fig. 28 describes the process of dynamic analysis for engine structures under pretension load. Before bolt pretension is applied, the parts of an engine can vibrate separately, but after bolt pretension is applied, the engine assembly vibrates as one single body. This behaviour is achieved by locking the contact areas where the parts are in contact. Other areas are kept uncoupled, where no contact is in place.

Once, contact locking is applied, eigenvalue analy-



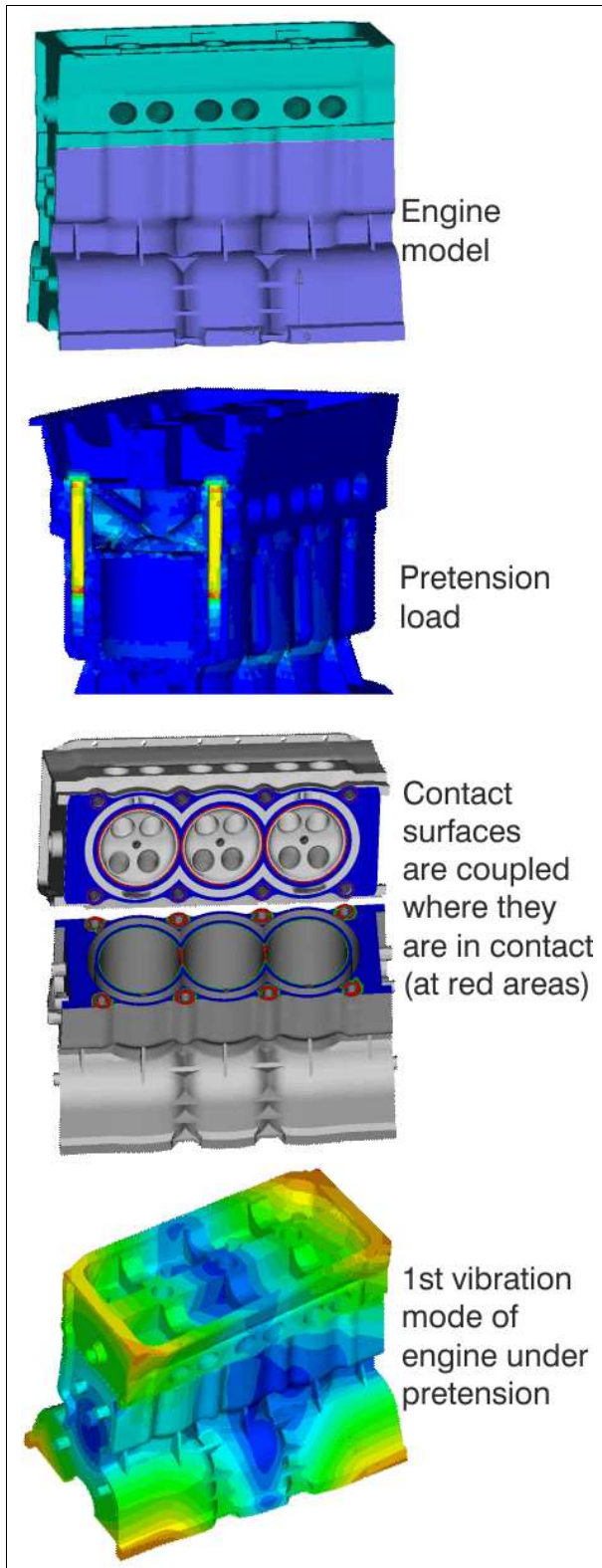


Figure 28: Dynamic analysis of an engine under pretension load

sis and frequency response analysis can be used. Even optimization of frequency response functions can be used, e.g. to reduce sound radiation of en-

gine. Fig. 29 shows the effect of moving the ribs on the engine surface and the effect of increasing rib thickness to reduce sound radiation of the engine block. The ribs are meshed incompatibly from the engine block. Hence, the ribs can easily be moved on the surface without re-meshing of engine block and ribs. This is used by a shape optimization.

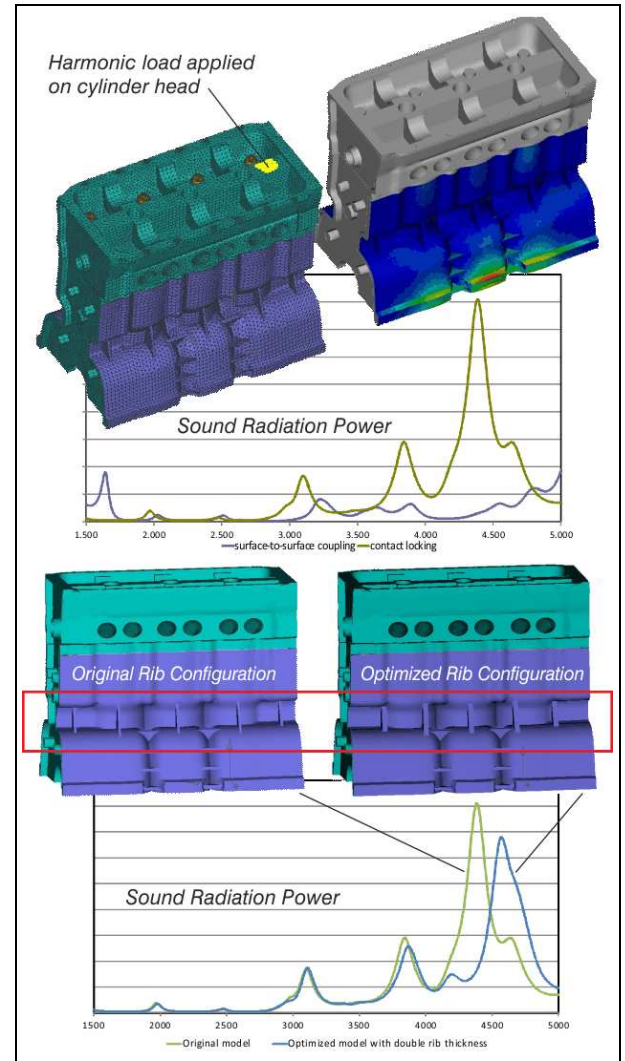


Figure 29: Optimization of rib thickness and rib position to reduce sound radiation from engine block surface

### Connection Elements

Following features can be seen as connection elements:

- **Bolt connections:**

They are often used under pretension. So, contact analysis is applied to define prestressed bolts (see page 68 for more details).

The thread coupling of bolts is of particular importance for short bolts (like in Fig. 30), because

any cut through the shaft of a short bolt will be warped under pretension.

- **Weld spot connections:**

Typical weld spot connections consist of an element at the weld point location, which is used to model the additional spotweld stiffness and an MPC condition which couples the element forces to the connected flanges. These flanges typically have incompatible meshes (see page 53 for more details on automated spotweld modeling).

A refined spotweld model is also available which shows improved stiffness representation and reduced sensitivity against different mesh sizes at the connected flanges (see page 93).

- **Sealing connections:**

For convenient modeling of sealings gasket elements are available which define the nonlinear behavior in a preferential direction by force-displacement curves.

Contact analysis is used to solve static sealing problems, where the force-displacement curves are handled as an internal contact.

In dynamic analysis, the typical frequency-dependent stiffness and damping of a sealing is modeled by spring-damper systems (see next list item and Fig. 115).

- **Spring-damper connections:**

In dynamic analysis, many joints have an influence on stiffness and damping. Such joints are modeled with spring-damper systems which also allows to model frequency dependent stiffness and damping properties (see Fig. 115).

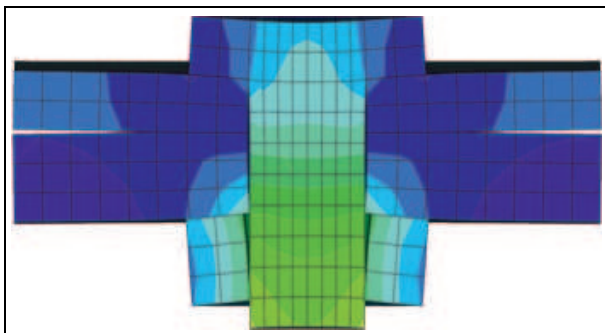


Figure 30: Short bolt under pretension

## Brake Squeal Analysis

Brake squeal is a known phenomenon since brakes are used, and despite intensive research for many decades there are still coming new cars to the market which are squealing so heavily that expensive warranty cases arise for the manufacturers. This holds not only for passenger cars but also for commercial vehicles, the same for rail cars or aircraft brakes or even bicycles. Also, not only disk brakes but also drum brakes are affected.

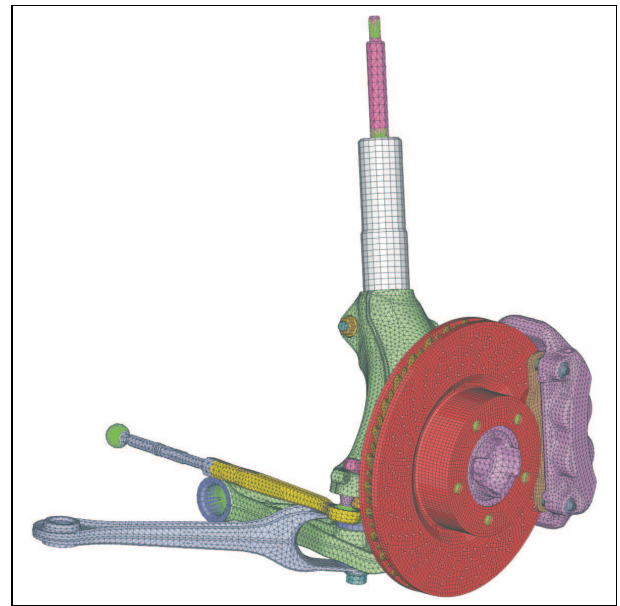


Figure 31: Brake model

by courtesy of Dr. Ing. h.c. F. Porsche AG in Stuttgart, Germany.

There is no lack of numerical approaches to make brake squeal computable but up to now the complexity of the phenomenon has prevented massive computations in this field due to very long computing times. As long as one set of parameters for one brake requires many hours of computing time, it is practically impossible to study geometrical modifications to get a configuration which does not exhibit squealing under all typical operating conditions.

Brake squeal is widely understood as friction induced dynamic instability. Therefore, two principal approaches are available: Transient analysis and complex eigenvalue analysis. Due to high computational effort for transient analyses obviously stability is more effectively studied by a complex mode analysis.

The analysis can be split up in following steps:



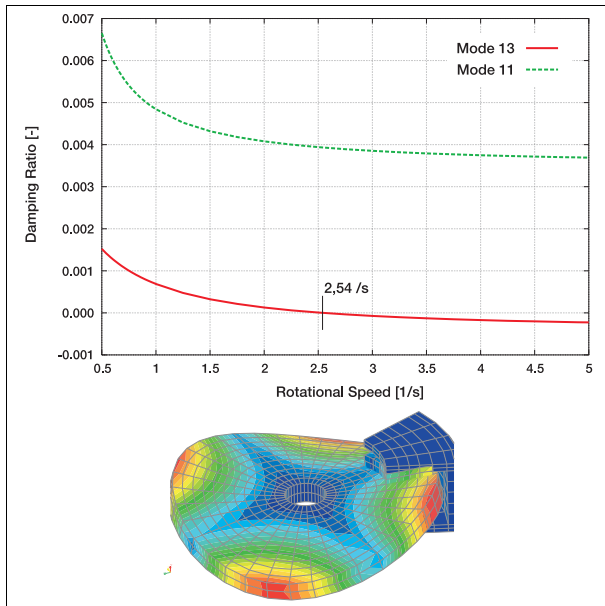


Figure 32: Simple brake model (1)  
with an unstable bending mode ( $m=2, n=1$ ) at 2,54 rps

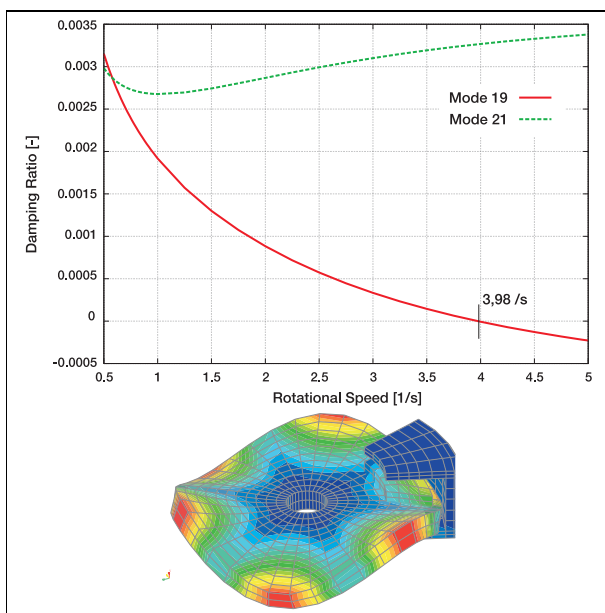


Figure 33: Simple brake model (2)  
with an unstable bending mode ( $m=3, n=1$ ) at 3,98 rps

- A linear static analysis with contact and friction under brake pressure and rotation. There, sliding between disk and brake pad can be prescribed by a rigid body motion to determine the sliding velocity.
- A real vibration mode analysis using the previously calculated contact status. This requires a linear model for the contact status which is achieved by *contact locking*.
- A complex mode analysis with additional fric-

tional and rotational terms. Gyroscopic and stiffness terms are taken into account which consider the disk as elastic structure in an inertial reference system. Additional stiffness and damping terms are derived from the frictional contact state perviously calculated in the contact analysis.

As usual instabilities are detected by a complex mode analysis if the real part of the complex eigenvalue becomes positive or the effective damping ratio becomes negative.

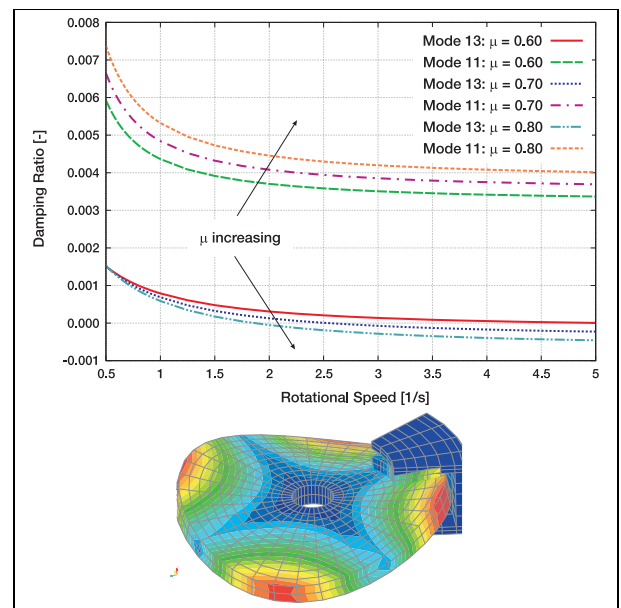


Figure 34: Simple brake model (3)  
with an unstable bending mode ( $m=2, n=1$ ) which becomes unstable at different rotational speeds dependent on the frictional coefficient.

A complex mode analysis is performed in one computing run for the full range of interesting rotational speeds. By this sweep all relevant points of instability are obtained for one set of brake parameters. Successive computing runs are then used to study parameter modifications in order to establish a stability map for all important influencing effects.

As an example a simple brake is used which exhibits various instabilities at different rotational speeds (see Fig. 32 and Fig. 33). Each of the diagrams shown was generated by one single computing run.

This analysis is repeated several times to get the influence of the frictional coefficient between brake disk and brake pad (see Fig. 34).

A similar study was made to get the influence of

Young's modulus of the brake pad. Figure 35 shows a corresponding example.

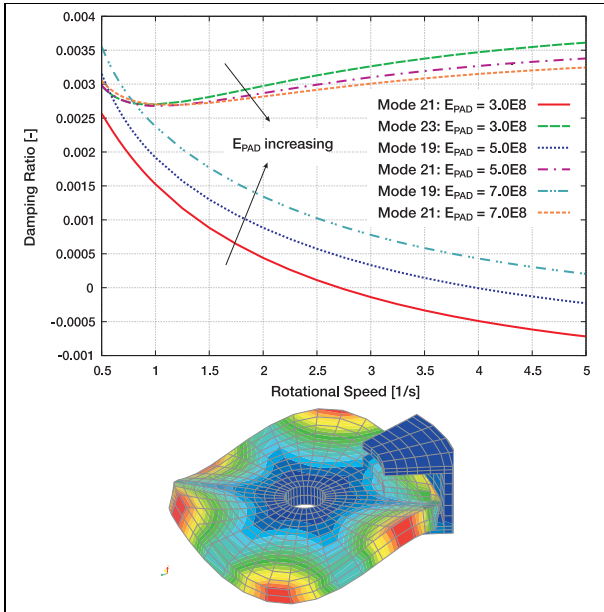


Figure 35: Simple brake model (4) with an unstable bending mode ( $m=3$ ,  $n=1$ ) which becomes unstable at different rotational speeds dependent on Young's modulus of the brake pad.

To illustrate the required run times for such analyses corresponding computing times are given for a large industrial model with the following characteristics:

300 000	Elements
500 000	Nodes
1.5 Million	Unknowns
158	Real eigenmodes
316	Complex eigenfrequencies
80	Rotational speeds

For this example the full elapsed run time with PERMAS Version 12 was 1 hour 10 minutes on a 4-CPU Itanium machine with 8 GB memory. The required disk space is about 90 GB.

On a 4-CPU quad core machine the run time was reduced to 19 minutes.

By such computing times an extensive parameter study of a brake will be possible in short time.

Such a parameter study can be used to generate a stability map of a brake (see Fig. 36). There, all unstable modes from complex eigenvalue analysis are collected for a large number of different parameter sets. This allows the identification of frequencies where squealing could occur. The information

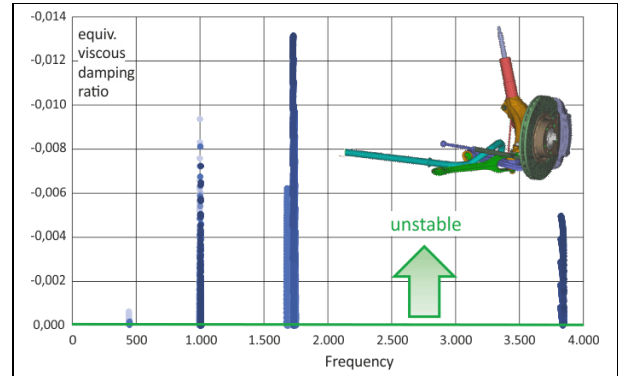


Figure 36: Stability map of a brake for 7110 parameter sets from rotational speeds, Young's modulus of disk, and frictional coefficient between disk and pad. The total run time for this stability map was 5 hours 12 minutes.

contained can be used to detect those parts of a brake which are candidates for modification to improve brake squeal behaviour.

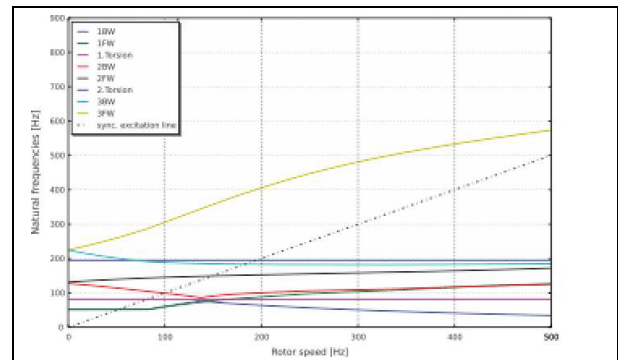


Figure 37: Campbell diagram for the evaluation of rotor dynamics

## Rotating Systems

The available static and dynamic analysis capabilities can be used to analyze rotating systems, which imply additional constraints to the solution.

Fig. 38 provides an overview on the analysis capabilities for rotating structures. Both co-rotating and inertial reference systems can be applied.

### Static Analysis

In a quasi-static analysis, which may include contact at the hub, the centrifugal forces due to rotation are taken into account. The reference system is

	Co-rotating reference system	Inertial reference system
Elastic rotor	arbitrary shape	axisymmetric
Several rotors	one single speed	different speeds
Bearing stiffness	isotropic	arbitrary, speed dep.
Stator	no	arbitrary shape
Static analysis	subcritical	subcritical
Dynamic analysis	sub- and overcritical	sub- and overcritical
Additional matrices	geometric stiffness, centrifugal stiffness, Coriolis matrix	geometric stiffness, convective stiffness, gyroscopic matrix
Modal damping +	material, viscous	speed dep. bearing, material, viscous in stator
Campbell diagram in one analysis	with mode tracking and stability evaluation	with mode tracking and stability evaluation
Modal and direct response	harmonic, periodic (steady-state), in time domain	harmonic, periodic (steady-state), in time domain
Sizing and shape optimization	for rotor	for rotor, stator, and bearing
Active damping	of rotor	of stator and bearing
Model reduction	of rotor	of rotor and stator

Figure 38: Rotor dynamics capabilities

co-rotating or inertial (with axisymmetric rotor). The static analysis is possible below critical speed.

In a linear analysis, the centrifugal stiffness and the geometric stiffness at the given rotational speed are taken into account. In a geometrically nonlinear analysis, an update of the centrifugal forces will take place.

### Dynamics

In order to get the relation between eigenfrequencies and rotational speed an automatic procedure is available (see Fig. 37 and page 75) which directly generates all values for a *Campbell* diagram.

For dynamics of rotating systems, the assumption is a linearized equation of motion with constant coefficients. A co-rotating or inertial reference system is taken. If rotating and non-rotating parts are present, the rotating part can be modeled as elastic body. The rotational speed is expected to be constant.

In the case of a coupling of rotating and non-rotating parts in a **co-rotating reference system**, no restrictions have to be observed for the rotating parts, but the non-rotating parts have to provide isotropic support to the rotor.

For such configuration, all direct and modal methods in time and frequency domain can be applied in the

subcritical and overcritical frequency range. During response analysis the *Coriolis matrix* is taken into account.

In the case of dynamics in an **inertial reference system**, no additional restrictions have to be observed for the non-rotating parts, but the rotating parts have to be axisymmetric.

Also for such configuration, all direct and modal methods in time and frequency domain can be applied taking into account the *gyroscopic matrix*. Modal methods remain applicable even for the overcritical range of rotational speeds.

To determine the *critical rotation speed* a Campbell diagram can be used. In the co-rotating reference system, the Campbell diagram will show zero eigenvalues at certain rotational speed.

*Rotation speed dependent stiffness and viscous damping* of rotor supports can be taken into account during complex eigenvalue analysis and for the generation of a Campbell diagram. This feature can be modeled by a special element (i.e. CONTROL6, siehe Abb. 112).

For dynamics modal *steady-state response* is of particular importance. There, the static stresses under centrifugal load are determined first. Then, with geometrical and centrifugal stiffness, the static displacements are derived. On the basis of real eigenvalue analysis, several modal frequency response analyses are performed for each harmonic. After back transformation to physical space, the results for all harmonics and the static case are superposed in the time domain (see page 77).

## Analysis of Machine Tools

For the development of machine tools, dynamic behaviour of the complete system is of utmost importance for the efficiency and precision of the machines. The complete system consists of structural parts, drives in different axes, and control. During machining the interaction between workpiece and tool generates cutting forces which may cause vibrations in the system. These vibrations have to be sufficiently damped by all system components. Finally, the machine tool has to provide high speed and high precision.

All dynamic analysis methods can also be used for machine tool analysis, like eigenmode and frequency response analysis, complex mode analysis, and time-history response analysis. In addition, optimization methods can be used to propose model modifications which improve the characteristics of the machine tool like weight, static response, and dynamic response

An example of a turning machine has been proposed by WZL Institute in Aachen (see Fig. 39). The parameter setting was supported by INDEX-Werke GmbH & Co. KG in Esslingen.

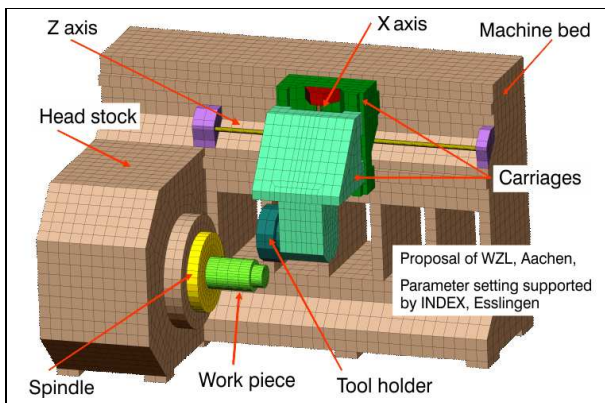


Figure 39: Simplified model of a turning machine as proposed by WZL in Aachen and supported by INDEX, Esslingen for setting suitable parameters

The following typical machine components were used to set-up the model:

- **Structural components:**  
Machine bed, carriages, and headstock are usually modeled by solid elements (see Fig. 39).
- **Guide rails:**  
They are part of structural components, but their proper connection is modeled by spring-damper combinations, where the spring and damper forces are connected to the solid structures taking incompatible meshes properly into account.
- **Ball screw drives:**  
They are modeled by beam elements. Their function is to transform a rotational motion of the drive to a translational motion of the carriage. This transformation is achieved by a proper MPC condition taking the diameter of the screw and the pitch of its thread into account.
- **Spindle with workpiece:**  
Spindle and workpiece are usually modeled with solid elements. The coupling to the headstock is done by a special element combined with a

suitable MPC condition. The special element is used to provide rotation speed dependent stiffness and damping values for the spindle support (see Fig. 112).

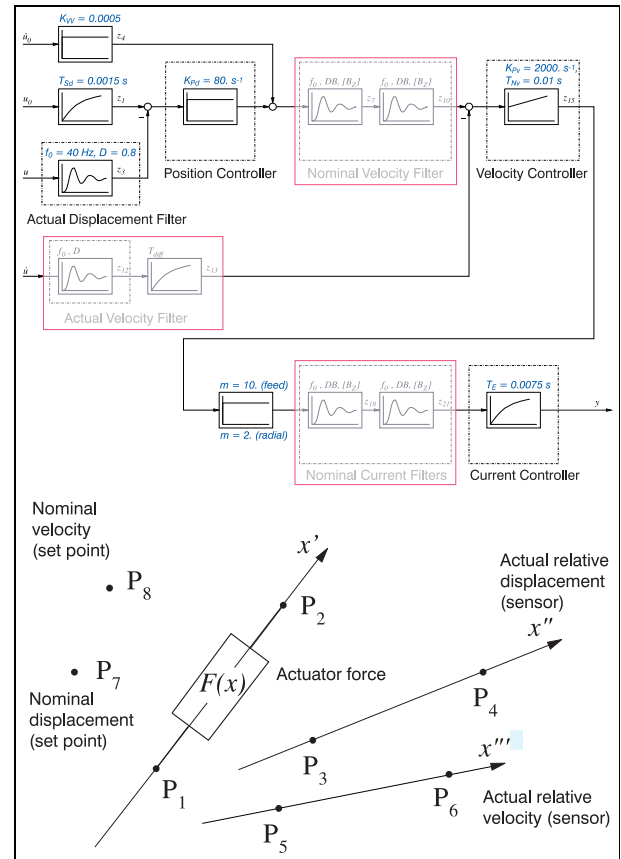


Figure 40: Block diagram of used controller and corresponding finite element representation

- **Control:**  
In order to include a controller in the model a finite element has been developed which represents a so-called *cascade controller* as shown in Fig. 40. This cascade controller is able to represent position and velocity control beside some filters and a current controller. This control element comprises the typical control steps used in machine tools and has been developed based on the experience of machine tool manufacturers.  
The actuator force is applied as a moment on the ball screw drives of both feed axes (Z axis and X axis). The sensors for relative displacements are located between machine bed and Z carriage and between Z carriage and X carriage respectively. The sensors for relative velocities (based on rotation speed sensors) are located at the ball screw drives.



The parameter settings have to reflect the actual controller which will be used for the machine tool. So, the parameters have to be provided to the analyst by the controller development (like other parameters, e.g. material). Then, the analyst will be able to support the machine design by full system analysis.

In the example model, there are two controllers, one for the translational motion parallel to the spindle axis (Z axis) and one for the motion in radial feed direction (X axis). The parameters of the controllers are almost the same except the parameter  $m$  which reflects the moved masses leading to higher value for the translational motion.

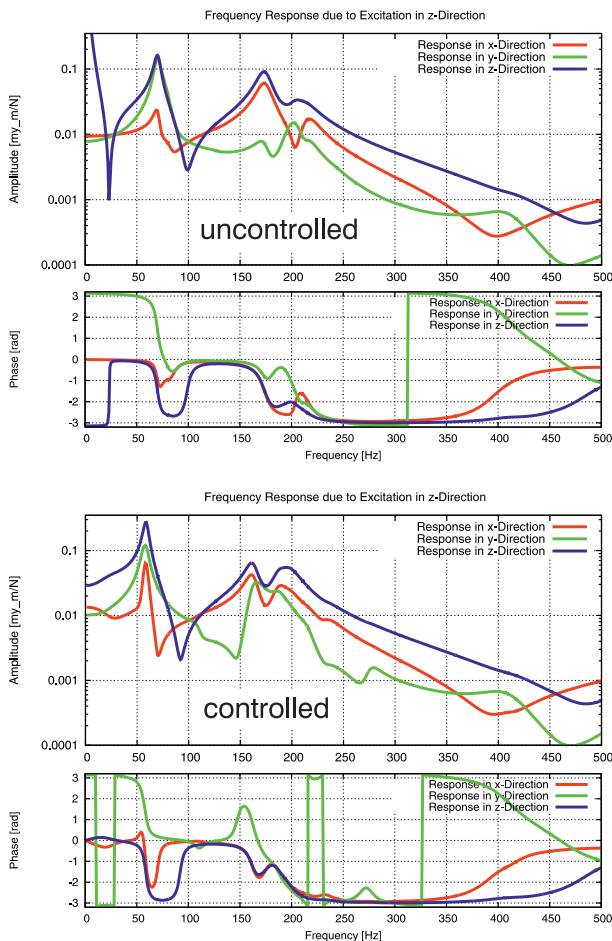


Figure 41: Frequency response between tool and workpiece due to excitation in Z direction for uncontrolled (above) and controlled (below) machine

Typical methods for dynamic analysis of machine tools include:

- **Eigenvalue analysis:**

Modal methods are often more efficient due to the reduction of the model size. But direct meth-

ods are also available.

- **Modal frequency response analysis:**

Fig. 41 shows the frequency response curves for amplitudes and phases between tool and workpiece for an excitation in Z direction under uncontrolled (above) and controlled (below) conditions.

- **Modal time-history response analysis:**

Fig. 42 shows the response to a sudden jump (above) and a smooth position change (below) of the tool in translational Z direction. It is worth mentioning that this calculation already takes the rotation of the spindle into account (at 2000 rpm).

In addition, it is possible to use nonlinear support conditions (like those for the spindle support) in modal time-history response analysis.

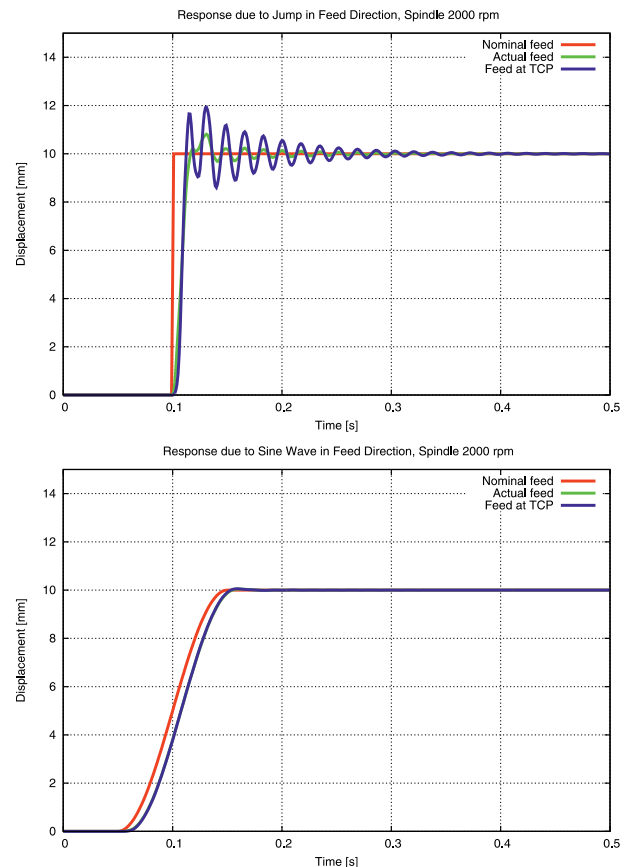


Figure 42: Time-history response due to excitation in Z direction for a sudden position jump (above) and a smooth position change (below) for the controlled machine

- **Complex mode analysis:**

The damping effect on the eigenfrequencies due to guides, spindle support, and control can be determined using complex eigenvalue analysis

(including gyroscopic effects).

- **Stability analysis of process:**

For the design of a machine tool, the prediction of instabilities during operation is of utmost importance to the manufacturer. The goal is to design a machine tool which works stable with high precision at high speed. The source of instability is the interaction between tool and workpiece, where the energy for selfexciting vibrations is coming from the rotating spindle.

In order to make the nonlinear turning process predictable a cutting force model has been developed. The cutting force model depends on the cutting speed, the cutting depth, and the feed per revolution. The cutting force model requires a number of coefficients which reflect the combination of a specific turning tool with the material of the workpiece. These coefficients have to be calibrated based on a few experiments with the same combination.

On this basis a stability chart and the related chatter frequencies can be derived (see. Fig. 43). There, one important effect of spindle rotation can be seen. The stability depends on the rotating direction of the spindle. Here, counterclockwise rotation allows for higher cutting depths than clockwise rotation of the spindle.

Fig. 43 also shows the time-history response for one point of operation (see red dot at 1300 rpm) which leads to stable behaviour for counterclockwise spindle rotation and to instable behaviour for clockwise spindle rotation.

Some examples of additional applications of optimization are:

- Topology optimization of structural parts for lighter machine tools taking control and dynamic conditions (like eigenfrequencies, frequency response) into account.
- Shape optimization for getting optimized positions of supports to ground.
- Sizing optimization of controller parameters to move eigenfrequencies and to improve frequency response.

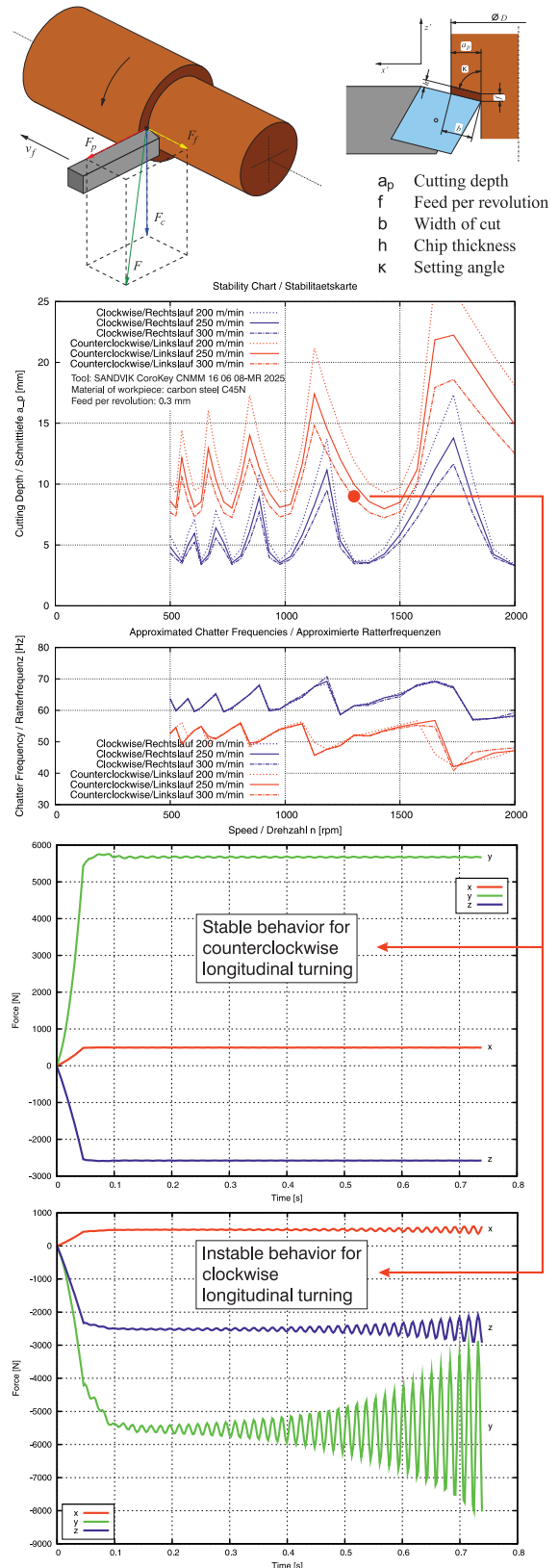


Figure 43: Stability chart for longitudinal turning with stable and instable behaviour under counterclockwise and clockwise turning.

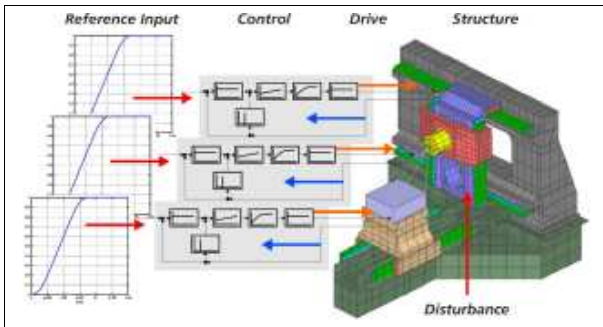


Figure 44: Principles of coupled analysis with control

## Actively Controlled Systems

In the past, the design of machine tools structures and their control have been made separate from each other. Today's drives have an essentially higher dynamics and this separation of the design is no longer appropriate due to a strong coupling between machine and control dynamics. So, the coupled simulation of structural dynamics and control becomes a basic requirement for a successful overall design.

Lightweight structures tend to show higher sensibility to vibrations. Sometimes, such vibrations can be damped by active control (see example of a composite box girder in Fig. 45). A dynamic analysis taking control into account provides a lot of information for the design of the structures like the required damping forces and the related power needed.

For the analysis of controlled structures, the following features are supported:

- Linear *control elements*:
  - Three-term (PID) controller,
  - Various cascade controllers.

Those control elements link a dynamic vibrational state (measured by a sensor) with a driving force using classical linear control parameters.

Linear controller elements are handled in the same way as any other element, i.e. they are defined by their topology together with some property values. Of course, there may be any number of controller elements in one model.

An eight node controller is also available as user programmable element where the user has to provide element stiffness and viscous damping matrices via subroutine.

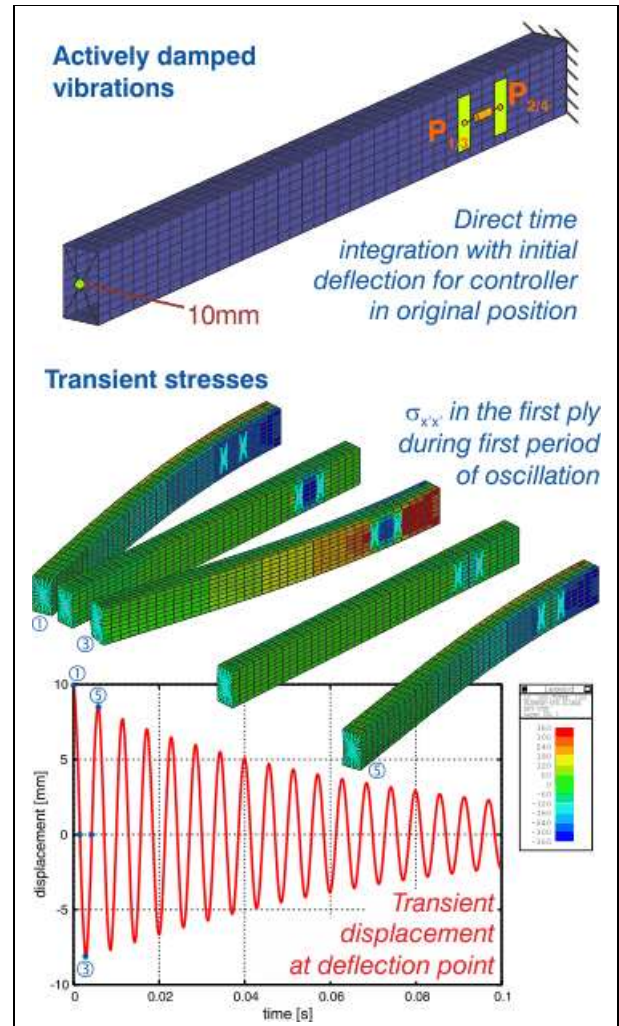


Figure 45: Active damping of a composite box girder

- Solution methods:
  - Linear controller elements may be used only in dynamic analyses, especially within the following solution methods:
    - Direct response analysis in frequency and time domain.
    - Modal response analysis in frequency and time domain, where the modal basis is enhanced by static mode shapes to represent the internal state variables of the controller elements.
    - Complex eigenvalue analysis to judge the effect of controllers on the dynamic behavior.
- Additional static mode shapes:
  - Beside the representation of internal state variables in modal space, static mode shapes added to the modal basis may be used to improve the accuracy of results of solution meth-

ods in modal space (see page 78).

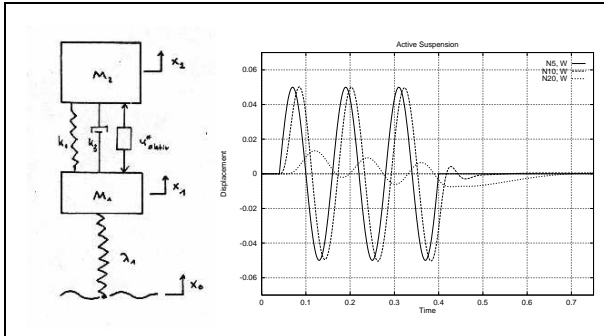


Figure 46: Active suspension unit  
 $u_{aktiv}^*$  between wheel M1 and body M2 as a function  $f(x_0, x_1, x_2, \dot{x}_1, \dot{x}_2)$  with a harmonic base excitation. The body does not show any increase of the amplitude.

- Nonlinear control elements:  
In addition, for nonlinear control additional control elements are available, where the dependency of the controller force on any result value in the model may be described by a general function, e.g. by a FORTRAN or C subroutine. Due to nonlinearities, the application of these elements is restricted to modal or direct transient response analysis.

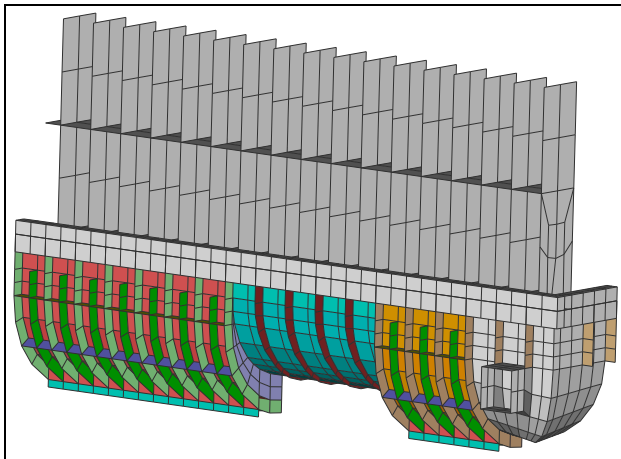


Figure 47: Optimization of a water box with 18 design variables and 19 stochastic basic variables

		Basic	Optimized	One Step
Mass:	$M$	$4.09 \cdot 10^{-4}$	$3.15 \cdot 10^{-4}$	$3.21 \cdot 10^{-4}$
Failure probability:	$P_f$	$4.78 \cdot 10^{-6}$	$6.80 \cdot 10^{-3}$	$1.55 \cdot 10^{-6}$
Failure rate (one of):		209205	147	645161

## Robust Optimum Design

In order to achieve a **robust design**, it is not sufficient to perform a simple optimization:

- Optimization often leads to reduced safety margins.
- The optimized design may have other critical parameters than the initial design.
- A “reliable” optimum may be different than a deterministic one.

The proposed solution is a combination of optimization and reliability analysis. For this combination two different approaches are available:

- Two step approach**
  - Basic design
    - FE-Analysis (e.g. static analysis)
    - Reliability analysis
  - Optimization taking into account reliability
    - Optimization
    - Reliability analysis of optimized design
    - If not sufficient:
      - Modify the design model according to reliability results
      - Repeat the optimization and the reliability loop
- One step approach**
  - Combined optimization and reliability analysis
  - Reliability as design constraint in optimization

For the one-step approach, the interplay of design variables in optimization and basic variables in reliability analysis is as follows:

- Design Variables
  - define the design state of the structure
  - may be modified by the optimizer
  - may be assigned to one of the following types:
    - Deterministic design variable
    - Deterministic mean value of a stochastic design variable
- Basic Variables
  - Basic variables define the stochastic properties of the problem
  - Following types of basic variables are possible:
    - Stochastic properties of the structure
    - Stochastic design variable with deterministic mean value
    - Load factors



- \* **Limit state function** parameter
- \* Parameter of another basic variable

In the one step approach, there two different states belong to each design point:

- Design state
- Limit state

For each state one FE-analysis is necessary. Therefore, for each step during optimization at least two FE-analyses must be performed.

The design state is the actual optimization state. It is given by:

- Actual values of design variables
- Mean values of basic variables

The objective function is evaluated for the design state. The design constraints are evaluated for the design state. The final design state must fulfil the design constraints.



Figure 48: Optimization of PROTEUS satellite with 28 design variables and 30 stochastic basic variables (Alcatel Space S.A.)

		Basic	Optimized	One Step
Mass:	$M$	324.8	308.9	312.5
Maximum stress:	$\sigma$	$9.6 \cdot 10^7$	$1.2 \cdot 10^8$	$5.5 \cdot 10^6$
Failure probability:	$P_f$	$8.2 \cdot 10^{-7}$	$4.2 \cdot 10^{-5}$	$1.0 \cdot 10^{-6}$
Failure rate (one of)		1.2 Mio	23809	1 Mio.

The limit state describes for a given design state the corresponding failure state. It is given by:

- Actual values of design variables
- Actual values of basic variables

The limit state function is evaluated for the limit state. The design constraints are meaningless for

the limit state, e.g. the limit values for the limit state function and the limit values for the design constraints are different.

The combined analysis gives the following results:

- **Final design state**
  - Objective function value
  - Design variable values
  - Elasticities of design variables with respect to objective function
  - Probability of failure
  - Values of active constraints
- **Final limit state**
  - Basic variable values
  - Parameter sensitivities of the limit state function
- **Always available:**
  - Selected data for each iteration



## VisPER



Figure 49: The logo of VisPER

### The VisPER History

PERMAS as a software for numerical analysis provides many functions which are not sufficiently supported by available pre-processors. So, for many years the developers of PERMAS have looked for a suitable *GUI* to support PERMAS as good as possible and to provide a more effective model description.

After three years of development a new graphical user interface for PERMAS has been announced on the PERMAS Users' Conference early April 2008 in Stuttgart. This product is named VisPER (for Visual PERMAS). The current version is VisPER Version 4 which is available with PERMAS Version 15.

VisPER was developed in co-operation by science+computing ag (s+c), Tuebingen ([www.science-computing.de](http://www.science-computing.de)) and INTES. The graphical functionality was developed by s+c whereas INTES supplied the data base functions, the interfaces and the computational functions.

In 2012, INTES acquired all remaining rights of VisPER from science+computing ag and employed the VisPER development staff from s+c. So, a fully integrated development of VisPER and PERMAS could be achieved.

### VisPER – A Short Introduction

VisPER (Visual PERMAS) is a GUI based model editor. It is used to complete finite element models for specific applications with PERMAS. To this end VisPER fills efficiently the gap between FE models generated by a mesher and PERMAS models which are ready to run (see Fig. 50).

VisPER can also be used as post-processor for PERMAS. In particular, post-processing is provided for those functions, where VisPER is used as model editor.

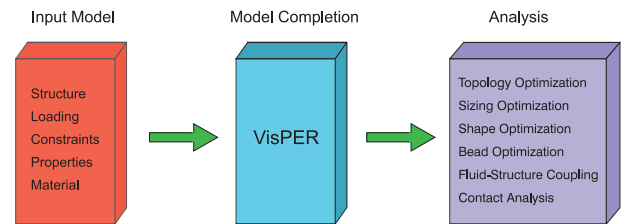


Figure 50: Model editing with VisPER

In order to demonstrate the potential of enhanced model editing a number of special functions are supported by VisPER:

- Topology optimization,
- Sizing optimization,
- Shape optimization (with free-form optimization),
- Bead optimization,
- Fluid-structure coupling,
- Contact analysis (with bolt pretension).

The use of VisPER is justified, if

- a FE analysis of a supported type is to be performed and a mesh is already available,
- a FE model is to be inspected for checking purposes (e.g. loads and boundary conditions),
- a substructure model has to be created and verified,
- results of a FE analysis with PERMAS should be post-processed.

Some of the advantages related to the use of VisPER are:

- VisPER provides an easy and fast way to complete a model,
- The model completion represents the logical structure of a PERMAS model. This is aimed at a reliable and nearly error-free specification process and shortens the way to a correct model,
- Due to a recording facility and a fully integrated scripting capability (*Python*) VisPER is highly customizable,
- The model data generated in VisPER can be exported separately and used in PERMAS together with the already existing model data,
- VisPER uses the same data structures and input facilities than PERMAS. Hence, models in

both programs are identical without translation or interpretation.

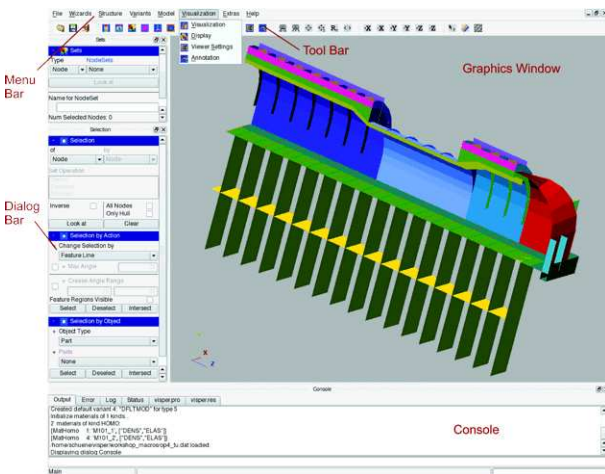


Figure 51: Main Window of VisPER

Fig. 51 shows the main window of VisPER. Each function can be accessed via pull down menus. Shortcuts are predefined for many functions which may be changed or new shortcuts may be added. The tool bar provides a fast access to frequently used menu functions. A dialog bar provides a range of commands corresponding to a selected item. The console provides a feedback of VisPER to the user's actions (including comments and error messages). The console and any dialog bar can be hidden to maximize the graphics window. For frequent use they may be pulled out of the main window to see them permanently.

## VisPER-BAS – Basic Module

This module comprises the complete VisPER infrastructure, the graphical user interface, and all basic functions for pre- and post-processing.

The infrastructure includes:

- configuration of VisPER with regard to a site, a user, and the integration of user-defined documentation (like *tooltips*),
- import and export of files (like model data and results),
- generation and use of macros,
- creation and use of PERMAS user control files (i.e. UCI).

The graphical user interface includes:

- the menu elements like menu bar, tool bar, dialog bar, wizards, consol panel, and information panel,
- the GUI elements for color definition, slider animation, and font selection,
- the control of mouse, space mouse, viewer buttons, and camera interaction,
- the interactive manipulations with the mouse,
- the measuring of a distance between nodes or elements,
- the definition and use of shortcuts.

The basic functions for pre- and post-processing include:

- functions on model categories like element quality, subcomponents, sets, surfaces, MPC, local systems, nodes, elements, and grouping.
- functions on variants like single point constraints, geometrical element data, and element properties,
- functions on modal data like search and presentation of model information, model configuration with definition of situations, and material definition,
- visualization functions on scalar results, vector results, displacements, and XY data. In addition, a mesh mapper supports the 3D-visualization of points (by a cube) and lines (by a pipe with polygonal cross sections). Generation of snapshots and videos is also supported,
- functions for clipping planes, show/hide elements, and annotations.
- Complex mode shapes can be visualized taking the phase angle into account.
- Direct creation of animated GIF images for animations.
- Extended selection of components for visibility by show or hide elements.
- Gasket pressure-closure curves can be plotted now.

A *RBM Assistant* (Rigid Body Mode Assistant) is available which provides an efficient workflow to generate elastic supports for static models. In particular for contact models, compensation springs can be comfortably generated by this assistant (see page 68).

VisPER also provides some useful tools to support the cooperation with PERMAS, e.g.:

- A *message tree* is provided to summarize diagnostics and to provide additional information



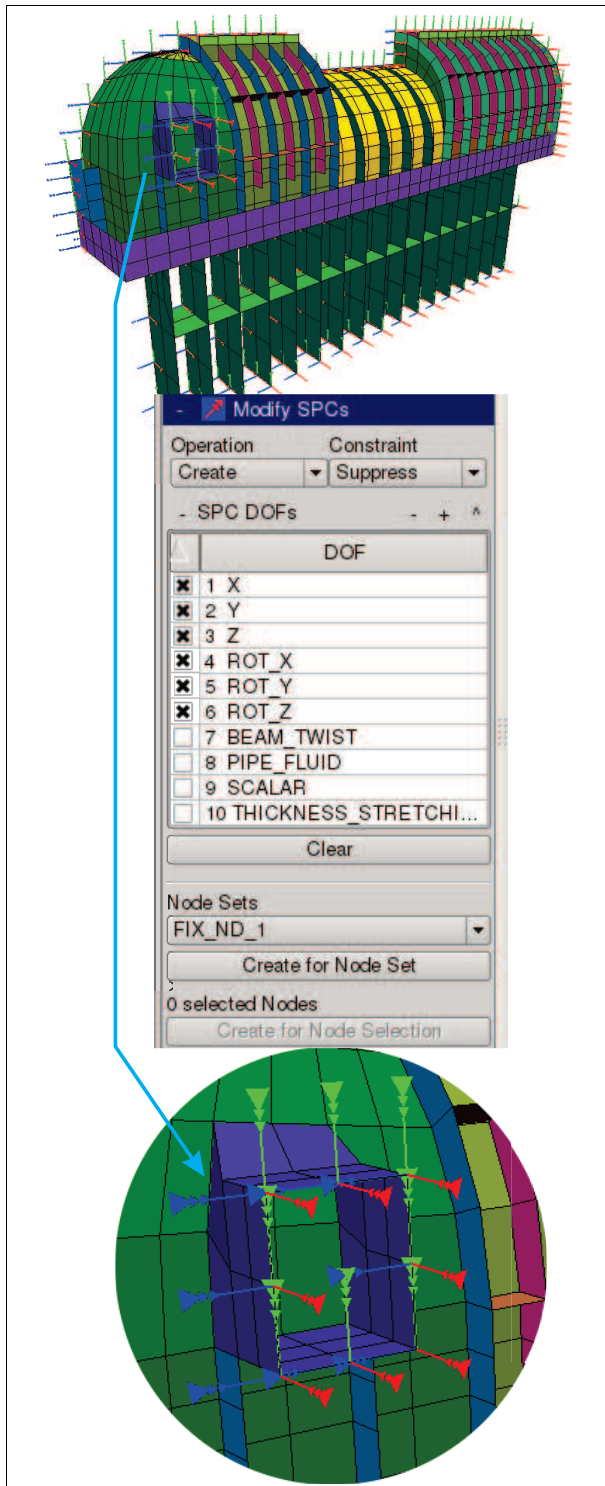


Figure 52: Specifying single point constraints

(like tool tips). The dialog provides parsing of the PERMAS protocol file. There is a direct relation to model entities mentioned in the diagnostics.

- Information about the model is provided by a tree view of all processed input files.
- VisPER can be configured to submit a PERMAS

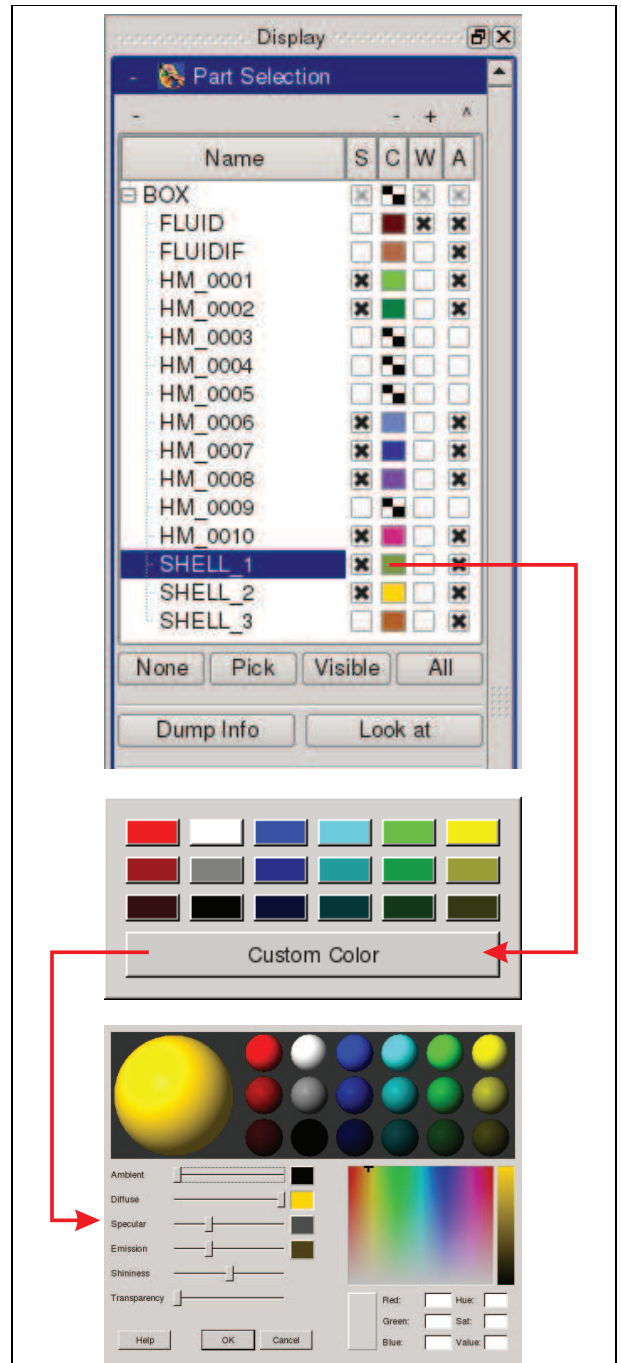


Figure 53: Graphics entities for parts and element groups (select S=Show, C=Colour, W=Wireframe, A=Active).

batch job.

- Imported results and the related post-processing model can be exported to other formats (like MEDINA or HyperView).

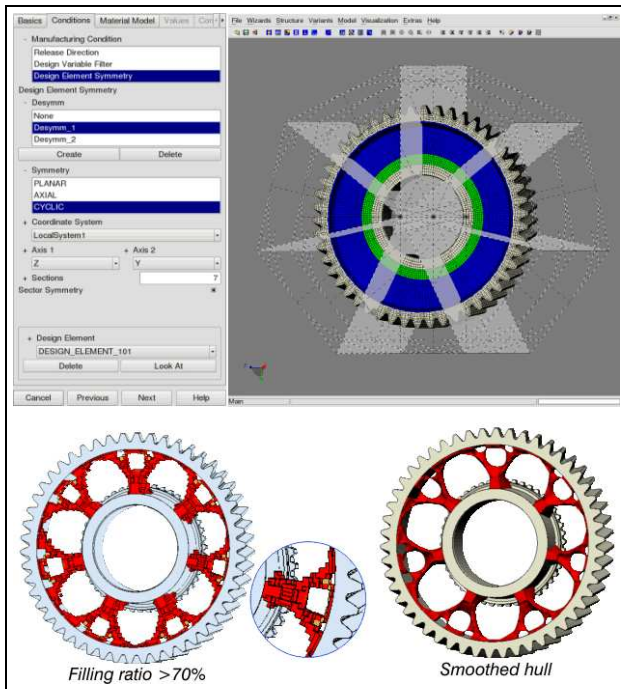


Figure 54: Topology optimization of a gearwheel  
Definition of cyclic symmetry and release directions, max. stress at interface between design space and gear rim, max. allowed weight, minimum compliance as objective function (by courtesy of Daimler AG, Stuttgart).

## VisPER-TOP – Topology Optimization

This module provides a *wizard* supporting the set-up of an optimization model for topology optimization tasks. The details of topology optimization in PERMAS can be found on page 86.

The wizard provides guidance through the optimization modeling by the following steps:

1. The basics contain the specification of the design space and the related design elements. For the filling ratio a number of conditions can be specified like initial fill, modification limit, and minimum and maximum values.
2. The *manufacturing constraints* contain the definition of release directions, filters for minimum and maximum member size, and symmetry conditions (see Fig. 54 as an example). A special function allows the specification of a *fixed mold parting line* which separates opposing release directions.
3. Design constraints can be specified dependent on certain analysis types for various result quan-

ties (like compliance, weight, displacement, frequency). One of the constraints is selected as design objective.

In addition, post-processing of topology optimization results is supported by VisPER, too. *Hull generation*, *smoothing* and *polygon reduction* can be performed beside the export of smoothed surface (see page 88 for details of these results). Fig. 54 also shows the smoothed hull of the gear wheel body.

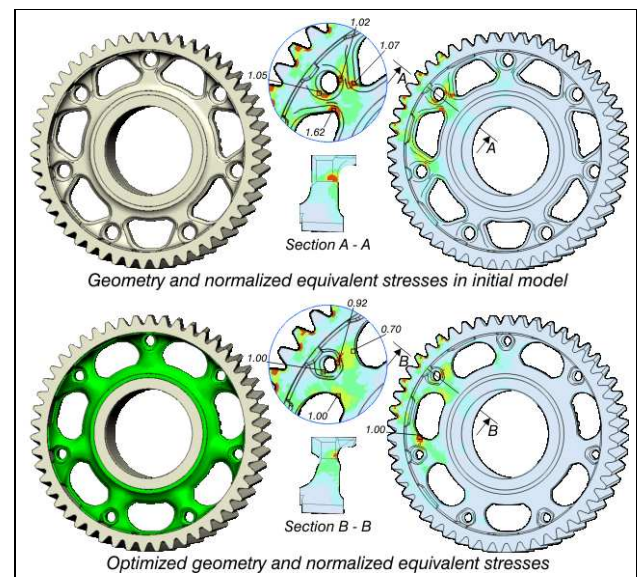


Figure 55: Shape optimization of a gearwheel  
Definition of cyclic symmetry, max. stress, minimum weight as objective function (by courtesy of Daimler AG, Stuttgart).

## VisPER-OPT – Design Optimization

This module provides three *wizards* supporting the set-up of an optimization model for design optimization tasks, one wizard for parametric shape optimization, one wizard for sizing optimization, and one for non-parametric free-form optimization. The details of design optimization in PERMAS can be found on page 83.

Design optimization comprises the following tasks:

- Sizing optimization which modifies the properties of elements (geometrical properties or material properties).
- Shape optimization which modifies the node coordinates of a model.

- Bead design which modifies the coordinates of a shell model to achieve certain stiffness targets by bead generation.

For *sizing optimization*, the wizard provides guidance through the optimization modeling by the following steps:

1. The global selection of properties and their relation to design elements including design and modification limits. The design elements are specified by element sets.
2. The basics contain the specification of the design space and the related design elements.
3. The detailed properties can be specified for each design variable.
4. Design constraints can be specified dependent on certain analysis types for various result quantities (like compliance, weight, displacement, frequency, temperature, contact pressure). One of the constraints is selected as design objective.

For *shape optimization*, the wizard provides guidance through the optimization modeling by the following steps:

1. The basics contain the specification of the design space and the related design elements.
2. The conditions contain the definition of restraints for shape changes, restrictions on bead design, and symmetry conditions.
3. Definition of *shape basis vectors* (SBV) as design variables and a number of conditions like initial values, modification limit, and minimum and maximum values. The shape basis vectors are calculated by PERMAS and provided for visualization. See Fig. 56 as example for the specification of shape basis vectors by *morphing*.
4. Design constraints can be specified dependent on certain analysis types for various result quantities (like compliance, weight, displacement, frequency, temperature, contact pressure). One of the constraints is selected as design objective.

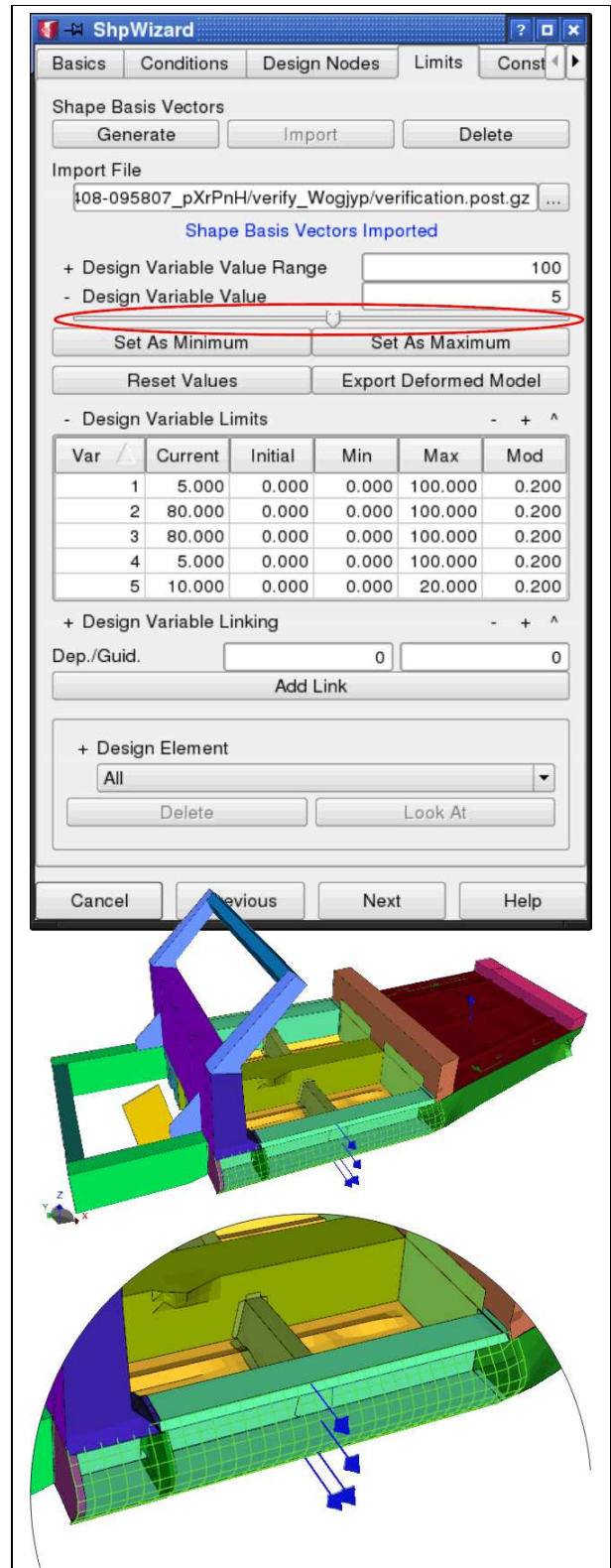


Figure 56: Definition of shape basis vectors of a car sill. The shape changes can be visualized by moving the slider in the menu.

A wizard for *free-form optimization* is available for a



parameter-free shape optimization to reduce stress concentrations (see Fig. 17).

In addition, post-processing of design optimization results is supported by VisPER, too:

- Parameters of the optimization process are available as XY plots for objective function history, constraint history, design variable history, and max. constraint violation history. The visualization of history plots allows a direct link between a touched curve in the plot and a highlighted area in the structure, e.g. a design variable history shows the related design elements.
- Resulting shell thicknesses can be presented.
- Shape changes can be visualized by showing the model with new coordinates.
- A **hull generation** is supported also for the new shape after shape optimization.

## VisPER-FS – Fluid-Structure Coupling

This module mainly provides a **wizard** supporting the semi-automatic fluid meshing of a cavity, where the surrounding structure is given as input. The wizard supports the following steps, where the numbers correspond to the numbers in Figs. 57 and 58:

1. A recommended mesh size is calculated based on a frequency range for subsequent analysis.
2. **Hole detection** is performed automatically. Only those holes are detected which are larger than the previously specified mesh size.
3. Due to difficult topological situations, some holes cannot be detected automatically. Then, undetected holes can be specified in addition.
4. Hole meshing is performed automatically. There, elements are used which do not introduce any stiffness or mass. They are just used to specify the topology of the hole and to limit the subsequent cavity meshing.

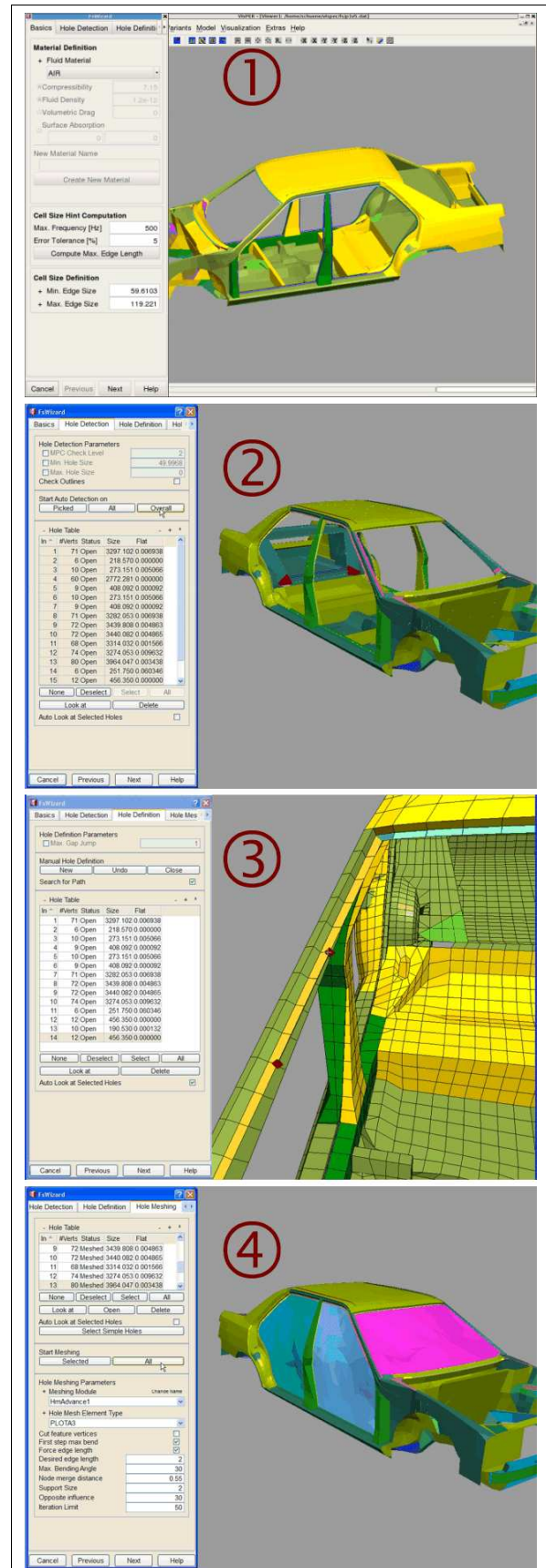


Figure 57: Steps 1 to 4 of the FS wizard



5. Seed point definition is made by the user. This specifies the starting point for the cavity meshing.
6. Cavity meshing is performed automatically. *Voxel* meshing is used and leads to a hexahedra mesh. The hexahedra mesh provides a low number of pressure degrees of freedom compared to tetrahedra meshes (where the element orientation influences the result, too).
7. When meshing is finished, there are some elements penetrating the hull of the cavity. These penetrations are resolved by a relaxation process which is performed automatically.
8. The last step automatically generates the coupling elements between fluid and structure. Their nodes have both pressure and displacement degrees of freedom.

Details of the fluid-structure acoustic simulation can be found on page 79.

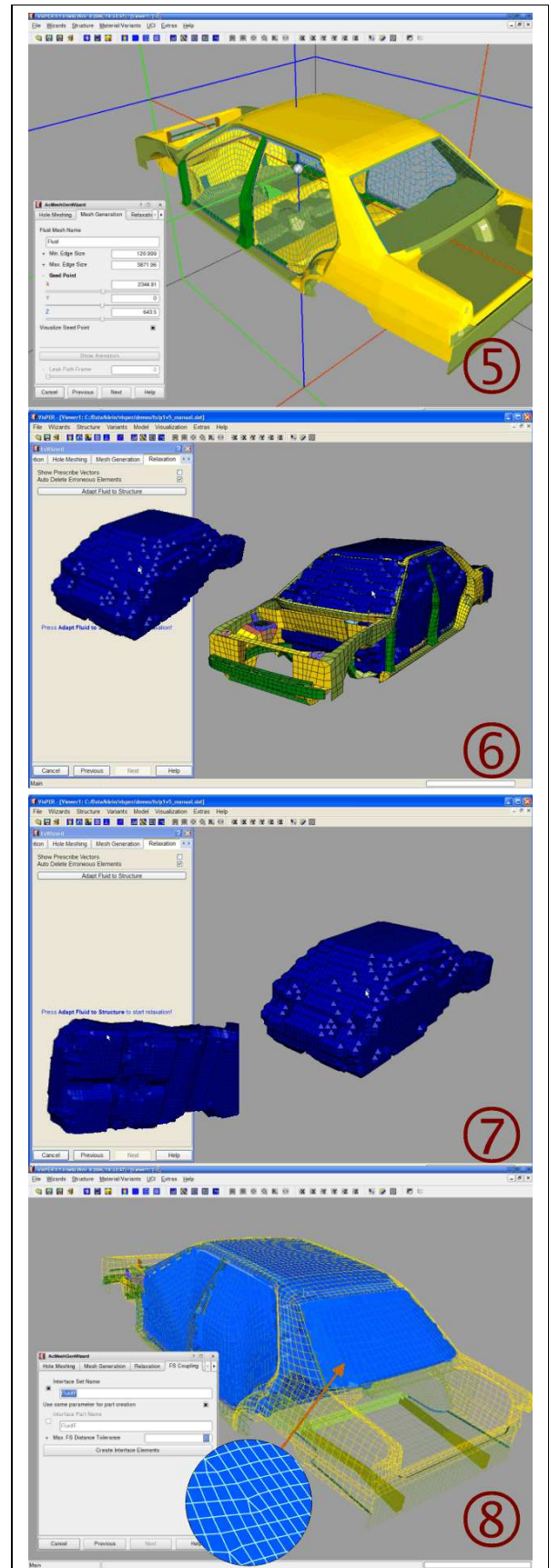


Figure 58: Steps 5 to 8 of the FS wizard

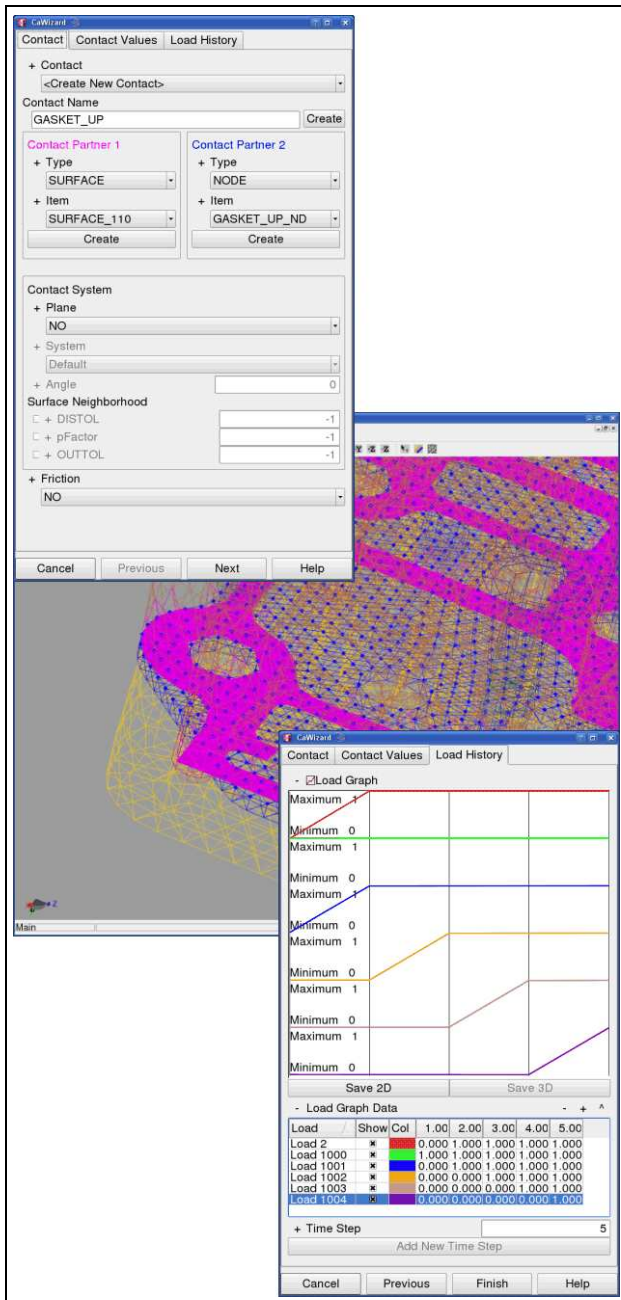


Figure 59: Contact modeling  
Contact topology and load history definition

## VisPER-CA – Contact Analysis

This module provides two *wizards* supporting the set-up of a contact model and bolt pretension. The details of contact analysis in PERMAS can be found on page 66.

The *contact wizard* provides guidance through the contact modeling by the following steps and pro-

vides a graphical check of already existing contact definitions:

- Definition of the contact topology with node-to-node, surface-to-node, or surface-to-surface relationship.
- Definition of the contact details like initial gap width and frictional coefficients. This information is specified as load case dependent data.
- Definition of a load history, which specifies the activation and deactivation for all loadcases in a contact analysis (or in a static analysis in general). See the example in Fig. 59.
- Checking of contact definitions by highlighting while the model is shown by contour lines.

A *pretension wizard* allows an easy description of complex bolt definitions according to PERMAS pretension technology:

- Bolts are detected automatically and grouped with respect to their length and direction.
- The coupling surfaces in the thread area are proposed and can be adjusted or confirmed after visual checking.

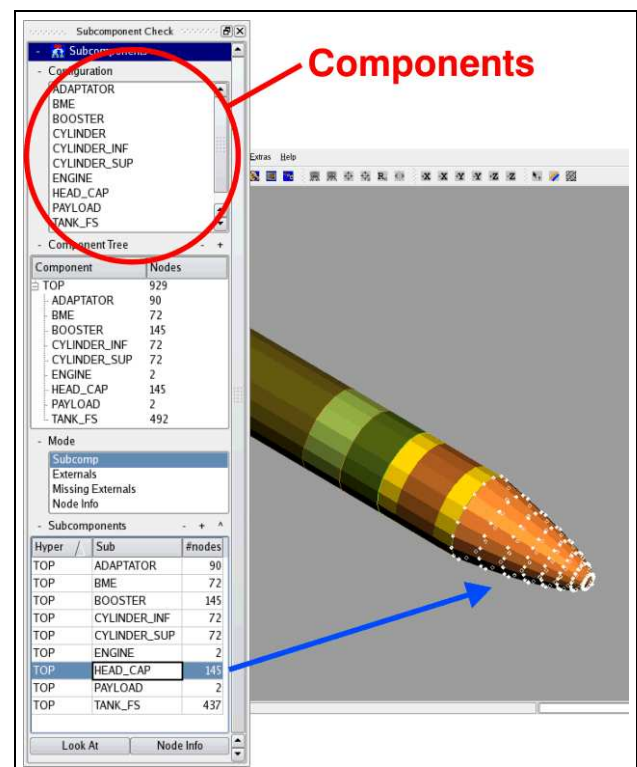


Figure 60: Navigation in the component tree  
of a substructure model

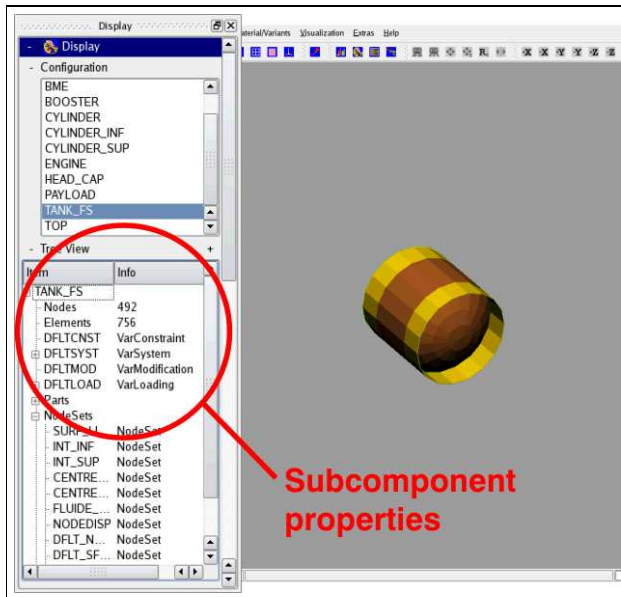


Figure 61: Visualization of components of a substructure model

## Substructuring

A *substructure wizard* has been provided to facilitate the set-up of substructure models. This function is part of module VBAS (see page 40).

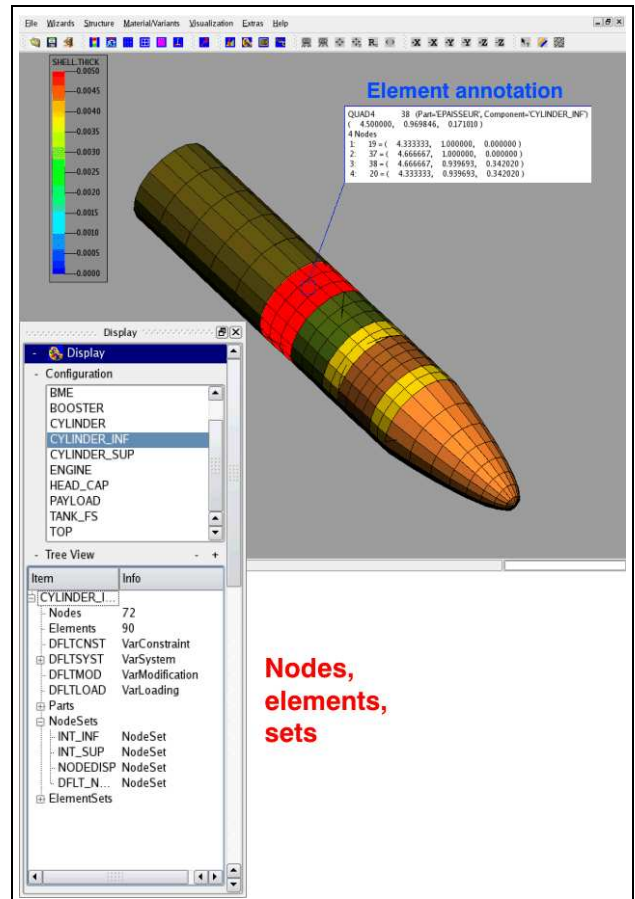


Figure 63: Checking element properties in a substructure model

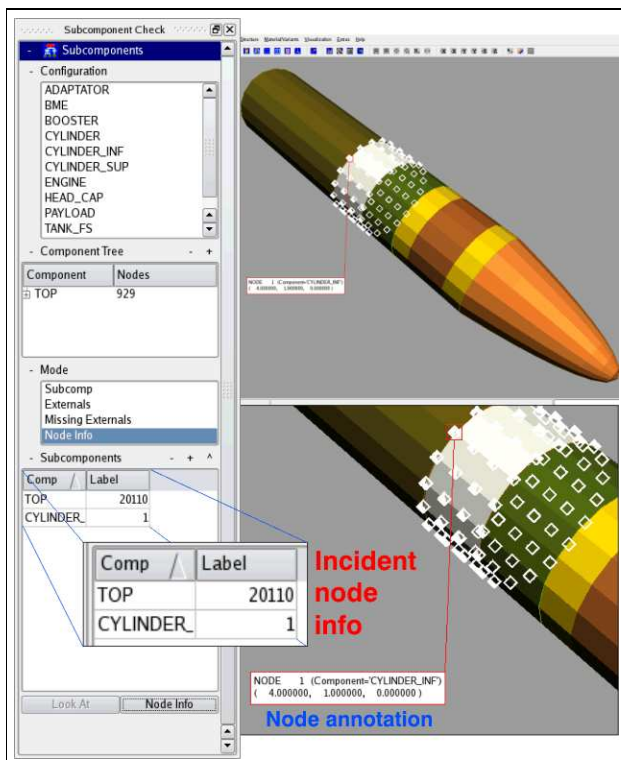


Figure 62: Visualization of coupling nodes of a substructure model

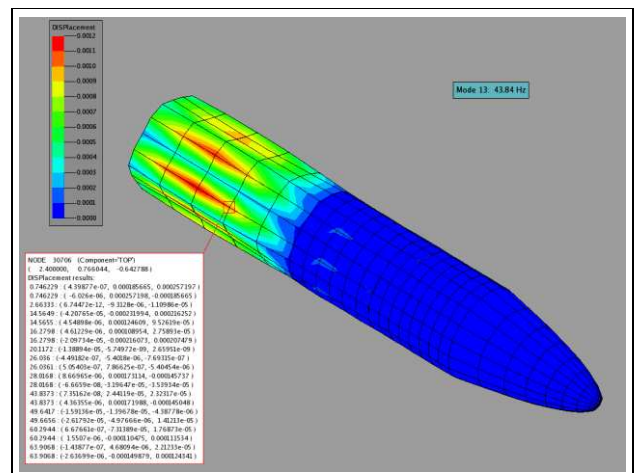


Figure 64: Display of mode shapes on full substructure model

Because VisPER uses the data structures of PERMAS, substructure models can be directly imported. Substructuring is explained in more detail on page 49.

The verification of substructure models includes the



following points:

- Navigation through the component tree (see Fig. 60).
- Visualization of single components (see Fig. 61).
- Visualization of coupling nodes (see Fig. 62).
- Check of element properties (see Fig. 63).
- Display of mode shapes on full structure (see Fig. 64).

## Evaluation of Spotwelds

A special post-processing feature has been included in VisPER providing the evaluation of spotweld forces. This function is part of module VBAS (see page 40).

Certain threshold values can be specified to show green, yellow, and red colour for certain levels of spotweld forces. This can be used as a light signal to mark uncritical, problematic, and critical force values. A full evaluation of spotwelds is possible together with the stresses in the flanges. This is combined with a spotweld representation as half spheres (see Fig. 65). In this way a fast overview on a high number of spotwelds in a large structure is provided (e.g. of a car body). An automatic detection of critical model areas enables a fast navigation in the model.

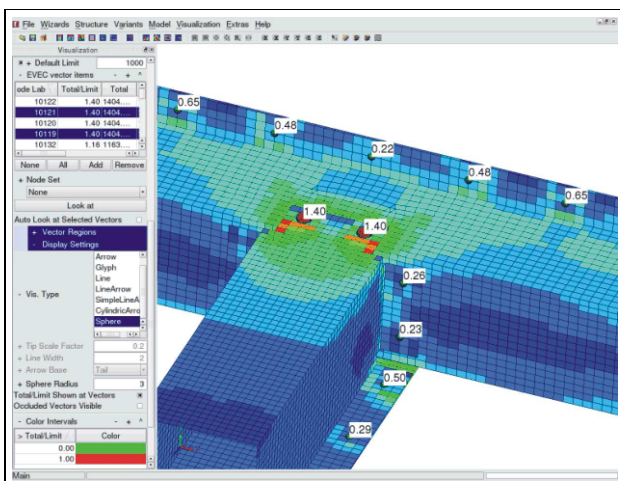


Figure 65: Evaluation of spotweld forces together with element stresses



## PERMAS Basic Functions

### Substructuring

PERMAS allows to decompose a model into substructures – the so-called **Components**. Like single elements in a FEA-model, these Components may be inserted into a superior structure – the **Configuration**.

- The number of Components is not limited and **each Component** may be **arbitrarily large and complex**.
- Components and Configurations are identified by user defined names.
- Each Component has its own name index for element-, node-IDs, etc.
- Each Configuration may consist of an arbitrary number of Component levels and each level may contain elements, loads and constraints. The specification of the coupling degrees of freedom in each component (the so-called 'external' degrees of freedom) allows the **automatic assembly** of the complete Configuration. The *condensation* of the components is performed using *Guyan's* reduction.
- In addition to the *static condensation* using *Guyan's* reduction a *dynamic condensation* using the *Craig-Bampton* method or the *Mixed-Boundary Craig-Bampton* method is available.
- Specific reordering concepts in conjunction with explicit and iterative condensation schemes and automatic selection of algorithms provide **highest solution efficiency**.

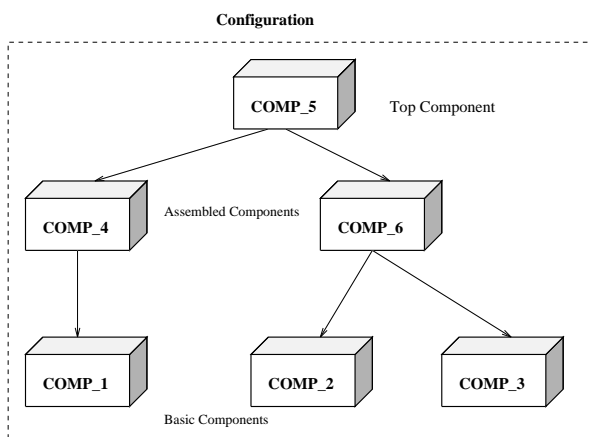


Figure 66: Substructuring in PERMAS

- The data base can hold an arbitrary number of Configurations.

- Multiple Configurations may share the same Components.
- Components may be extracted and saved for future substructuring either by their model description or by the corresponding condensed matrix models (see section Matrix Models, page 61).
- Such matrix models may be forwarded to customers and suppliers in place of the real geometric models.
- Substructuring permits the separated modeling and verification for all parts of the structure, prior to the final assembly.
- Single FEA models from distinct modeling sources, can be easily combined.
- Areas with design variations or nonlinear properties may be assembled into separate Components, thus concentrating modifications and iterations on this Component only. This will measurably cut computation time and resources.

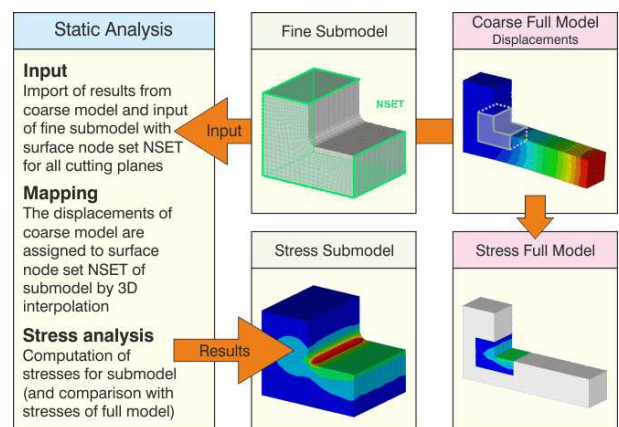


Figure 67: Submodeling in static analysis

### Submodeling

This feature supports the use of previously calculated results from a coarse (global) model as boundary conditions for a refined mesh of a part of the model. This enables e.g. a subsequent more precise analysis of stresses (see Fig. 67).

In a static analysis the displacements at the boundary of the refined part are taken from the full model as prescribed values.

For temperature fields, the interpolation on the side of the fine mesh is usually done in 3D (see Fig. 68).

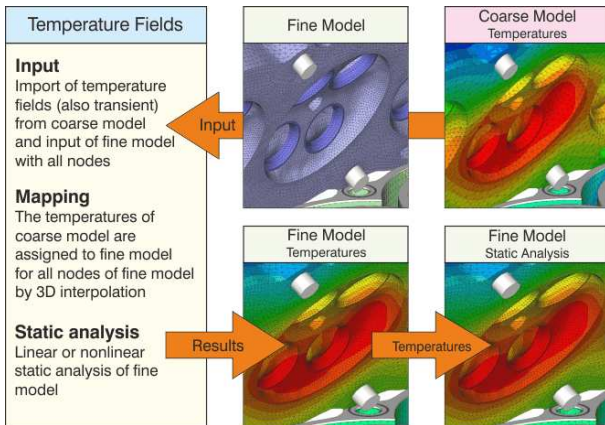


Figure 68: Mapping of temperature fields

Sometimes temperature fields have also to be extrapolated to nodes which do not yet have temperatures (e.g. additional parts like gaskets or bolts in engine analysis). This extrapolation can be performed by the mapping or by an additional temperature field analysis (or both as explained by the example in Fig. 69).

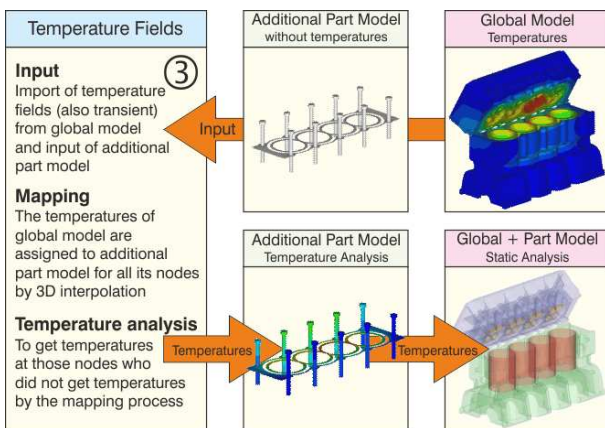


Figure 69: Mapping with extrapolation of temperature fields

## Variant Analysis

PERMAS offers an easy way for examining different variants of one FEA model. Variable model properties are held within the so-called **Variants**:

- **System Variant:**
  - material assignment of elements,
  - element properties (thickness, cross section etc.),
  - element local coordinate systems.
- **Constraint Variant:**
  - suppressed degrees of freedom,
  - prescribed degrees of freedom,
  - contact definitions,
  - coefficients of general kinematic constraints,
  - local coordinate systems for the degrees of freedom at each node.
- **Loading Variant:**
  - Dynamic loads and an arbitrary number of static load cases or combinations hereof.
- **Result Variant:**
  - For more detailed specification of required results, like
    - \* load pattern combination rules,
    - \* list of excitation frequencies for frequency response,
    - \* load steps where results are requested.
- **Modification Variant:**
  - specification of design model for sensitivity analysis and optimization.

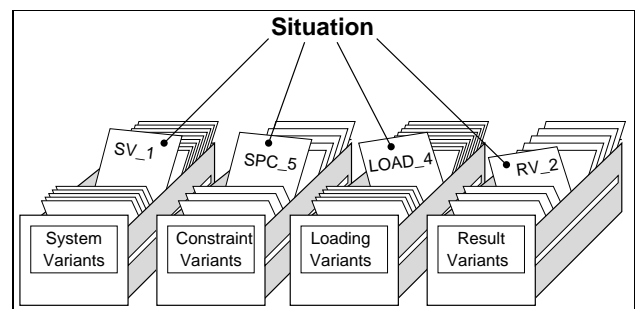


Figure 70: Variants in PERMAS

Basic properties like nodal point coordinates, element topology and global coordinate systems are invariant.

Single Variants may be selected and examined together as a so-called **Situation**.

- Variants and Situations are identified by user defined names.
- The number of Variants and Situations is not limited.
- There is no restriction in combining the variant definitions with any **substructuring** feature.
- In each substructure the selection of system, constraint, loading and result Variants is independent of the selections made for other substructures (e.g. useful for models with symmetry/antisymmetry).
- PERMAS keeps a record of all calculation steps already completed. In this manner, multiple cal-

culations are avoided when using a Variant repeatedly.

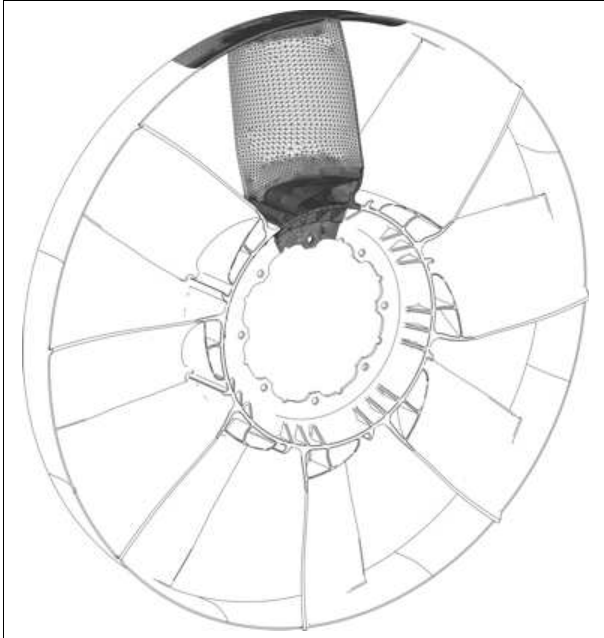


Figure 71: Cyclic symmetry of a fan  
Behr GmbH & Co., Stuttgart, Germany.

## Cyclic Symmetry

Cyclic symmetric structures like Fig. 71 are characterized by a number of identical sectors each rotated by a multiple of the sector angle around the axis of symmetry.

Cyclic symmetric structures can be handled directly for static analysis with matching cyclic symmetric loads and real eigenvalue analysis (see modules LS and DEV on pages 65 and 73). There, the analysis of the whole structure is replaced by a series of analyses for one sector with different boundary conditions. In addition, the number of nodal diameters has to be specified, i.e. number of waves along the perimeter.

Substructuring can be used to model a sector and the analysis of cyclic symmetry is performed in the top component.

In topology optimization (see module TOPO on page 86) the result can be forced to become cyclic symmetric.

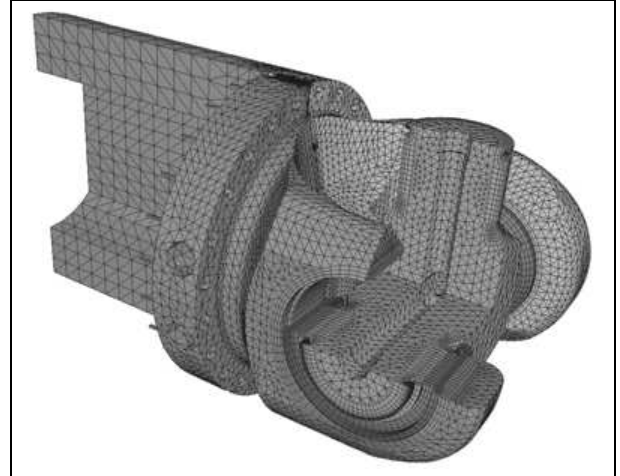


Figure 72: Incompatible meshes  
Cardan shaft model in sectional view  
(Voith Turbo GmbH & Co. KG)

## Surface and Line Description

The description of surfaces in PERMAS is used for the specification of structural parts which have to be coupled automatically (see next section). This description is made using one of the following methods:

- by specifying element surfaces, e.g. for a set of volume elements (for Sets see page 57),
- by specifying geometry elements (see element library page 55).

For surfaces very accurate coordinates are frequently required (as e.g. for contact with incompatible meshes), which are not available through the pre-processor used. Such surfaces can be smoothed internally by correction of runaways, which leads to very precise surfaces for more accurate analysis results.

In the same way does the description of lines serve as specification of structural parts which have to be coupled automatically along lines (see next section). This description is made using one of the following methods:

- by specifying element edges, e.g. for a set of face elements (for Sets see page 57),
- by specifying geometry elements (see element library page 55).



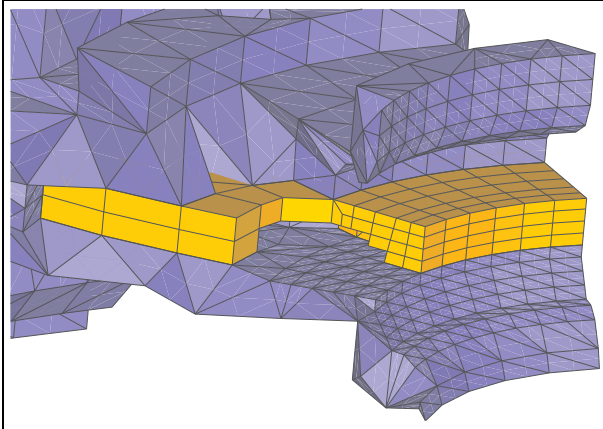


Figure 73: Incompatible meshes  
Element transition (HEXE8/TET10)

## Automated Coupling of Parts

The automated coupling of incompatible meshed parts brings a number of benefits for the user:

- It allows for a much more flexible organization of the model generation where single parts are administrated and exchanged.
- It facilitates a fast modification cycle for virtual prototypes.
- Accurate mesh transitions without bad elements are possible.
- Welding spots and surface contact become much easier with automated part coupling.

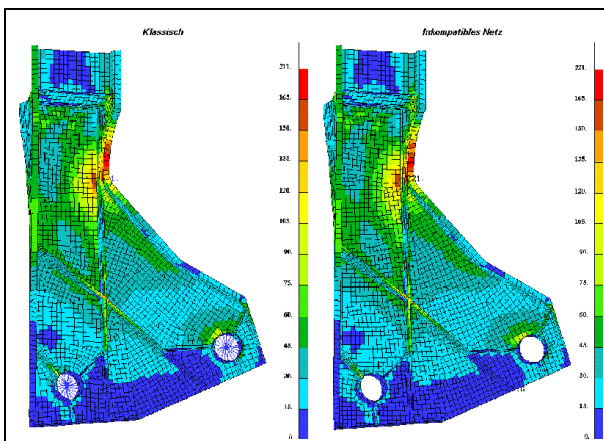


Figure 74: Ribbed shell model  
with compatible meshes (left) and  
incompatible meshes (right)

The coupling definition consists of the following steps:

- Definition of a guiding surface and the degrees of freedom to be coupled

- Definition of a dependent surface or node set

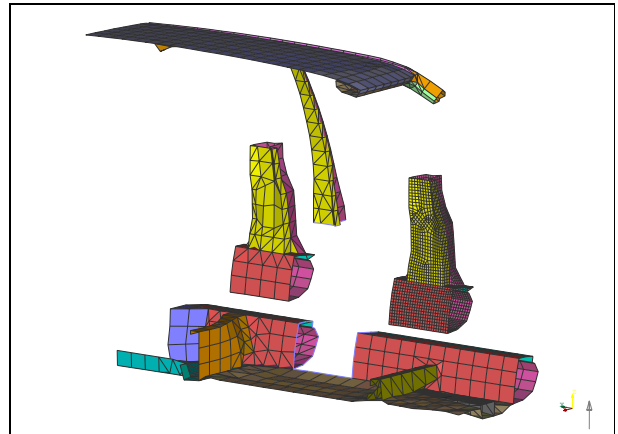


Figure 75: Application of local mesh refinement

Subsequently the neighborhood computation takes place and the parts are connected by *MPC*-conditions automatically. The result of the neighborhood computation is available for post-processing and verification purposes.

The coupling is a general feature that may also be used for coupled analyses, where different mesh densities occur due to the modeled physics. One example is a coupled fluid-structure acoustic computation, where the acoustic mesh may be coarser than the mechanical part.

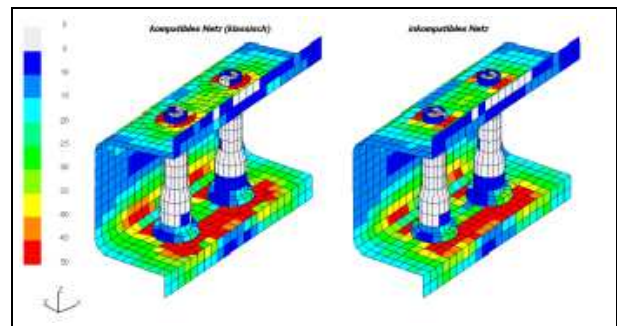


Figure 76: Results for part assembly  
with compatible (left) and incompatible (right) mesh

One valuable advantage of incompatible meshes is their use in shape optimization. The position of bolts or ribs can be optimized in a shape optimization without any remeshing of the structure. Fig. 77 shows an example, where the initial position and the final position of the bolts are very different. Nevertheless, after a number of iterations the final position is reached which is still symmetric and shows identical bolt forces for all the bolts.



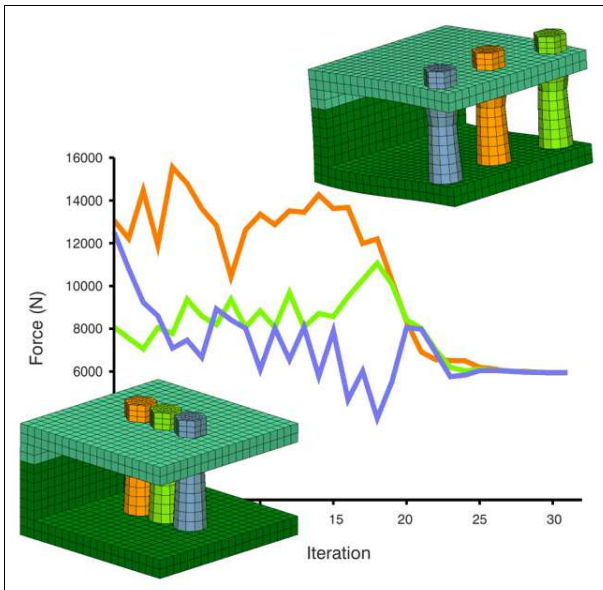


Figure 77: Shape optimization with incompatible meshes to achieve smallest relative displacements and identical bolt forces (top: initial design, bottom: final design; in between: bolt forces for each iteration)

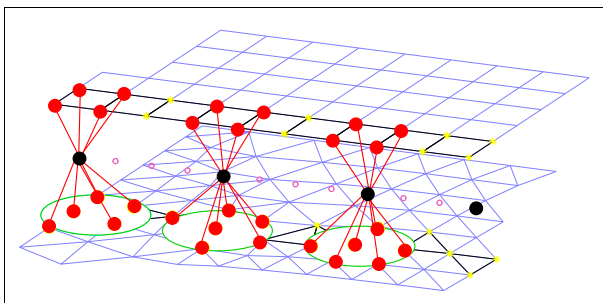


Figure 78: Connection of spotweld forces

## Automated Spotweld Modeling

For the automated modeling of spotwelds neighbored surfaces are connected using predefined points:

- Specification of spotweld positions
- Selection of (incompatible) faces
- Specification of spotweld stiffness and (optionally) the spotweld diameter

The spotweld stiffness is modeled by a spring element, which is coupled to the neighbored parts by automatically generated *MPC* conditions (see Fig. 79). Available results are the spring forces and the reaction forces at the coupling nodes of the joint parts.

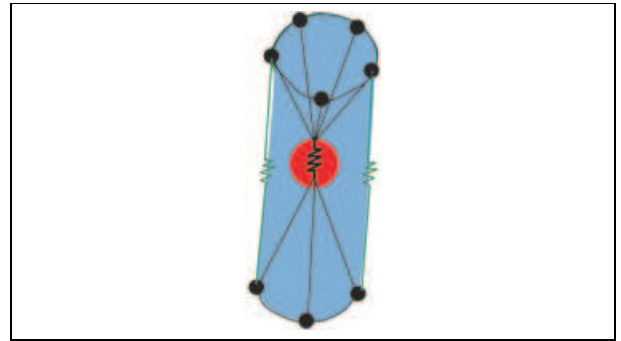


Figure 79: Spring element at spotweld position

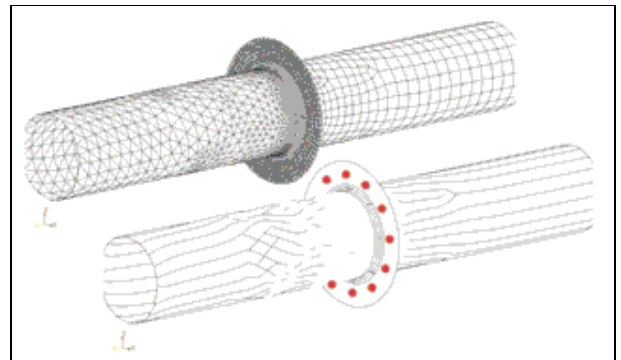


Figure 80: Spotwelds and incompatible meshes

For verification purposes the topology and the connecting vectors of the generated spotweld connections can be issued for graphical post-processing (see Fig. 81).

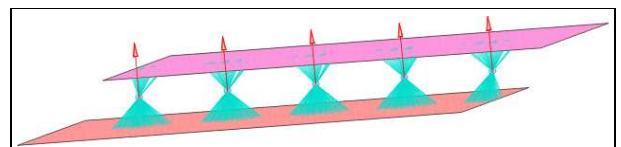


Figure 81: Verification of spotweld connections

A more sophisticated spotweld model has been developed with module WLDS (see page 93). Using very simple modeling it achieves best possible results.

## Kinematic Constraints

For the specification of single point kinematic constraints (absolute constraints) suppressed and prescribed degrees of freedom are available for zero or non-zero displacements, respectively.

Multilinear kinematic constraints (relative con-

straints) between several degrees of freedom are described by the use of so-called Multi Point Constraints (**MPCs**).

PERMAS offers a great number of various MPCs – all of them comfortable tools for modeling:

- Multiple degrees of freedom may be forced to have identical freedom values by simple **Assignment** (for modeling swivels, hinges or sliding surfaces and for boundary conditions in cyclic symmetry).
- **Rigid Bodies** allow the modeling of rigid parts within an elastic structure. There can be one or several guiding degrees of freedom and one or several dependent degrees of freedom, too.
- **Interpolation Regions** may be used for mesh refinements, coupling of incompatible meshes, distributing loads, or transfer of results between different meshes:
  - lines with 2 or 3 guiding nodal points,
  - triangular and quadrilateral areas with 3 or 6 and 4, 8 or 9 guiding nodes, respectively,
  - Volume areas as hexahedra (with 8, 20, 27 nodes), pentahedra (with 6, 15, 18 nodes), tetrahedra (with 4, 10 nodes), pyramids (with 5 nodes).

The interpolation regions may be used also for a **Volume-Shell Transition**, i.e. the connection of plates and shells with solid element structures (see Fig. 82). On the basis of a guiding and a dependent surface, the corresponding constraints for the dependent nodes are generated automatically (see also the Surface Description on page 51 and the Automated Coupling of Parts on page 52).

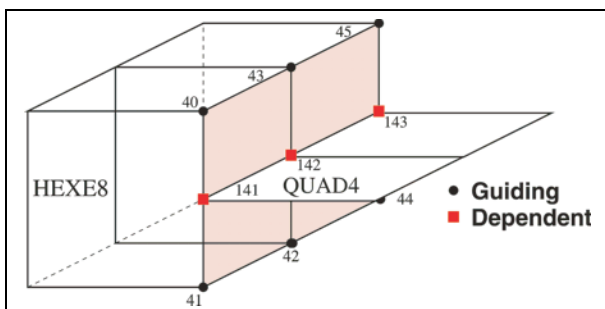


Figure 82: MPC example: Volume Shell Transition

- **Linearization areas** are used to linearize surfaces of bodies with quadratic elements. Among others, this can be used in contact analysis to reduce the number of surface contacts.
- **General MPCs** allow any linear combination of

the involved degrees of freedom.

General MPC conditions can be used to model **Press Fits**, e.g. shaft-hub connections where the shaft has a slightly larger outer diameter than the inner diameter of the hub. Typically, such configurations are modeled using contact. If any change in the contact is not expected, it would be more efficient to use a general MPC to model the interference in order to keep a problem in the linear range instead of a nonlinear contact analysis. It was proven that contact analysis and general MPC give identical results for corresponding models (see Fig. 83).

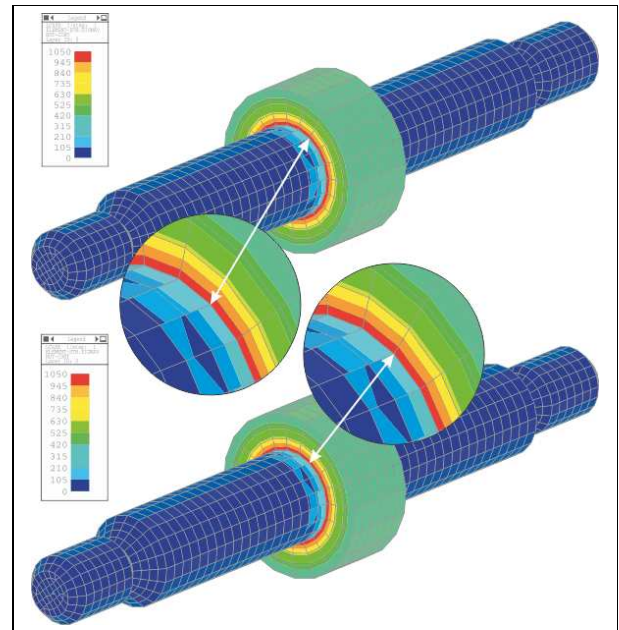


Figure 83: Press fit between shaft and hub  
Identical stress results of MPC (above)  
and contact solution (below).

The number of dependent degrees of freedom for each MPC is not limited. Also multi-level conditions (hierarchical MPCs) may be used as long as there is no recursive interdependence.

The coupling of single components using the **sub-structure technique** is defined by 'external degrees of freedom'. On the higher component's level these external degrees of freedom can be part of single or multipoint constraints.

## Handling of Singularities

In static analysis there are two different kinds of singularities, which are detected automatically and communicated to the user in a suitable way:

- For redundant degrees of freedom, which have no stiffness (e.g. perpendicular to rods or membranes), the user gets a list on the result file.
- For rigid body degrees of freedom the related displaced shapes are issued on a post-processor file. They can easily be inspected in order to detect the missing supports or other modeling errors.

VisPER provides an RBM Assistant (see page 40) which allows for the elimination of rigid body modes prior to analysis.

In dynamic mode analysis the rigid body modes are detected and decoupled automatically.

In dynamic response analyses in the time or frequency domain the absolute response results form a superposition of the elastic and the rigid body response.

## Element Library

The PERMAS elements are generally usable for different types of physical degrees of freedom (like displacements, temperature, electromagnetic potential, etc.), beside some specific, application-dependent elements. Currently, the following elements are provided:

- **Solid Elements:**
  - tetrahedra with 4 or 10 nodal points and straight or curved edges,
  - pyramid element with 5 nodal points,
  - pentahedra with 6, 15, or 18 nodal points and straight or curved edges,
  - hexahedra with 8, 20, or 27 nodal points and straight or curved edges.
  - hexahedra with 8, 20, or 27 nodes, pentahedra with 6, 15, or 18 nodes, tetrahedra with 4 or 10 nodes, and pyramid element with 5 nodes as *fluid elements*.
- **Flange (Rod) and Membrane Elements :**
  - flange elements with 2 or 3 nodal points,
  - triangular elements with 3 or 6 nodal points

and straight or curved edges,

- quadrilateral elements with 4, 8 or 9 nodal points and straight or curved edges,
- quadrilateral shear panel with 4 nodal points.
- **Beam Elements** with 2 nodal points, optionally with or without rigid lever arms (offset nodes):
  - beams with arbitrary shaped solid cross section,
  - thin-walled profiles and tubes with open or closed cross sections,
  - thin-walled open or closed tubes with cross sections tapered along the beams' length axis,
  - fluid-filled or fluid-surrounded straight or curved pipe elements.

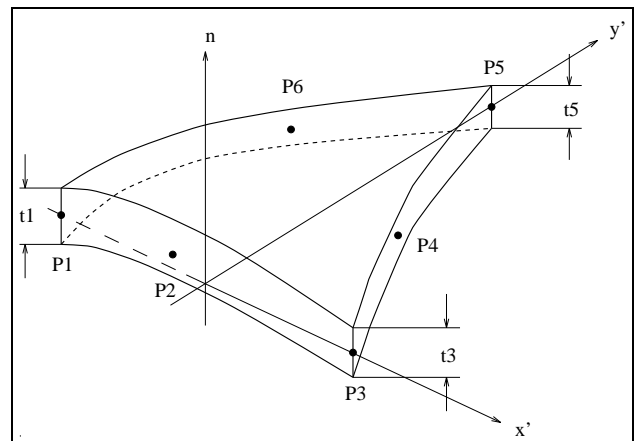


Figure 84: Triangular element with curved edges

- **Plate and Shell Elements :**
  - triangular or quadrilateral elements for thin and moderately thick plates and shells.
  - triangular or quadrilateral elements *sandwich shells*.
  - a thin-walled triangular plate element following *Kirchhoff's* Theory.
  - triangular (with 3 or 6 nodes) and quadrilateral elements (with 4, 8, or 9 nodes) with 3-dimensional shell formulation for linear and non-linear material.
  - triangular and quadrilateral elements for layered cross-sections (*composites*).
- **Discrete Elements:**
  - linear and non-linear spring elements,
  - various mass elements,
  - linear and non-linear damper elements,
  - control elements,
  - elements with direct matrix input.
- **Elements for contact analysis:**

- Compensation springs to ground or between 2 nodes with 3 translational DOF, or between two nodes with 1 translational DOF.
- **Scalar Elements:**
  - spring elements between 2 degrees of freedom or for a support of 1 degree of freedom to ground,
  - damper elements between 2 degrees of freedom or for a connection of 1 degree of freedom to ground,
  - scalar masses for 1 or 2 nodal points,
  - scalar mobility element for fluid meshes.
- **Load carrying membranes** in form of triangular respectively quadrilateral areas for load application and stress evaluation.
- **Plot Elements** in form of points, lines and triangular respectively quadrilateral areas for result evaluation.
- **Geometry Elements** in form of lines and triangular respectively quadrilateral areas for line and surface definition.
- **Convectivity Elements** to model the convectivity behavior and radiation on free surfaces in thermal analysis.
- **Fluid-Structure Coupling Elements** for coupled acoustics and surface absorption.
- **Surface Wave Elements** for acoustic analyses.
- **Semi-Infinite Elements** for acoustic and electromagnetic analyses.
- **Radiation Boundary Condition (RBC) Elements** for the modeling of acoustic radiation effects.
- **Axisymmetric Elements** for structures, heat transfer, acoustics, and electromagnetics.

## Standard Beam Cross Sections

The following standard geometries are supported for thin-walled beam cross sections:

- General thin-walled open section (see Fig. 85),
- Circular cross section,
- Box cross section.

Using those cross sections allows to get stress results at certain points of the cross sections.

The parameters of the cross sections are directly available as design parameters in an optimization run (see page 83).

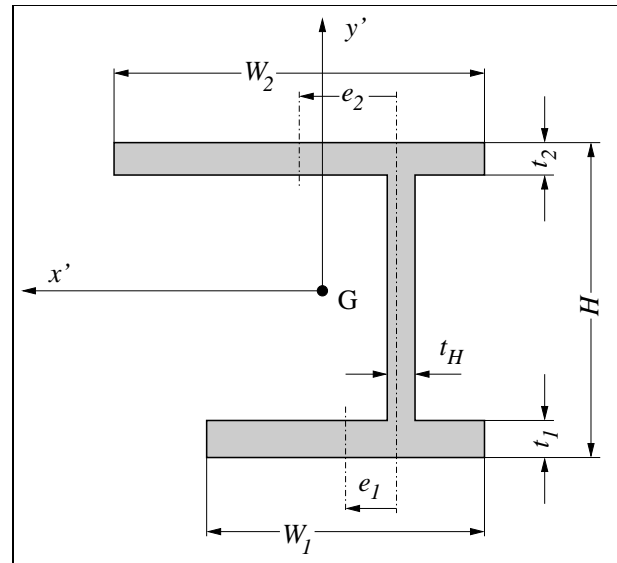


Figure 85: General thin-walled open section

## Design Elements for Optimization

For design optimization purpose (see page 83), all parts of the structure which may be modified are assigned to so-called Design Elements. They are used to define possible modifications for elements and nodes belonging to the design element.

To facilitate shape optimization a number of additional design elements are available to collect structural elements belonging to one design element. All types of finite elements are allowed to be part of these design elements:

- design element for smooth shape optimization,
- design element for bead pattern optimization,
- design element for free shape optimization,
- design element for topology optimization.

## Error estimator

The element size in FE meshes influences the accuracy of the results (in particular stress results). But the 'right' element size is a local characteristic and depends on the force flow and its gradients. Particularly, notch stresses are highly dependent on element size.

With the *refinement indicator* pre-processors like



MEDINA may be able to perform an adaptive mesh refinement (see Fig. 86).

But this indicator can also be used to identify those mesh regions where stresses are possibly not very reliable.

In addition, the error estimation can also be used to improve the stresses by a smoothing process without a new mesh or a repeated analysis run.

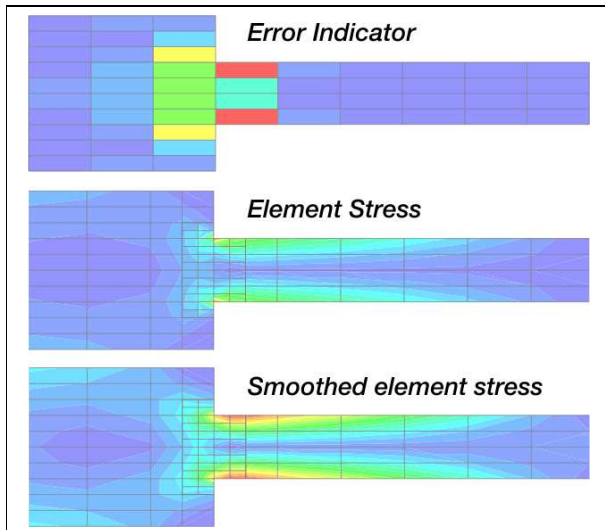


Figure 86: Example for *error indicator*, *element stress*, and smoothed element stress

## Material Properties

The description of material is made independently of the other model data. In doing so, the needed material parameters can easily be taken from already existing data sets building a certain kind of material data base.

Beside isotropic material all kinds of *anisotropy* can be applied in a simple and suitable way: transverse isotropic, orthotropic, monoclinic, and general (triclinic) material.

Because of the application of PERMAS in different fields, different material properties can be defined like elasticity, density, compressibility, damping, thermal expansion, heat conductivity, heat capacity, absorption, volume drag, electric conductivity, dielectricity, magnetic permeability. There, only the required data are really used, all additional material properties don't have any effect.

For nonlinear material, additional input facilities are provided like stress-strain curve, yield load, creep behavior.

Almost all material properties can be specified as temperature-dependent. Following a given temperature field, the actual material properties are determined by an interpolation from the given distribution function.

The definition of damping can be frequency-dependent.

Layered composites (laminates) can be defined by different materials (like fibre-reinforced material) for each layer (see page 93).

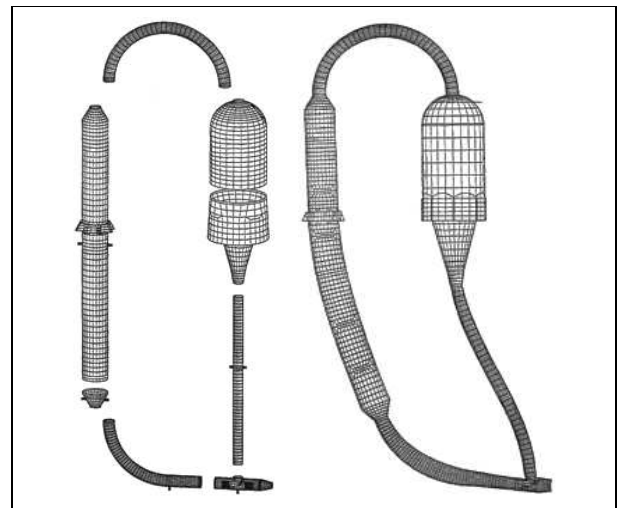


Figure 87: Mode shape of a reactor

## Sets

For a lucid model description node and element sets may be used. Such sets may be generated from other sets using Boolean operations. In addition, several sets can be collected in a set bin, which denotes a set of sets.

Beside the model description such sets may be used also for the specification of result output. For the print output of results and the export to connected post-processors the amount of issued results can be restricted to the relevant data by sets. In case of analyses in the time or frequency domain, the required run time and disk space can be drastically reduced by the specification of sets for which the results are to be determined.

All interfaces preserve the sets and their identifiers from the pre-processing via the solver to the post-processor. An accompanying text marks a set more precisely.

---

## Mathematical Functions

For the description of complex data relations a library of mathematical functions is provided for: polynomes, trigonometric functions, exponential functions, and discrete functions. E.g. they allow for the definition of time-dependent loading and temperature-dependent loads.

Beside the library functions also tabular functions or user functions (defined by Fortran or C subroutines) may be defined and used.

On all of these functions sums, products, or chains may be defined in order to specify complex compound functions.

In addition, a special function to derive the *Fourier* coefficients of a time signal is provided. Time-dependent excitations are analysed by a Fourier transformation and the resulting frequency-dependent coefficients are used in a frequency response analysis (see Fig. 88).

---

## Loads

In static analyses the following mechanical loads may be used:

- Global loads:
  - Inertia loads,
  - *Inertia relief* (quasi-static acceleration).
- Nodal loads:
  - Concentrated loads,
  - Distributed loads,
  - Prescribed displacements,
  - Temperature loads,
  - State of contact.
- Element loads:
  - Distributed loads,
  - Initial strains.

For distributed surface loads, e.g. not specified at

the nodes of a mesh but a coarser grid, special load elements (see page 56) can be used which are coupled to the main structure by interpolation (see page 54).

In heat transfer analyses the following thermal loads may be used:

- Nodal loads:
  - Concentrated heat flows,
  - Distributed heat flow,
  - Prescribed temperatures.
- Element loads:
  - Distributed heat flow.

Temperature fields may be taken directly from a previous heat transfer analysis. One temperature field may be used to define temperature dependent material properties, and another temperature field may be used for initial strain calculation or as initial condition for a transient thermal analysis.

All distributed element loads can be defined as coordinate-dependent like hydrostatic pressure.

Load case combinations may be applied to create new loads from existing patterns.

Time-dependent loads are defined as a product of a static or thermal load and a time-dependent function. In addition, initial conditions for time-dependent mechanical or thermal analysis can be specified. These can simply be taken from a previous analysis.

For frequency-dependent dynamic loads it always holds that they are defined as a product of a static load and a spectrum of amplitudes and phase angles.

Time dependent loads of a periodic process can be transformed to frequency dependent loads by an automatic *Fourier* analysis (see Fig. 88). Then, one or more frequency response analyses can be performed. Finally, by superposition of the results in the time domain a periodic response can be achieved.

---

## Model Verification

With increasing complexity of models the need to verify the correctness of model descriptions increases. PERMAS pays attention to this point by

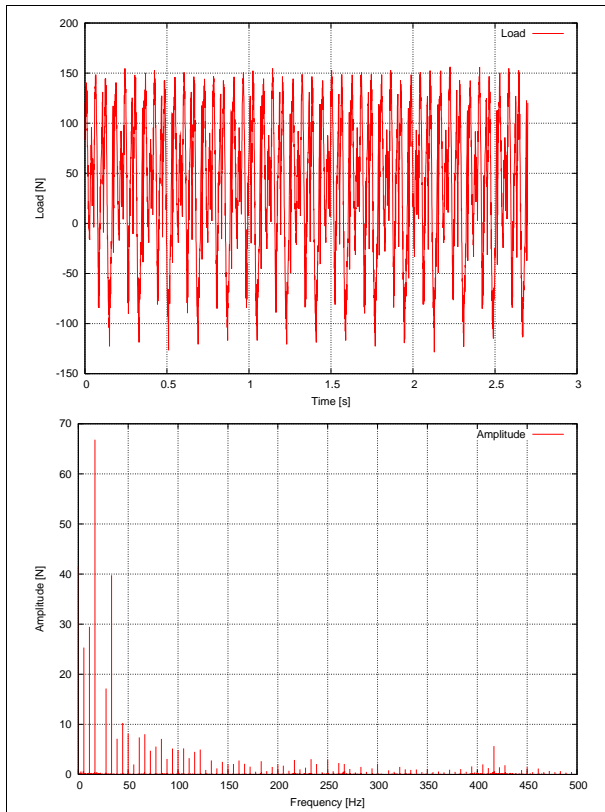


Figure 88: Example for the transformation of a periodic load (above in time domain, below in frequency domain)

providing model properties in a form that they can be post-processed like results.

It is an important principle of model description that all quantities which can be specified can also be exported for visualization and checking purposes. In particular, this holds for all automatically generated quantities which facilitate model description.

Since VisPER is available, verification of PERMAS models became much easier. It does not only provide visual means to check the model, but it provides also capabilities to visualize the verification data generated by PERMAS (see examples below). For PERMAS models, VisPER is the recommendable verification tool.

The following list gives a few examples of verification aids:

- Element test results are exported for identification of erroneous elements.
- Mean element thickness at nodes can be exported to check the thickness distribution in a shell structure.
- Rigid body modes detected during the analysis

can be exported for viewing in order to verify support conditions.

- For MPC interpolation regions (see MPC conditions on page 53) the offset vectors are exported to verify the result of the automatic interpolation.
- For spotweld connections the generated weld base vectors are issued (see Fig. 81).
- VisPER can be used to check distributed loads (even those with a function dependent on geometric position).
- Element temperature loads and distributed nodal temperature loads can be exported as temperature result for visualization.
- Initial strains are available for postprocessing.
- For contact modeling a number of entities are available for verification like contact definition, contact geometry, contact coordinate system, initial contact status, initial gap width (see Fig. 89).
- For the pretension of bolts an additional number of description entities can be postprocessed like pretension definition, pretension coordinate system, pretension thread vectors (flank normal, downhill and pitch direction).
- For heat transfer analyses with heat exchange by radiation the radiating surface can be coarsened automatically. The element mesh of the coarsened surface can be exported for checking.
- In fluid-structure coupled analysis the interface elements and their orientation can be checked.
- In optimization applications the use of design elements can be checked by VisPER (see pages 42 and 42). Besides, the assignment of finite elements to design elements can also be checked.

For a number of input data, PERMAS can generate postscript files using *gnuplot* (if gnuplot is available on the currently used machine). It will generate gnuplot command files, and a table with the xy-data of the curves will be exported to a file. Such input data include load history (see page 67), material curves for plasticity, and transient loads in timehistory response analysis.

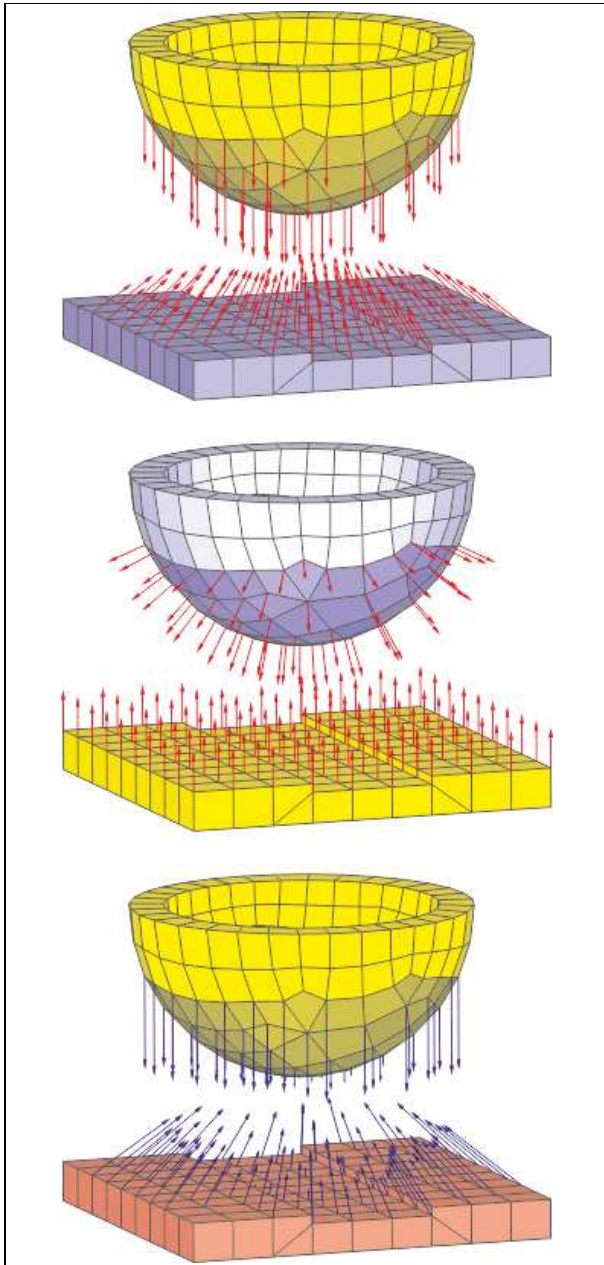


Figure 89: Verification of a contact model

Top: contact normal vector  
 Middle: surface normal vector  
 Bottom: nadir vector

## Interfaces

The integration of PERMAS in the pre- and post-processor is of top priority for the interface development. Therefore, all interfaces are directly integrated without any separate software tools. These interfaces are denominated as 'Doors', which allow for a very direct access to the original model data. Above the model description, some interfaces allow for standard solutions, which make the working en-

vironment more comfortable.

Beside its own input and output formats PERMAS offers Doors not only to various pre- and post-processors but also to model files of other FEA systems:

- MEDINA (page 95)
- PATRAN (page 95)
- I-DEAS (page 95)
- ADAMS (page 96)
- DADS (page 96)
- SIMPACK (page 96)
- EXCITE (page 96)
- MOTIONSOLVE (page 97)
- HYPERVIEW (page 97)
- VAO (page 97)
- Virtual.Lab (page 97)
- MATLAB (page 97)
- NASTRAN (page 97)
- ABAQUS (page 98)
- MpCCI (page 99)

Moreover, a growing number of extra interfaces to PERMAS are available from partner companies or INTES (see page 99).

All users with a heterogeneous environment of pre- and post-processors benefit from the excellent interfaces to other CAE products.

During **input** complete FE-models are read and translated into autonomous PERMAS data structures. If requested, any part may be combined with or completed by additional PERMAS data files.

During **output** the calculated results may be written in various forms (listing, xy-plot files, several post-processor formats). In addition, the complete FE model may be output as PERMAS model file or as post-processing model in diverse pre- and post-processor formats – independent of the kind of input.

The result evaluation even for very large models is supported by comprehensive selection facilities for the exported result data. Beside the selection of relevant loading cases, time steps, and frequencies the specification of node and element sets is used to reduce the amount of exported results.

This high-level Door concept endows a number of **advantages**:

- High comfort:



- no external interface necessary,
- similar command syntax for all Doors,
- no intermediate files necessary.
- The translation process is very fast.
- Simple **input mixing**:
  - Several input models may be concatenated to one big model, no matter from which Door the model files were read.
  - The model description may be spread on several files, utilizing the possibilities of all Door input formats involved.
- With the direct translation, it is even possible to support features not quite compatible with any DAT file input.

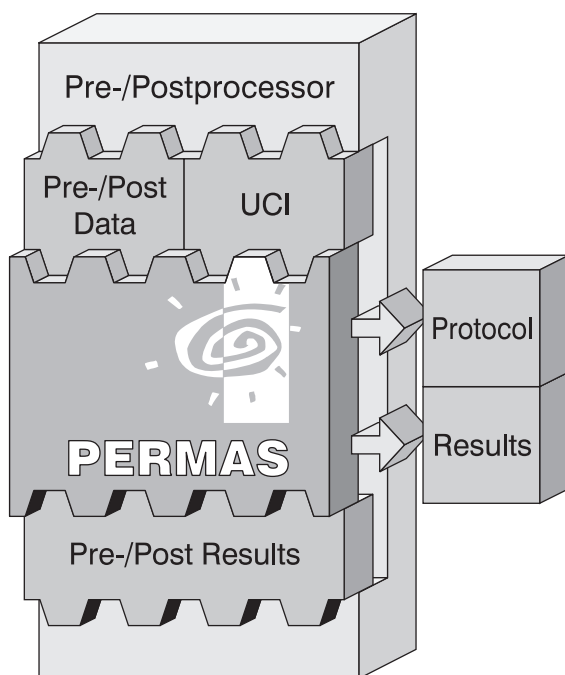


Figure 90: Integration of PERMAS  
in pre- and post-processor

- Quality assurance for modeling is improved:
  - Common consolidation of all input data (extensive model testing).
  - External identifiers are preserved and are used throughout the PERMAS run.
  - Additional data may be given by a separate input file – the original input file remains untouched.
  - The mixing feature enables the user to define difficult parts with the most convenient input format. Naturally, this will be the least error susceptible definition, too.
- Because the export format is independent of the input format, PERMAS allows for the transla-

tion of any pre-processor format to any post-processor format.

## Matrix Models

PERMAS data objects for results or matrices can be output to or input from external files in different ways and in binary or ASCII format.

This tool can be used to organize the data exchange with third party software, where no direct interface is available. In addition, it can also be used to store the data for later use in another PERMAS run to save computer time.

A typical application is the intermediate storage of condensed models. There beside the condensed matrices also a condensed model can be generated and exported in order to use them in another run. For a detailed post-processing the results can then be transformed back to this model. To reduce size of matrices and results this transformation can be restricted to sets in advance.

Condensation of substructures is used beside static analyses mainly for vibration analysis and response analysis in time and frequency domain. There, beside stiffness and mass matrices eigenvalues, mode shapes, loads, and damping matrices are provided. Additional static mode shapes can also be taken into account.

Another application is the generation of modal models (e.g. by *generalized modal condensation*, see page 74). These modal models are intended for modal response analyses outside of PERMAS.

All standard ASCII output files can be generated as compressed files (gzip) saving disk space and time (for large files). Because the input is possible in compressed format, too, there is no explicit need to store model and result data in full ASCII format.

Special formats available for matrix export are MATLAB format and *Rutherford-Boeing* format.

## Combination of Results

After the computation, different primary and secondary results may be combined to new results. For this purpose different summation rules and mathematical functions (see page 58) are applicable. In this way, even results from different variants may be combined.

The generated results may overwrite existing results or generate new ones. The output of the combined results is exactly like the output of the original results.

## Transformation of Results

All nodal results are calculated in the global coordinate system of the actual Component. From these results the following transformations may be performed:

- into the displacement coordinate system of the respective node.
- into a specific coordinate system for all nodes (cartesian, cylindric, spheric).
- using a special transformation for each node.

All transformations may be performed later in backward direction to the Component system.

Beside the transformation of real results, complex results of a frequency response analysis can be transformed to another coordinate system, too.

## Comparison of Results

The following facilities are available for the comparison of dynamic analysis results between two variants:

- **MAC** (Modal Assurance Criterion):  
Compares two sets of eigenvectors (with same number of nodes) from different situations and gets their degree of correspondence. MAC factors are computed as:

$$MAC = \frac{(\mathbf{X}_1^t \mathbf{X}_2)^2}{diag(\mathbf{X}_1^t \mathbf{X}_1) diag(\mathbf{X}_2^t \mathbf{X}_2)}$$

- **CoMAC** (Coordinate Modal Assurance Criterion):

Compares two sets of eigenvectors (with same number of nodes) from different situations and gets their degree of correspondence. CoMAC factors are computed as:

$$COMAC = \frac{diag(\sum_{m=1}^M |\mathbf{X}_1^m \mathbf{X}_2^m|)^2}{diag(\sum_{m=1}^M \mathbf{X}_1^m)^2 diag(\sum_{m=1}^M \mathbf{X}_2^m)^2}$$

The result gives a measure of the correlation of the vector sets in each degree of freedom direction.

- **COF** (Cross Orthogonality Factors):  
Compares two sets of eigenvectors (with same number of degrees of freedom) from different situations and gets their degree of correspondence. Cross orthogonality matrix and factors are computed as :

$$COFM = \mathbf{X}_1^t \mathbf{K} \mathbf{X}_2; \quad COF = COFM^t COFM.$$

All comparisons may be performed for two variants of a model and the compared model parts can be restricted to specified node sets.

## XY Result Data

On the basis of element and node sets, xy data can be generated directly and issued on all connected output files for graphical processing.

Usually, xy data are directly extracted from the corresponding result item such, that the abscissa value is given by the column (i.e. loading case, iteration) and the ordinate values are the result data. If the columns are associated with certain values (like frequencies, time, load factors), these values will be used for the abscissa.

As a special case in order to study e.g. the stiffness behavior of a part, it is possible to create xy data for nodal point results with the coordinate direction as abscissa. Then, the abscissa values are prepared in ascending coordinate direction for the selected nodes. A local coordinate system may be used to select the coordinate directions.

For a number of input data, PERMAS can generate postscript files using **gnuplot** (if gnuplot is available on the currently used machine). It will generate gnuplot command files, and a table with the xy-data of the curves will be exported to a file. Such input data include load history (see page 67), material curves

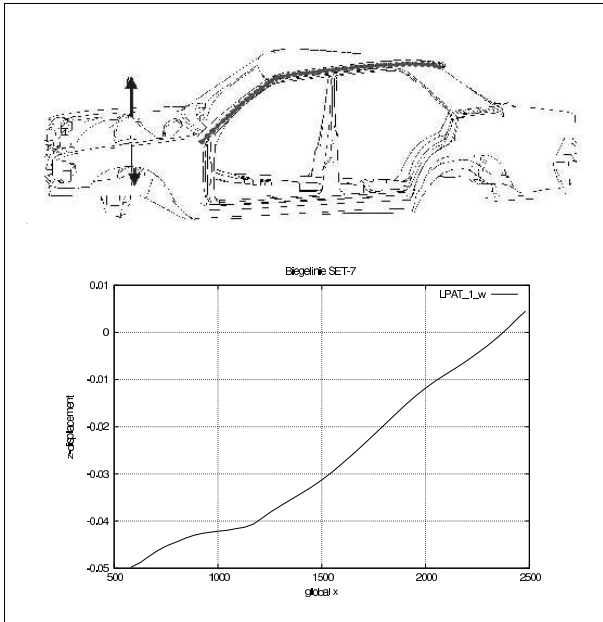


Figure 91: Determination of elastic line for torsional load case (displacement for the marked nodes)

for plasticity, and transient loads in timehistory response analysis.

Within *PERMAS Tools* plots can be created using *PERMASgraph* (see page 102 for details).

## Cutting Forces

On the basis of element and node sets the cutting forces can be determined and exported for post-processing for almost all points of a structure. In addition, the sum of all forces and moments over a cut is calculated and printed for a previously specified point in space (see Fig. 92).

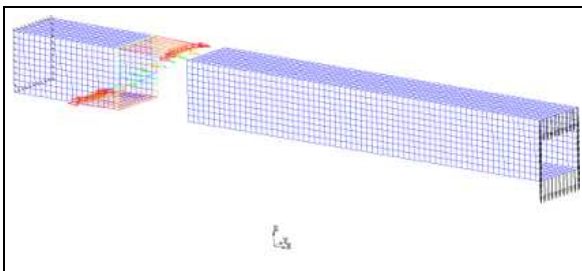


Figure 92: Determination of cutting forces

## Restarts

Each PERMAS run opens a data base file, which may be used for subsequent runs. At every restart, the latest status of the data base is always available from the previous run. This includes all intermediate results possibly obtained only by using considerable computation time.

In doing so, e.g. different load variants in several runs may be processed without assembling and decomposing the stiffness matrix every time.

## Open Software System

PERMAS is an open software system with respect to its capability to include user-written routines in every program execution, which may be invoked during run-time.

On the one hand, this feature is used to offer maximum flexibility in defining data dependencies, e.g. with user-defined mathematical functions (see page 58).

On the other hand, PERMAS may be used as subroutine library in order to write own solutions or result evaluations. These can be invoked at the right place during program execution.

The programming language is Fortran 77/90 in any case, but on some platforms C is also available. The PERMAS library routines are available as Fortran programs.

## Direct Coupled Analyses

PERMAS enables different coupled analyses in one run, e.g.

- initial values for time-dependent dynamic analysis may be taken from a previously performed static analysis.
- initial values for a transient thermal analysis may be taken from another transient thermal analysis.
- a thermal stress analysis may be performed on the basis of a previous thermal analysis.

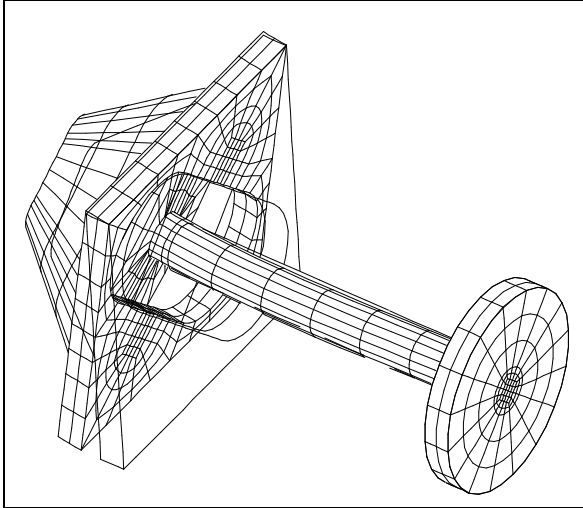


Figure 93: Clutch Element, Static Displacements

- a thermal or mechanical analysis on the basis of a previous electromagnetic analysis.
- a (fully) coupled fluid structure acoustics analysis.
- all optimizations, if different analysis types are used simultaneously (like static and eigenvalue analysis).

For all kinds of coupled analysis the same types of elements may be used for the different partial analyses. As far as appropriate, most element types are available for all different analysis types.

cation LIBrary (COCOLIB) has been developed by FhG/SCAI, which allows for a weak coupling of a structural mechanics software (CSD) and a fluid dynamics software (CFD). Both packages are running simultaneously and the data exchange is during run time following previously defined criteria and coupling algorithms (Gauss-Seidel, Jacobi). This software has been further developed and is now called *MpCCI* (more information at [www.mpcci.org](http://www.mpcci.org)).

The transferred boundary conditions comprise thermal and mechanical quantities (force  $\mathbf{F}$  or pressure resp., heat flux  $\mathbf{Q}$ , displacement  $\mathbf{r}$ , velocity  $\mathbf{v}$ , temperature  $\mathbf{T}$ ). There the coupling library does any interpolation and projection necessary due to incompatible meshes between structure and fluid.

The calls to *MpCCI* have been integrated into PERMAS. So, PERMAS and any CFD code where MpCCI is adopted, too, can be used to solve applications in mechanical, thermal, or thermo-mechanical coupling. Module PERMAS-CCL provides all necessary functions from the PERMAS side to support the coupling.

## Coupling with CFD

A coupling of structural mechanics and computational fluid dynamics has been realized within the scope of the CIPAR ESPRIT project by a loose coupling approach.

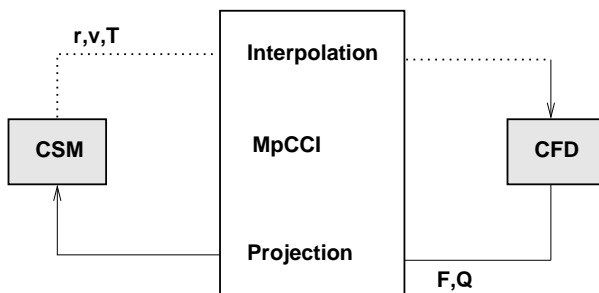


Figure 94: Coupling of CFD and CSM

Within the project a general COupled COmmuni-



## PERMAS Analysis Modules

### PERMAS-MQA – Model Quality Assurance

The PERMAS-MQA basic module builds the system kernel for all other software modules.

Among others this *system kernel* contains the data management system, the UCI command language, the standard input of PERMAS models, the standard output of results, the model consolidation, the *substructure technique*, numerous element types, and sampling for parameter studies.

The characteristic features of PERMAS-MQA are the concepts and tools for the **quality assurance** of the analysis process and for the investigation of different model parameters by **sampling**.

The **quality assurance** of finite element models becomes more and more important.

- Today, Finite Element calculations are used in the product development to early assess different design variants and to accelerate the development process.
- The skill of FEA systems users changes from an expert level to a more general background.
- The FE applications become more and more complex.

**Sampling** is the repeated execution of analyses where the values of design or basic variables are specified explicitly. Sampling works analogous to optimization and reliability analysis. So, all possible design variables (see page 83) or basic variables (see page 89) can be used for sampling and all analysis steps supported by optimization (see page 83) and reliability analysis (see page 89) can be used for sampling as well. For all desired results, sampling collects this data and prepares special XY data for convenient post-processing. Some examples for suitable variables are geometric parameters, Young's modulus for stiffness considerations, or different load parameters.

Beside the software quality (see page 19) the reliability of FEA results depends on the following points:

- **Comprehensive model testing:**  
PERMAS performs very intensive tests of the input data. There are several thousands of differ-

ent plain text system messages to react to complex inconsistencies in an appropriate way.

In particular, the automatic detection of singularities can save much time for the user (see page 55).

- **Avoidance of erroneous analysis runs:**

In PERMAS a **task scanner** has been introduced in order to avoid faulty runs:

- The analysis steps are checked in a group-wise manner to verify the feasibility of the complete analysis.
- The resources in terms of CPU time and disk space are estimated in advance.
- The input model data are checked for completeness and compatibility with respect to the analysis steps requested.
- Even the control of those PERMAS modules can be checked which are not licensed on the active platform.
- The model tests can be used in addition to similar tests of the applied pre-processor.

- **Relief of the user:**

Cumbersome routine work can be reduced and the overview of all model related information can be improved by comfortable input facilities, direct interfaces, and all of the test tools listed above. So, the user can concentrate on the objectives of the analysis and the evaluation of the results.

In particular, comfortable interfaces allow for a smooth transfer of model data from the pre-processor (see page 60).

A comment feature supports improved communication with *SDM (Simulation Data Management)* systems. These comments provide a means to describe any entity in the model description in any desired level of detail. To this end, the comments can be included in the model input file or the comments are linked to an additional file which also can include *XML* documents.

### PERMAS-LS – Linear Statics

This module allows for linear elastic calculations, based upon the assumptions of small displacements, small strains, and linear material behavior (isotropic or anisotropic). Therefore, this module of-

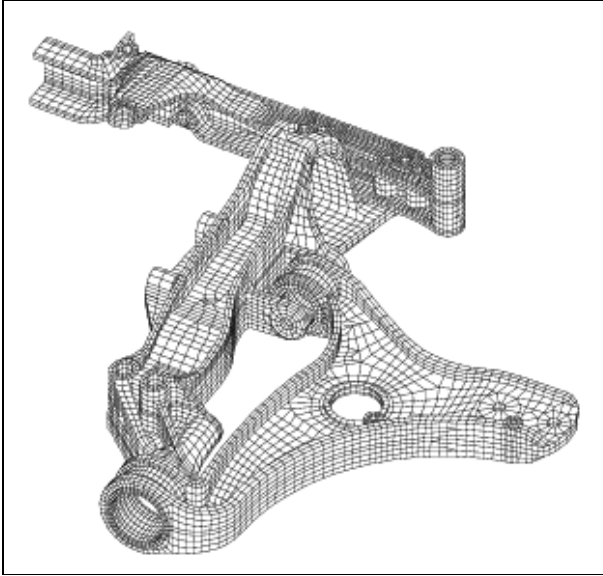


Figure 95: Model of a front axle  
Porsche AG, Weissach

ten builds the first step in Finite Element analysis.

- The following definitions of kinematic boundary conditions are available:
  - Suppressed degrees of freedom,
  - Prescribed degrees of freedom,
  - Linear constraints (*MPCs*) (see page 53).
- For free or partially free structures a quasi-static analysis can be performed (*Inertia Relief*). There, on the basis of a rigid body decoupling the inertia forces are computed which are in equilibrium with the applied forces. Subsequently, a static analysis is performed under the applied loads and these inertia forces.
- Different kinds of static loading are available (see page 58).
- The following primary results are calculated:
  - Displacements,
  - Mass and moments of inertia.
- From that additional results are derived:
  - Reaction forces,
  - Stresses and stress resultants,
  - Strains,
  - Residual forces,
  - Strain energy density, i.e. mass- or volume-specific strain energy.

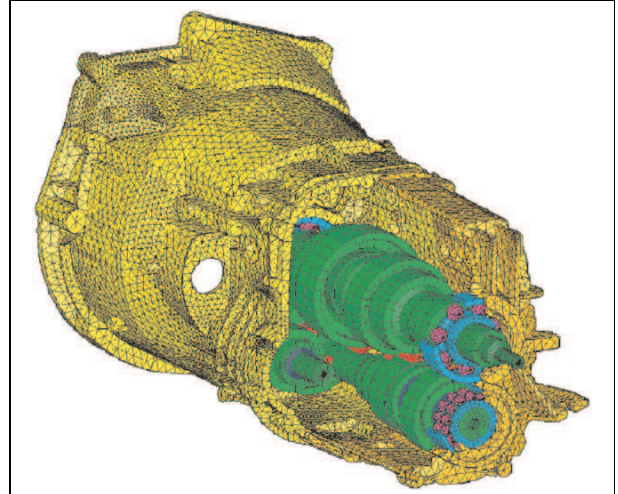


Figure 96: Bearing housing ZF AG

## PERMAS-CA – Contact Analysis

Static analyses with non-linear boundary conditions (contact problems) can be analyzed using the PERMAS-CA module.

Contact boundary conditions may be present between elastic bodies or between elastic bodies and a rigid counterpart. The bodies may behave also non-linearly.

In addition *self contact* is supported by contact analysis (see Fig. 9), i.e. contact models where a surface may get in touch with itself, e.g. due to warping or other forms of large deformations, and where contacting regions are not known in advance.

Several methods to describe contacts are available:

- specification of contact node pairs,
- specification of nodesets for each contact zone (the node pairs are detected automatically),
- assignment of nodes/nodesets to surfaces (incompatible meshes),
- definition of general surface-to-surface contact (incompatible meshes).

The feature to define contact with incompatible meshes allows the independent meshing of the contacting bodies. This simplifies the modeling of complex contact surfaces (like tooth contact between gearwheels) essentially.

The direction of contact and the initial gap width may be specified explicitly or determined automatically

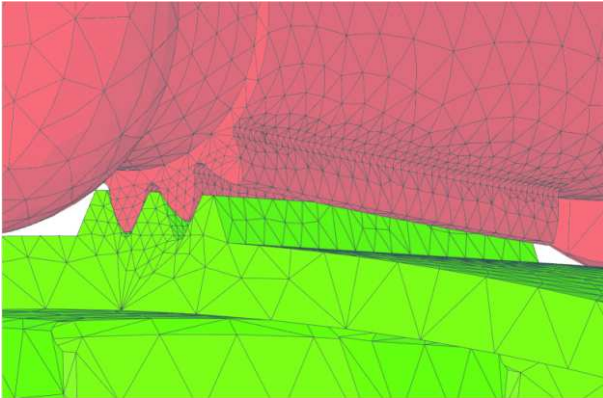


Figure 97: Contact with incompatible meshes

from the geometry. Any *press fit* is easily modeled by the specification of a negative gap width.

The contact analysis can handle isotropic or anisotropic frictional contact with sticking or slipping according to *Coulomb's law*.

For large contact problems and for complex frictional tasks module PERMAS-CAX provides a suitable extension of contact analysis (see next section). Both modules together cover a wide application area for contact analysis. This combination is complemented by module PERMAS-CAU (see section after next section) which takes into account large relative displacements of contacting bodies.

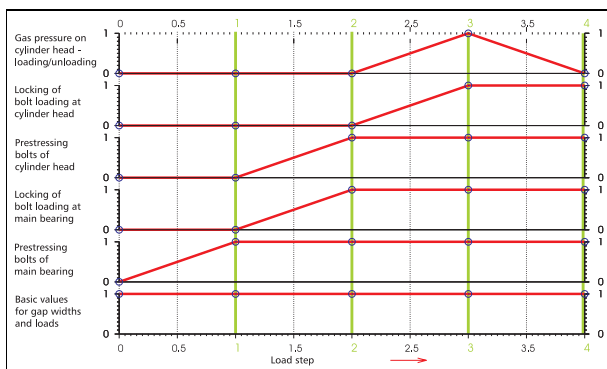


Figure 98: Representation of a load history

The specification of a load history allows the correct simulation of assembly and working loads and any contact situation with slipping and sticking friction. This facilitates the convenient simulation of such situations in a quasi-static analysis. A postscript plot file of the load history can be exported to view its graphical representation (see Fig. 98).

The load history can be amended by pretensioning

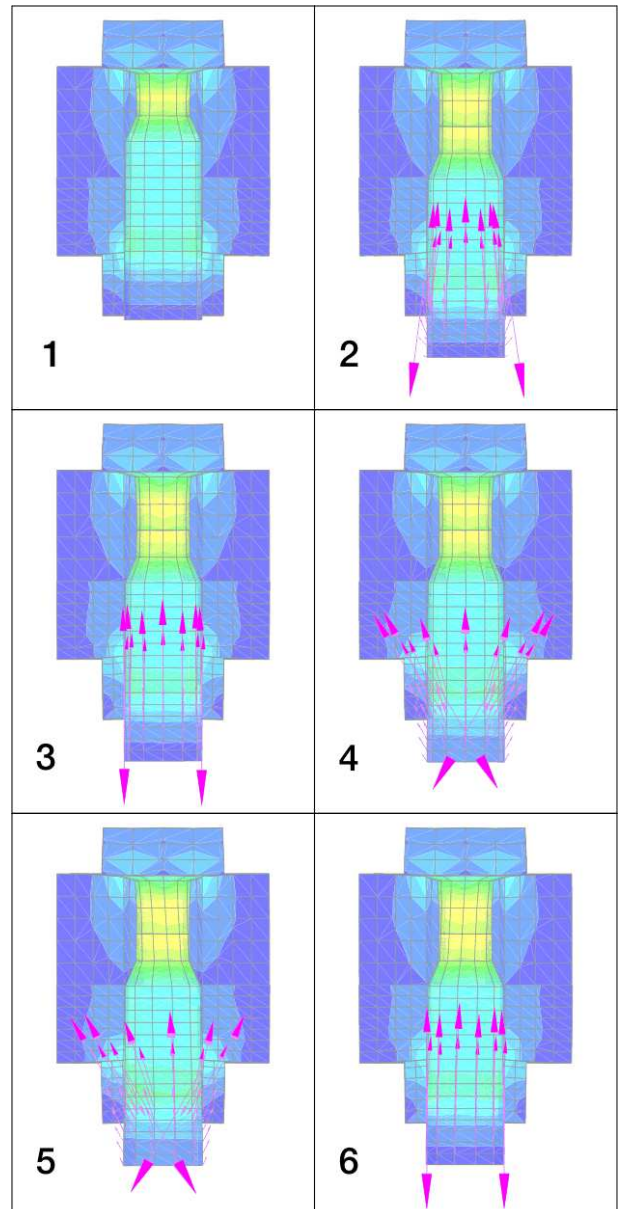


Figure 99: Bolt pretension  
Nodal point stresses and pretension forces  
for deformed mid-size screws:

- (1) with cutting plane in bolt
- (2) with radially joined thread
- (3) thread without radial coupling
- (4) thread with flank but no pitch angle
- (5) thread with flank and pitch (standard M10)
- (6) left-handed thread with pitch but no flank angle

(e.g. of bolts), where the contact analysis is used to describe the pretension. In this way, the screw tightening torque is modeled by a known contact force in the barrel of the bolt.

A generalized concept for **bolt pretension** is provided. Beside the classical approach using a cut-



ting plane with pretension in normal direction, a new approach using cylindrical thread coupling with pretension in axial direction is available. This highly innovative feature offers a convenient definition which can take into account the detailed effects of radial spreading and axial torque caused by the thread's flank and pitch geometry without the need of modeling the flank shape or thread line explicitly (see Fig. 99).

Contacting bodies are often only supported by contact forces. To provide an elastic foundation to such bodies, a family of *compensation springs* (also called *zero force springs*) can be used, where the spring force is compensated by an additional contact force (see Fig. 10). By this means, a *force-guided contact* is facilitated. In addition, the *RBM Assistant* in VisPER helps to place and properly define such elements (see page 40).

Comprehensive checks allow the verification of contact models like type of contact, its geometry (gap-width and normal vector, see also Fig 89), and the contact coordinate system (for normal and frictional force directions). In addition, the contact status is available in all iteration steps for checking purposes.

For frictional contact the quality of surfaces is of utmost importance. Therefore, PERMAS can smooth contact surfaces in order to essentially improve frictional behavior.

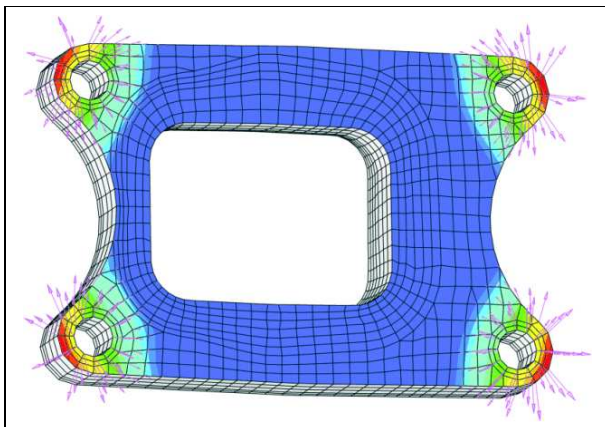


Figure 100: Contact pressure and shear vectors

The analysis procedure uses a reduced flexibility model which is derived from the set of contact degrees of freedom. This procedure has the following advantages:

- The iteration is very efficient making it best suited for extremely large models with an arbitrary number of contact nodes.

- The accuracy of the results is fully preserved, because no additional stiffnesses are introduced by the modeling of contacts.

The simultaneous analysis of an **arbitrary number of loading cases** is possible. The contact parameters, i.e. gap width and coefficients of friction, may be different for each loading case. The contact boundary conditions are taken into account automatically by the static analysis procedure. No additional user request is required for a contact analysis.

For efficient calculation of successive contact variants *contact status files* are available for easy job recovery and considerable run time reductions.

In addition to all results usually derived from a static analysis the contact analysis provides for the contact status, the contact forces, the contact pressure (see Fig. 100), the gap widths, and the relative gap displacements.

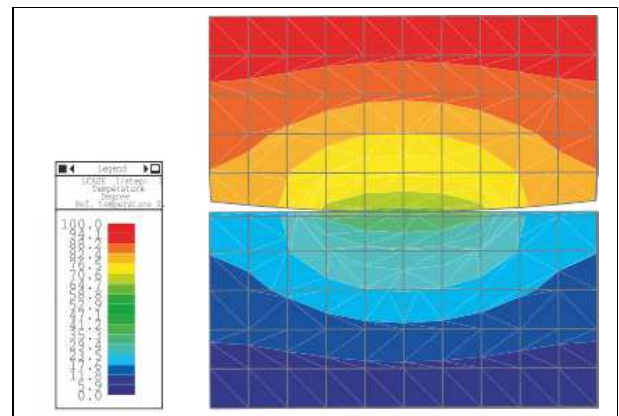


Figure 101: Temperature distribution in two contacting bodies

For subsequent analyses, contact states can be locked. This *contact locking* leads to linear constraints according to the current contact state. To achieve this, the active contacts are automatically transformed into kinematic constraints. With this new model various kinds of subsequent analyses are possible like eigenvalue analysis, heat transfer analysis (see Fig. 101), or *submodeling*.



## PERMAS-CAX – Extended Contact Analysis

This module has been designed to provide new contact solution algorithms for ambitious slip-stick problems and for large contact models (more than 10,000 contact node pairs) in order to essentially accelerate contact analysis runs again (by a factor of 3 and beyond). The module is used as an add-on to module PERMAS-CA (see preceding section).

The functionality comprises the following features:

- High performance iterative solution algorithms to accelerate standard contact analysis with normal and frictional contact.
- An additional very stable iteration method for critical slip-stick problems (e.g. if all contact pairs get into sliding state, see Fig. 102).

*Gasket elements* are handled as integral part of the contact iteration through contact controlled non-linear gasket analysis (*CCNG analysis*) by default. This leads to a remarkable reduction of run time compared to the classical solution method with non-linear material behaviour. In particular, this run time reduction will be much higher, when the nonlinear features are contact and gasket elements only. The run time reduction is still significant, if additional nonlinearities are applied.

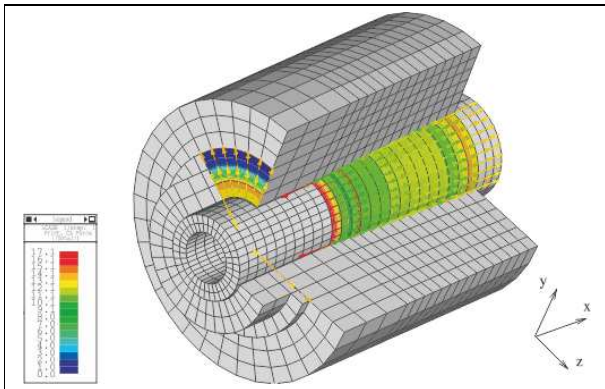


Figure 102: Conical *press fit*  
(slipping forces after assembly)

For sliding friction between dynamically moving parts a velocity field can be prescribed to take it into account in a quasi-static contact analysis (see section Brake Squeal Analysis on page 28).

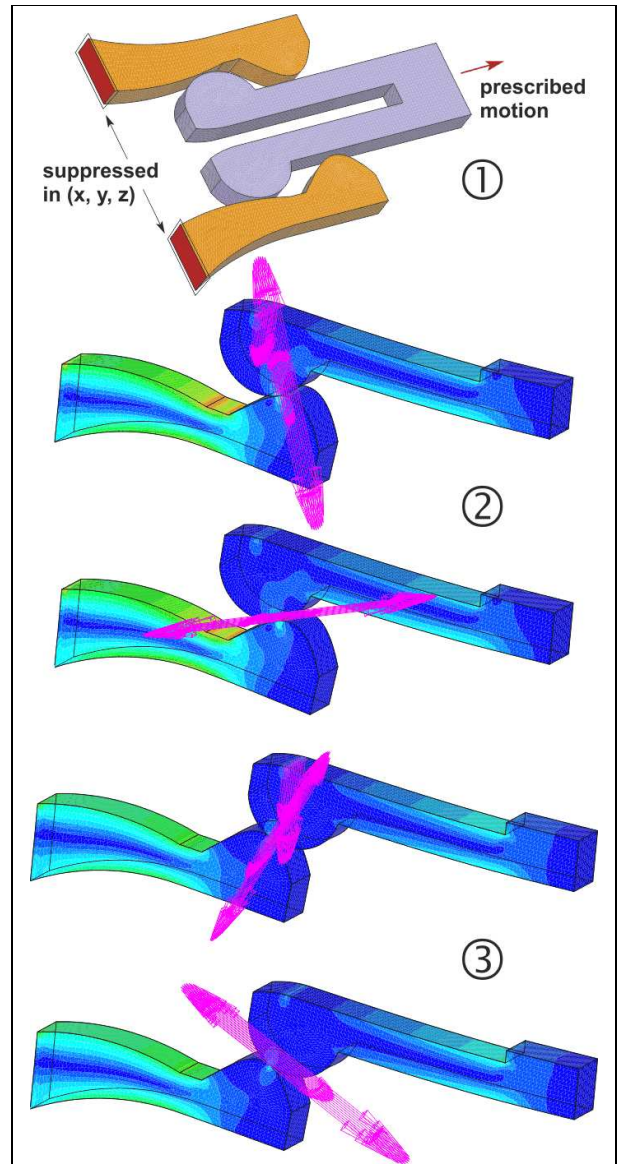


Figure 103: Clip example for contact geometry update

Large relative motion with friction. Picture 1 shows the model; pictures 2 show the contact pressure (above) and the frictional shear (below) with nodal point stresses before the bottleneck is reached; pictures 3 show the contact pressure (above) and the frictional shear (below) with nodal point stresses after the bottleneck is passed.

## PERMAS-CAU – Contact Geometry Update

This module has been designed to provide new contact solution algorithm for significant changes of contact geometry during loading. The contact geometry is characterized by the nadir points (i.e. the projections of the contact points of one body to the

the contact surface of the other body) and the contact direction (i.e. the normal to the surface at each contact node). The following application cases require the use of contact updates:

- In case of large relative displacements of two bodies in contact, the update of the contact properly reflects the change of the relative position of both bodies (see Fig. 103 and Fig. 104).
- If very accurate contact results are required, contact updating even reflects very small changes of the contact geometry which gives exact contact forces, displacements, stresses, etc..

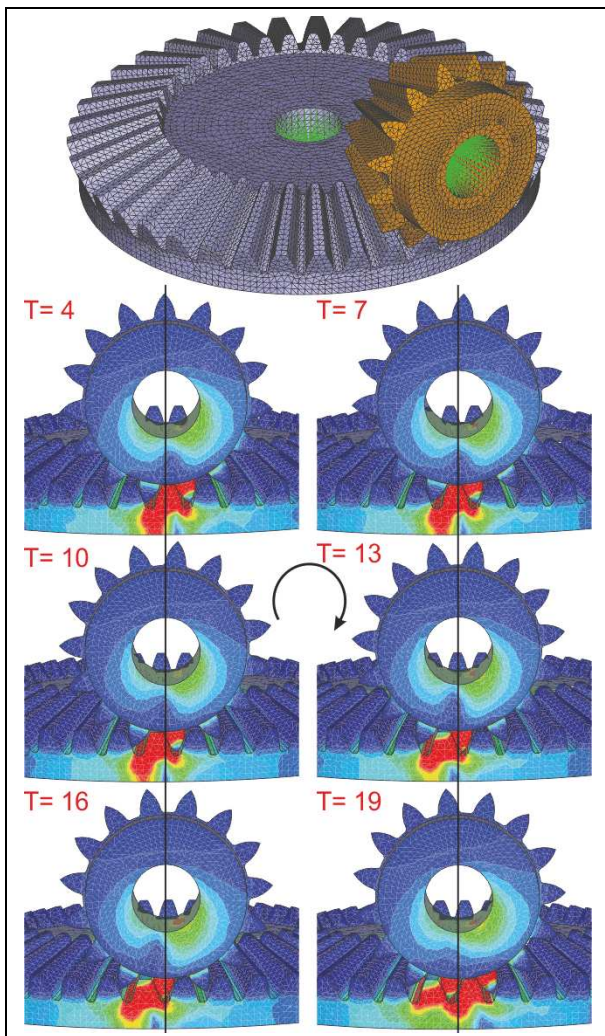


Figure 104: Driven gear set

The following features are supported by this module:

- Special modeling is not required. Because the update process changes the position of the contacting surfaces, the contact geometry must be specified by surfaces (e.g. surface-to-surface or surface-to-node). Hence, node-to-node contact

and contact to ground are not updated.

- Contact updating works with and without friction.
- Contact updating works with linear and nonlinear static analysis. For nonlinear applications, some modeling features are still restricted.

The geometry update is principally explained in Fig. 105. Steps 1 and 2 show the typical contact procedure and steps 3 and 4 show the first iteration of the updating process which will be repeated until a given number of iterations is performed or until a given accuracy is achieved.

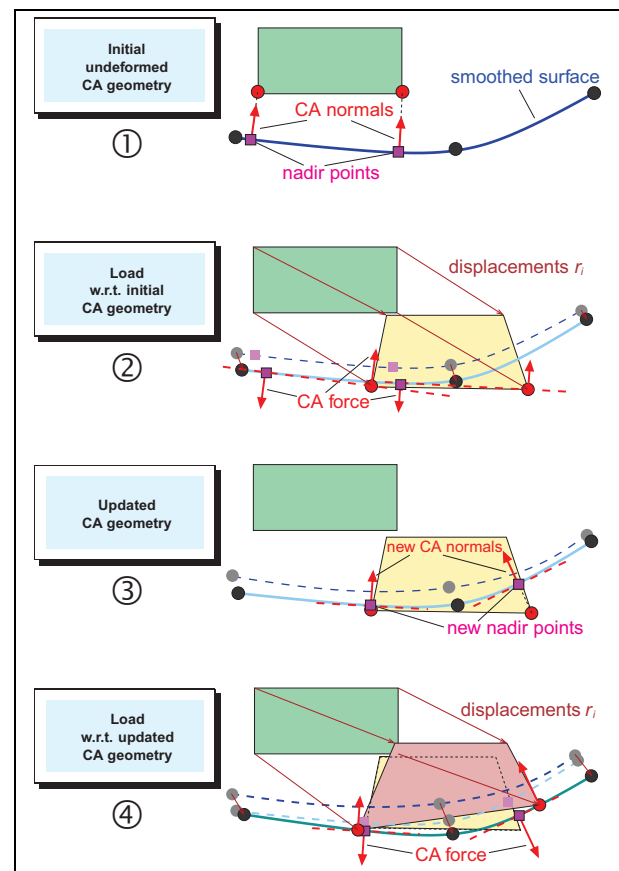


Figure 105: First steps in contact update

## PERMAS-NLS – Nonlinear Statics

### Geometrically nonlinear behavior

This module part allows for the geometric nonlinear analysis of models. There, large displacements with small strains (i.e. linear elastic material behavior) are assumed. Beside an automatic load step control different nonlinear solvers are available (like

*Newton-Raphson*, Modified Newton-Raphson, Secant Newton, Line search, Arc length method). The nonlinear characteristic may be represented by XY plots, where the load increments can be chosen automatically or manually.

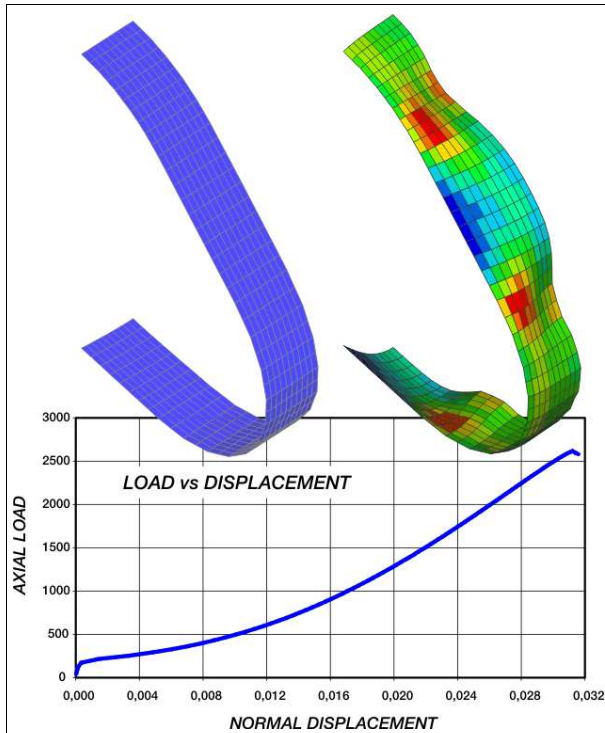


Figure 106: Nonlinear NAFEMS Test

### Linear but temperature-dependent material

When temperature is changing during a nonlinear load history, then the update of the stiffness and load matrices requires a nonlinear solution approach. If material and geometric nonlinearities can be neglected, a special algorithm is available which has a high potential for saving run time compared to a fully nonlinear solution. This approach can be combined with nonlinear gaskets using *CCNG analysis* (see page 69) in particular for efficient analysis of combustion engines (see engine example in Fig. 12).

### Material nonlinearities

This part of the module allows the analysis of nonlinear material behavior of models with small strains under isotropic conditions:

- Nonlinear elasticity (of *Cauchy* type)
- Plasticity (*von Mises*, *Tresca*, *Drucker-Prager*, *Mohr-Coulomb*)
- *Visco-plasticity* of power-law type for von Mises

yield criterion.

- Creep with
  - nonlinear elasticity or
  - plasticity

The material can be defined temperature-dependent for Young's modulus, yield stress, and the stress-strain curves. A time-dependent characteristic is present for creep calculations in addition. *Hardening* in plasticity can be defined *isotropic* or *kinematic* or *mixed*.

For the use of shell elements with material nonlinearities, an element family of elements with a 3-dimensional shell formulation is available, which is applicable for linear analyses, too. This element family (triangles and quadrangles with linear and quadratic shape functions) has been designed for nonlinear analysis with already existing shell models.

For the modeling of gaskets a family of *gasket elements* is available. These elements are used to define the nonlinear behavior in a preferential direction by a measured force-displacement curve of the real gasket.

An incremental and iterative solver strategy is based on *Newton-Raphson*, Modified Newton-Raphson, and *Thomas* method. An automatic load step control allows for an optional specification of initial load step and total applied load (or time). The material laws may be defined either in tabular form or as user-written subroutine (Fortran or C).

Applications using *inertia relief* (see page 66) can also take nonlinear material behavior into account.

Structural behavior can be influenced by any pre-treatment (like *casting*, *rolling*). The resulting internal strains can be used as initial conditions (without displacements).

*User defined material* laws can be used. User subroutines allow the incorporation of own material laws. The subroutine does the necessary calculation of stresses and strains together with the tangent matrix associated with the material law.

### Combination of material and geometrical nonlinearities

Analyses with material nonlinearities can take into account the geometrical nonlinear effects, too.



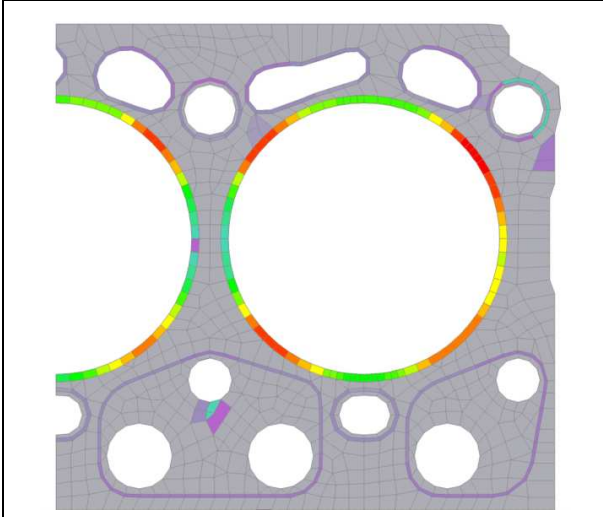


Figure 107: Pressure of cylinder head gasket

There, also follower loads like pressure loads, temperature loads, and inertia loads can be taken into account.

An application example is post-buckling of curved shells (see Fig. 13). This allows the static analysis beyond plastic buckling. In addition, after nonlinear analysis a linear buckling can be performed to study all possible buckling modes at this point of loading (see page 73).

### General

In case of contact definitions the nonlinear analysis takes them into account automatically performing a nonlinear contact analysis.

Extrapolation of stresses for linear and nonlinear elements is fully provided. For linear elements, stresses are calculated at the stress points and either assigned or extrapolated to the element corner nodes. For nonlinear elements, stresses are calculated at the integration points and either assigned to the element nodes or extrapolated to the element corner nodes. If stresses are not assigned to the element midside nodes, these stresses will be calculated through linear interpolation from the element corner nodes. In plasticity analysis, stresses at the element nodes are showing stresses at the integration points or they are showing extrapolated stresses which do not lie on the yield surface any more. In case of subsequent generation of nodal point stresses in post-processing, the original element stress results can be seriously modified.

Initial states like for rotating structures can be taken

into account in nonlinear analyses.

The results of a nonlinear analysis may be used for subsequent analysis like a dynamic mode analysis.

In many cases, the major part of the model is linear. This is an ideal prerequisite to apply *substructure technique* (see page 49), where the linear parts are put in subcomponents and all nonlinear parts are put in the top component. This procedure will lead to a significant reduction in run time.

If many loading cycles have to be calculated for realistic models (e.g. with creep in engine models), then a recommendable strategy is to use *submodeling* to reduce the the model to the area of interest. The results of the global analysis are mapped to a sub-model (see section on submodeling on page 49). Then, the nonlinear analysis will be performed for the submodel. Run times are reduced dependent on the size of the submodel and may allow for time consuming nonlinear analysis.

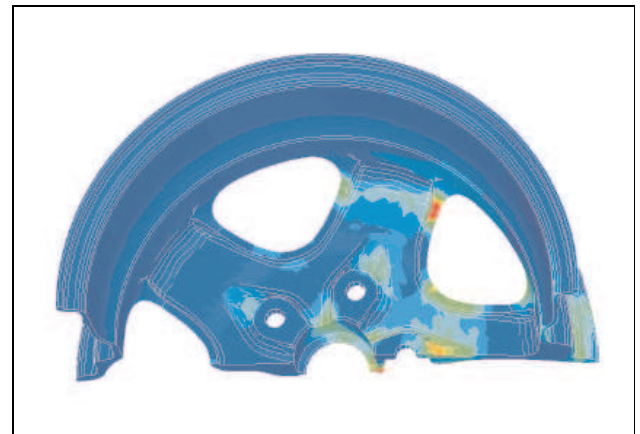


Figure 108: Impact test of a wheel with effective plastic strain.

## PERMAS-NLSMAT – Extended Material Laws

This module comprises a number of additional material laws to complement the standard material laws in module NLS:

- A material law for *cast-iron* is available taking into account the different behavior under tension and compression (see Fig. 109).
- Furthermore, a *nonlinear kinematic hardening*



model (following *Armstrong-Frederik*) is available as a model for cyclic loading.

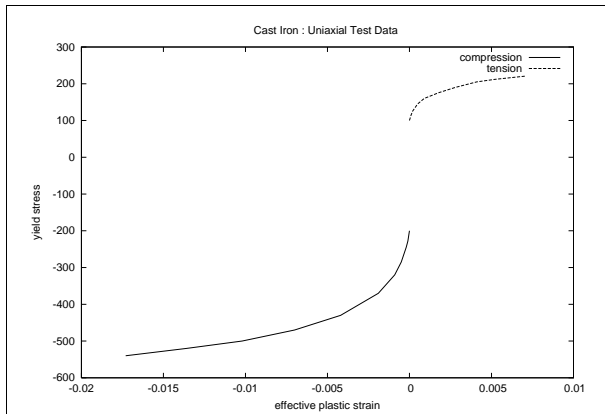


Figure 109: Uniaxial test data of a cast iron material

## PERMAS-BA – Linear Buckling

Based on a linear static analysis the related buckling modes with load factors and mode shapes can be determined.

The calculation of *modal participation factors* allows for the assessment of the nonlinearity of the *pre-buckling behavior*.

Load factors and mode shapes are available for any kind of post-processing.

## PERMAS-DEV – Dynamic Eigenvalues

The PERMAS-DEV (Dynamics/Eigenvalues) module provides for the calculation of *real eigenvalues* and mode shapes of the structure (*modal analysis*). The specification of a number of modes and an upper frequency limit is supported. The very efficient *subspace iteration* algorithm used is capable of solving very large eigenvalue problems. Rigid body modes are detected automatically or may be explicitly defined and are decoupled prior to the subspace iteration.

If the number of modes is increasing, one can observe a disproportional increase of solution time, because the last modes take more time than the first modes. Therefore, a *shift method* has been intro-

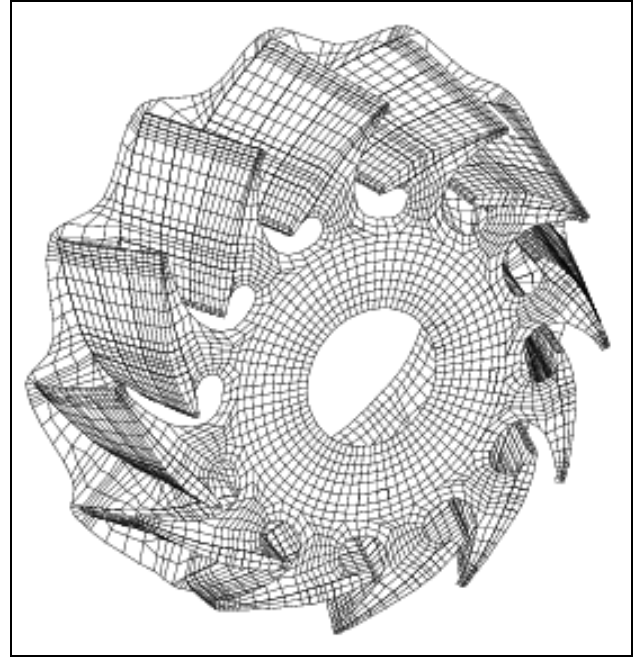


Figure 110: Half Model of a Turbine  
Half Model of a Turbine, 8th Mode Shape with  
Antisymmetric Boundary Conditions

duced which are used to split the frequency domain in several parts and to solve each part separately. Then, even the last modes can be solved as fast as the first modes resulting in an overall run time reduction for the complete analysis. The shift method is used automatically by PERMAS in cases, where a large number of modes is required (> 5000 modes) or where the standard method takes a long time to converge. In addition, a higher accuracy of the mode shapes can be expected from this method.

The stiffness matrix can be modified taking into account additional stiffness effects:

- *Geometric stiffness* for any load,
- *Centrifugal stiffness* for rotating parts under constant rotational speed referring to co-rotating reference system,
- *Convective stiffness* for rotating parts under constant rotational speed referring to inertial reference system,
- *Pressure stiffness* for shell elements and fluid-filled pipe elements under pressure.

Additional tools are available for the further processing of modes:

- Modal stresses can be derived from modal displacements.
- In addition, modal potential and kinematic ener-

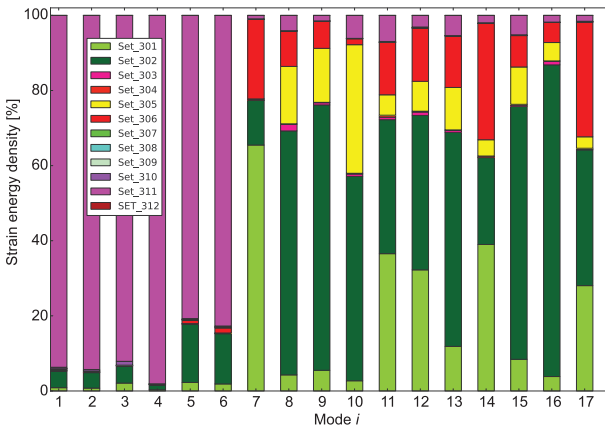


Figure 111: Analysis of strain energy density in element sets and for various modes

gies can be calculated and exported.

- For the evaluation of modes, e.g. with respect to local or global mode shapes, energy balances can be determined and exported for all sets in a structure.
- **MAC** (Modal Assurance Criterion) factors and other factors are available to compare modes between two different modal analyses (see page 62).
- As a measure for the completeness of the modal model, **effective masses** are generated and printed on the result file.

A **generalized modal condensation** is available to establish system matrices in modal space for external applications. Export of modal models is either supported by interfaces (e.g. to MBS systems) or by direct specification of the matrix items.

## PERMAS-DEVX – Extended Mode Analysis

This module provides additional methods for dynamic eigenvalue analysis:

- **Dynamic condensation**
- Complex mode analysis
- Eigenfrequencies over rotational speed for rotating structures

### Dynamic Condensation

Two methods are available for dynamic **condensation**:

- Craig-Bampton method (CB condensation),

- Mixed-Boundary Craig-Bampton method (MBCB condensation).

#### *Craig-Bampton method:*

The **Craig-Bampton** method uses fixed-interface vibration modes and the static deflections due to unit displacements of the interface degrees of freedom for the dynamic reduction of substructures (**CB condensation**). Like for the **Guyan's** reduction, an explicit and an iterative scheme is available in order to achieve good performance (see also page 49).

The functionality may be summarized as following:

- **Supported solutions**
  - Structural dynamics
  - Acoustics
  - Coupled fluid-structure acoustics

Two **condensation** options are available for coupled fluid-structure acoustics (see also page 79):

- **“Dry” Interface**
  - Solution of a coupled eigenvalue problem on subcomponent level, i.e. **isolation** of the acoustic component. External modes are coupled modes.
  - Global solution may be a mechanical vibration analysis.
- **“Wet” Interface**
  - Separate computation of mechanical and acoustic modes on subcomponent level.
  - Global solution is a coupled vibration analysis.
  - Condensation of the fluid-structure interface can also be made.

#### *Mixed-Boundary Craig-Bampton method:*

The method allows for free (or mixed) boundary conditions to derive the vibration modes (**MBCB condensation**). The corresponding static deflections are derived using **inertia relief**. This method is advantageous when the condensation is used for structures with free boundaries, because the condensation method can represent the boundary conditions. In such cases, less number of required modes and a higher accuracy can be expected from the Mixed-Boundary Craig-Bampton method compared to the classical Craig-Bampton method.

The above mentioned “dry” condensation of parts with enclosed fluids also works with the Mixed-Boundary Craig-Bampton method.

### Complex Mode Analysis

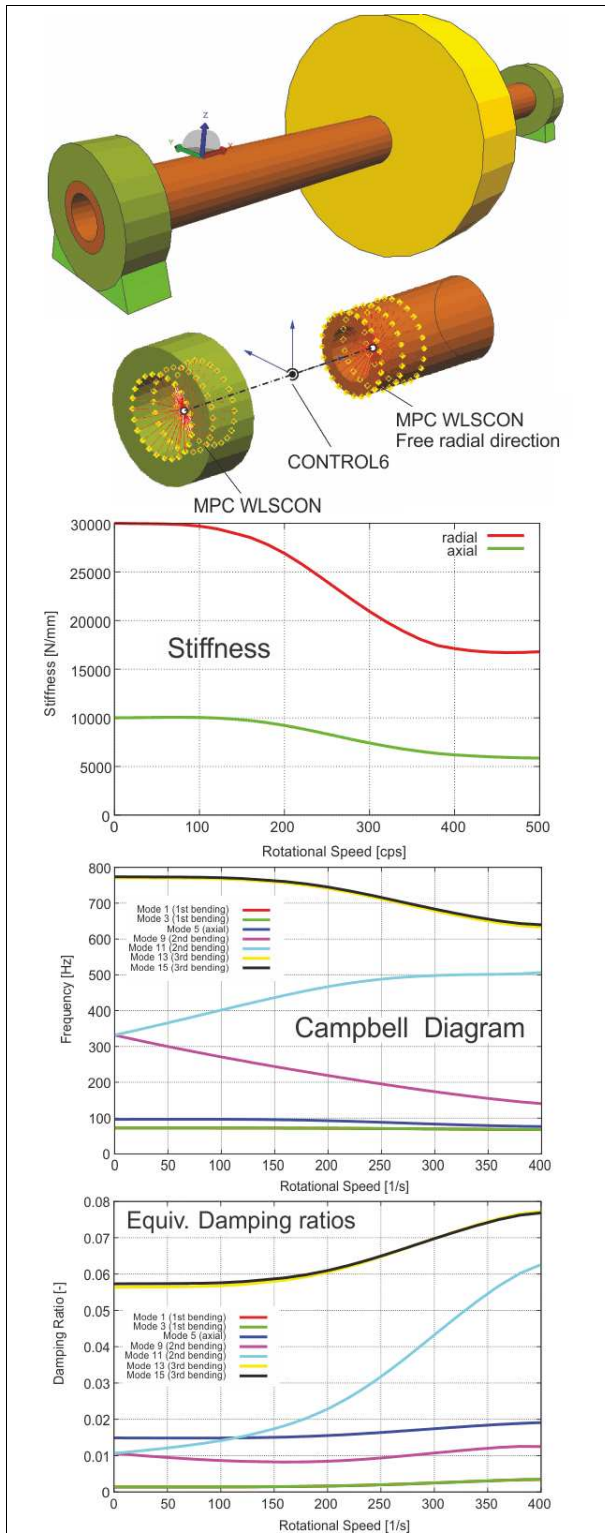


Figure 112: Rotor  
with rotation speed dependent supports

This includes the calculation of complex eigenvalues and eigenvectors in modal coordinates. This method is based on a previous solution of the real eigenvalue task.

The results of this analysis are as follows:

- Frequencies
- *Complex eigenvalues*
- Complex eigenfrequencies (each with damping coefficient and circular frequency)
- Equivalent viscous damping ratios
- Complex mode shapes with physical and modal representation. The modal displacements of the complex modes represent the modal participation of the underlying real modes.

A suitable post-processor (like VisPER) can be used to visualize and animate complex mode shapes

### Eigenfrequencies of Rotating Systems

For rotational systems (see also page 30) it is often required to generate a so-called *Campbell* diagram, which relates the eigenfrequencies to the rotational speed. The values of such a diagram can be generated automatically in one single run. From these values all frequencies of interest can be selected for a subsequent frequency response analysis.

For rotating structures, any number of rotational speeds is defined in a separate input. A reference rotational velocity is used in the static pre-run. From this pre-run, additional matrices are built for the reference rotational velocity. The specified rotational velocities are used to scale the additional matrices during dynamic response analysis. This procedure makes the generation of Campbell diagrams very efficient and the response analysis of rotating structures is facilitated.

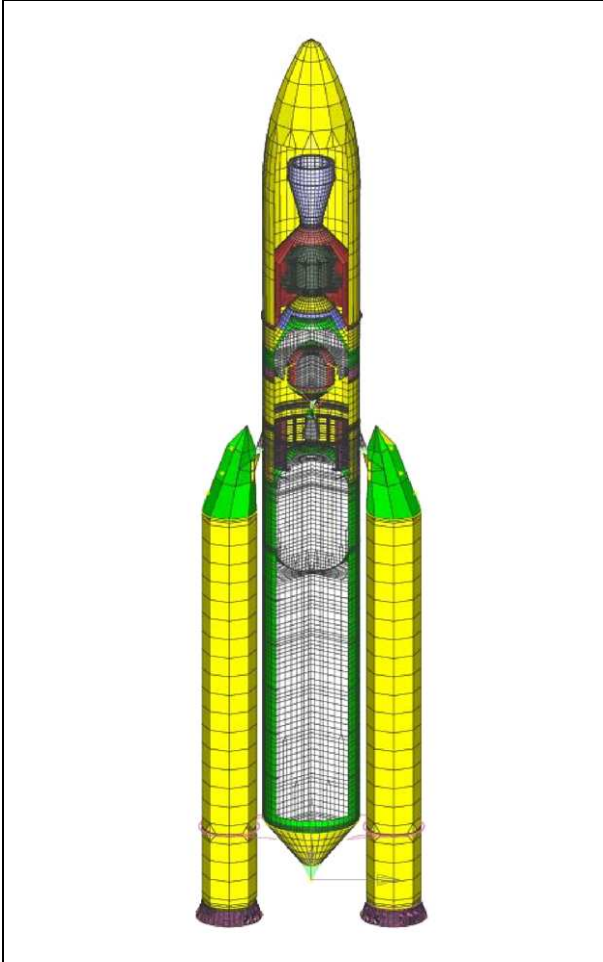


Figure 113: Ariane 5 launcher model  
by courtesy of ASTRIUM Space Transportation, Les  
Mureaux

## PERMAS-MLDR – Eigenmodes with MLDR

The calculation of eigenvalues with modules DEV (page 73) and DEVX (page 74) is complemented by another method. This method can also be used for the calculation of coupled fluid-structure modes.

The application of this method is advantageous in those cases where the elapsed run times are mainly determined by I/O like for large models with a high number of modes to be calculated. The larger the models and the larger the number of modes, the higher is the benefit in elapsed run time through the application of the MLDR method.

This benefit in elapsed run time can be essentially raised more, if there is a subsequent dynamic response analysis calculating the response behavior just at a small number of nodes. Then, the genera-

tion of the global mode shapes can be saved resulting in considerable computing time savings.

The MLDR method is based on an automatic partitioning of the model where each part does not exceed a preset quantity. In addition, the coupling between the parts has to be as low as possible. These parts are then groupwise combined as substructures using *dynamic condensation* (see substructuring on page 49 and module DEVX on page 74). This procedure is hierarchically carried on until the complete model is represented in one component. In this component only a small number of nodes and elements remain and the dynamic behavior is mainly determined by the modes and frequencies taken over from the substructures and combined following the rules of dynamic condensation. Due to this procedure the method's name is **Multi-Level Dynamic Reduction (MLDR)**

If certain nodes and elements should be present in the main component, the user can specify them explicitly. So, selected model parts can be pushed to the main component and any subsequent processing of the modes is rather beneficial due to the small size of the remaining matrix system. In this way, dynamic simulation, coupling to MBS, optimization of the remaining system, or the consideration of nonlinearities can be performed with very low computing times.

Additional reductions of computing time are possible using multi-processor systems, because the method has been fully parallelized. Altogether, the use of MLDR is a big step forward to more productivity and allows, for example, dynamic simulation in a higher frequency range as in the past together with a possible increase in model size for more accurate results.

## PERMAS-DRA – Dynamic Response

The PERMAS-DRA (Dynamic Response Analysis) module allows for the determination of structural responses in the time or frequency domain.

The solution of the dynamic equation is performed either directly using physical coordinates or in modal coordinates after a transformation into the modal space.

- The **response in the time domain** (*transient re-*



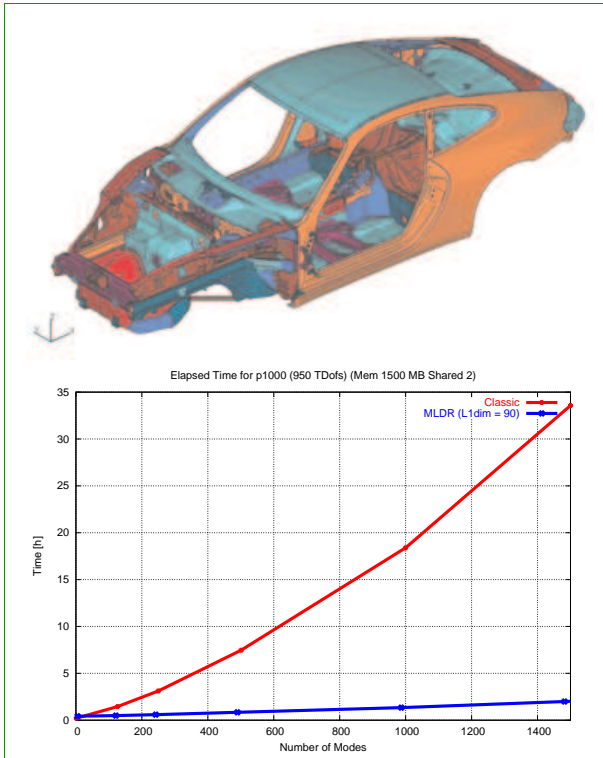


Figure 114: Comparison of elapsed run times for Subspace Iteration (upper curve) and MLDR (lower curve) with increasing number of modes (157412 Nodes, 164301 Elements (QUAD4), 944472 Unknowns)

*sponse*) is determined by an integration of the equation of motion:

- Absolute transient response with or without rigid body response.
- Direct integration of the equation of motion or integration after a transformation to the modal space. Available solvers are *Newmark  $\beta$*  and *HHT (Hilber-Hughes-Taylor)*.

Local nonlinear effects are taken into account by

- nonlinear spring elements,
- nonlinear damper elements, and
- nonlinear control elements.

- The **response in the frequency domain** (*frequency response*) is determined by the solution of the linear complex equation system for each excitation frequency requested:
  - Absolute frequency response with or without rigid body response.
  - Direct solution of the equation system or solution after a transformation to the modal space.

*Frequency and rotation speed dependent stiff-*

*ness and damping* can be taken into account with CONTROL6 element (see examples in Fig. 115 and Fig. 112).

- Without running through the transient phenomenon a calculation of the **steady-state response** can be performed. To achieve that, a number of frequency response analyses are superimposed in the time domain. In addition, a static load case can be taken into account. This is facilitated for all periodic excitations with known harmonic composition.

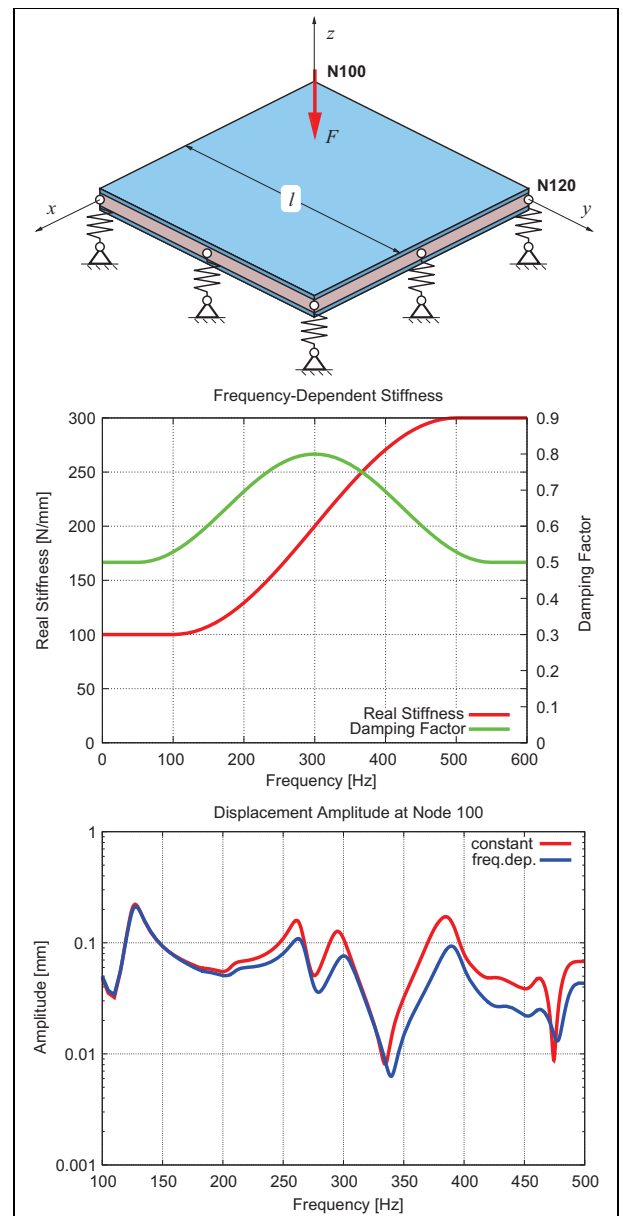


Figure 115: Frequency-dependent stiffness and frequency dependent viscous damping in frequency response analysis

By specification of a node set (see page 57) the run time and disk space for modal superposition methods can be drastically reduced when the results are determined for the set members only.

The features below hold for both time-history and frequency response methods:

- The **damping** properties may be modeled by one of the following methods:
  - material or structural damping for elements,
  - global structural damping for components,
  - proportional damping (Rayleigh damping),
  - viscous damper elements,
  - modal viscous damping,
  - modal structural damping,
  - modal structural and proportional damping for subcomponents,
  - direct input of modal damping matrix ,
  - direct input of damping matrix.

For calculations in frequency domain, the structural damping may be defined as a function of frequency, alternatively.

Frequency-dependent viscous damping parameters can be specified using a special element (i.e. CONTROL6 element).

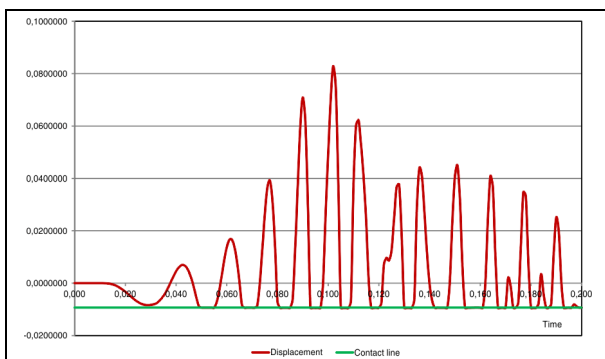


Figure 116: Transient response under contact conditions

- The excitation is defined by static loading cases modulated by functions of time or frequency respectively (see page 58). The load definitions may consist of:
  - concentrated forces or moments,
  - distributed loads (loads applied to lines, surfaces or volumes),
  - inertia loads,
  - prescribed displacements.
- The primary results are:
  - displacements,

- velocities, and
- accelerations.

For the drawing of XY plots, these data may be output as a function of time or frequency respectively.

- Moreover, the following results may be derived:
  - reaction forces,
  - stresses and stress resultants,
  - strain energy,
  - kinetic energy,
  - specific sound radiation power density.

When modal methods are applied additional functions are available:

**Static Mode Shapes** can be generated in order to enhance the modal basis. This has the following two advantages:

- In the low frequency range the results will become essentially more accurate.
- They allow to consider non-structural degrees of freedom like the internal state variables of control elements.

The static mode shapes can be specified using one of the following cases:

- directly by nodal displacements,
- by external loads,
- by results from another analysis,
- by natural loads of specified elements (like spring forces),
- implicitly by internal degrees of freedom of controller elements.

**Assembled situations** can be used to highly accelerate frequency response analysis with many different load cases. Instead of solving all dynamic load cases separately, a combined response analysis can be performed.

Additional tools are available for the further processing of modal results:

- For the evaluation of a structural response *modal participation factors* of the primary results can be derived and exported.
- For the evaluation of the contribution of single degrees of freedom to a structural response node participation factors can be derived and exported.
- For a transient response a statistical evaluation of stresses or element forces over the time domain can be performed, which gives the maxi-

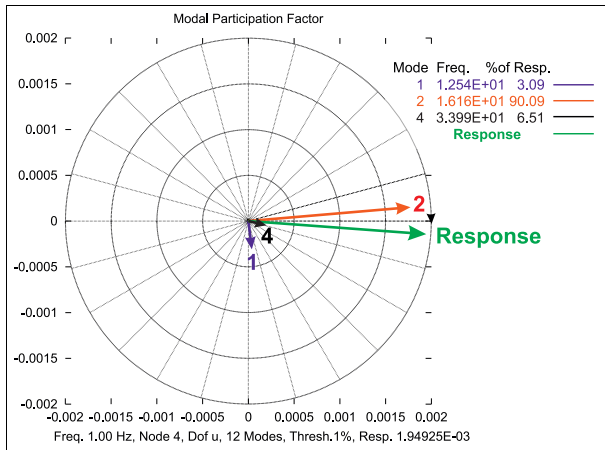


Figure 117: Modal grid participation factors

mal and effective values. These statistical values can be used in durability considerations, for example.

## PERMAS-DRX – Extended Dynamics

This module comprises additional methods for structural response analysis:

- Spectral Response Analysis (or *earthquake spectral response* Analysis),
- Random Response Analysis.

### Spectral Response Analysis

In case of a prescribed ground motion, like in earthquake analysis, the dynamic response behavior is determined by a special method, which results in maximum response values. There, the following requirements have to be fulfilled:

- uni-directional and translational motion of the ground,
- no other loads,
- analysis can be performed in modal space,
- only modal viscous damping.

After the specification of the direction of the ground motion the load is defined by a spectrum of the transient excitation (*response spectrum*). Then the analysis is performed as follows:

- Calculation of all dynamic modes up to the highest interesting frequency.
- Calculation of the maximum mode contributions.
- Summation of the maximum contribution factors using one of 7 available *summation rules* (like

*CQC or 10% rule*).

- Export or print of peak values.

## Random Response Analysis

Frequently, vibrational loads are not predictable like for cars on a bumpy road, for a house under wind loads, or for a ship on rough sea.

Conveniently, such stochastic loads can be described by random processes. Correlations between such processes and their transformation into frequency domain are leading to the central concept of *power spectral density*.

One special phenomenon is white noise which describes a constant power spectral density over the full range of frequencies.

The loads are specified as power spectral densities and the results are derived as RMS quantities and power spectral densities as well.

This method is implemented as a modal method, i.e. an eigenvalue analysis is performed first followed by the response analysis in modal space and a subsequent back transformation into physical space where the results are made available for export and post-processing.

## PERMAS-FS – Fluid-Structure Acoustics

The PERMAS-FS (Fluid-Structure Acoustics) module allows for the calculation of acoustic modes, as well as eigenvalues of coupled fluid-structure systems and the computation of coupled or uncoupled response in the frequency or time domain. This computation is provided either directly or in modal coordinates. Fig. 118 shows an overview on the analysis capabilities for fluid only and for coupled fluid and structure.

The fluid may be compressible or incompressible. All fluid absorption or damping properties are also available for the uncoupled computation. The fluid damping may be frequency dependent.

A number of special features is provided to model boundary conditions:

- Surface waves are modeled by specific elements.

Acoustics	Fluid only	Fluid coupled with structure
Degrees of freedom	pressure variation	+ displacements
Element types	fluid, radiation bound. cond., surface waves, semi-infinite elements for added masses	+ coupling, structure
Material values	density, compressibility (also incompressible)	+ density, Young's modulus, Poisson's ratio
Real eigenvalues and eigenmodes	yes	coupled, with energy distribution
Additional static modes	yes	yes
Dynamic Condensation	Craig-Bampton	dry (only for displacement dof) or wet (also pressure dof)
Damping	absorption at surfaces and in fluid volume	+ any structural damping
Modal dynamic response	in frequency and time domain	in frequency and time domain
Direct dynamic response	in frequency domain	in frequency domain
Primary results	pressures	+ displacements, velocities, accelerations
Secondary results	sound particle velocity	+ stresses, strains, etc.

Figure 118: Overview of acoustic capabilities for fluid only and for fluid coupled to a structure.

- Special coupling elements are provided at the boundary of the fluid to the structural model. These elements are also used to model surface absorption. In addition, another acoustic damping facility is available through volumetric dampers (like seats in a car).
- Semi-infinite elements are provided to handle an infinite surrounding space.
- Radiating boundary conditions (RBC) can be modeled using special element families, one following the theory of *Bayliss-Turkel* and another the theory of *Engquist-Madya*.

For the coupling elements mentioned above the face normal has to be oriented from the fluid to the structure. This condition is checked automatically in order to avoid conflicts and sources of mistakes in the coupling of fluid and structure.

For the calculation of dynamic mode frequencies, a difference is made between the fully coupled modes and the structural modes modified by the additional fluid mass:

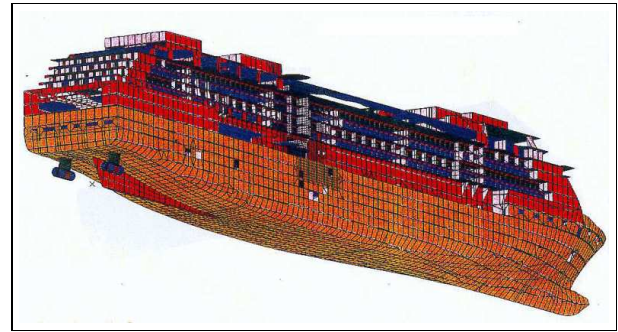


Figure 119: Ship model  
by courtesy of Chantiers de l'Atlantique, Saint-Nazaire

- The calculation of real eigenvalues and mode shapes of the coupled structure is realized by a simultaneous vector iteration. The specification of a number of modes and an upper frequency limit is supported. The special formulation of the algorithm used is capable of solving very large eigenvalue problems in an efficient way.
- The added mass problem can be solved with the standard structural eigenvalue solver (see page 73), where the mass of the fluid is taken into account to calculate the structural modes.

The runtime for a coupled eigenvalue analysis in case of large models with a high number of modes can be drastically reduced by the MLDR method (see page 76).

For the calculation of the dynamic response behavior, the following methods are available:

- The response in the **time domain** (transient response) is determined by a modal solution of the equation of motion. Available solvers are *Newmark  $\beta$*  and HHT (*Hilber-Hughes-Taylor*).
- The response in the **frequency domain** (frequency response) is determined by the modal or direct solution of the linear complex system of equations for each excitation frequency requested. In general, a fully populated equation system has to be solved. In modal space an iterative solver makes the analysis much faster.
- Additional *static mode shapes* can be taken into account.

By specification of a node and/or element set (see page 57) the run time for modal superposition methods can be drastically reduced when the results are only determined for the set members. The **reduced** response results in enormous disk space savings.



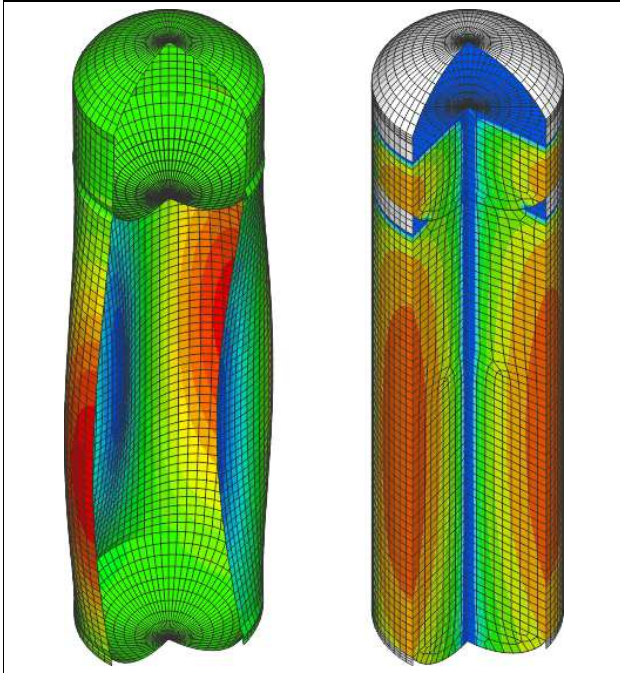


Figure 120: A coupled vibration mode shape of a fluid-filled space tank with corresponding pressure mode

The following damping features are available:

- For the structure:
  - material or structural damping for elements, also frequency dependent, if required,
  - global structural damping,
  - Rayleigh damping.
- For the fluid:
  - boundary absorption,
  - volumetric absorption (also frequency-dependent).
- For the coupled system:
  - modal viscous damping (coupled system).

The excitation is defined by static loading cases modulated by functions of frequency (see page 58).

The load definitions may consist of:

- Structural loads as described for the DRA module
- prescribed pressures.

The primary results are:

- displacements,
- pressures,
- velocities, and
- accelerations.

For the drawing of XY plots, these data may be out-

put as a function of time or frequency respectively. Moreover, the following results may be derived:

- reaction forces,
- stresses and stress resultants,
- strain energy,
- kinetic energy,
- specific sound radiation power density,
- sound particle velocity.

## PERMAS-NLD – Nonlinear Dynamics

This module provides time integration in structural analysis including nonlinearities:

- Material nonlinearities like creep, nonlinear elasticity, plasticity, and visco-plasticity.
- Nonlinear elements like nonlinear springs or gasket elements as well as control elements.
- Large translational motions of elastic bodies handled by updating MPC conditions (with incompatible meshes).

Geometric nonlinear effects and contact are not yet included.

Time integration is done using the *Newmark* method or the generalized  *$\alpha$ -method*. The latter includes numerical damping to stabilize the integration scheme.

Different solution methods like *Newton-Raphson* or modified Newton-Raphson are available. An automatic time stepping technique supports the use of appropriate time steps.

*Substructure technique* and *dynamic condensation* can be used to reduce the purely elastic parts before the nonlinear dynamic analysis starts.

## PERMAS-HT – Heat Transfer

*Temperature fields* will be analyzed using the modules PERMAS-HT and PERMAS-NLHT (see next section).

- The temperature field may be steady-state or transient.
- Nonlinear material data for conductivity and heat capacity may be specified by tabular input.

- Temperature and space-dependent convectivity coefficients may be specified in a very general way by functions (see page 58).

PERMAS-HT provides a complete set of **convectivity elements** to model surface convection. In addition, they allow for the determination of the surface area of a set of elements in order to get the drained or injected heat through the related surface. Moreover, an optional film thickness may be specified for the convectivity elements, which allows to model the heat capacity of boundary layers in transient analysis.

All finite elements applicable in static analysis may be used in heat transfer analysis. For shell elements a temperature gradient between top and bottom surface is allowed.

Coupled analysis of thermo-mechanical problems is fully automatic, i.e. the resulting temperature field is directly used to derive the related displacements, strains and stresses. The material data for the static analysis (elasticity and thermal expansion data) may be temperature-dependent.

In addition, using PERMAS-CCL convective boundary conditions can be imported from a CFD analysis or fully coupled analyses can be performed, respectively (see page 64).

A transient analysis may be continued by reference to the results of the previous run. Among others, during the simulation of complete cycles of thermal loads this feature easily allows for sudden changes of the surrounding conditions.

A **modal analysis** allows for the computation of eigenvalues and eigenmodes for heat transfer problems.

Available loads and boundary conditions:

- Stationary 'loads' may be defined as point heat fluxes or distributed heat fluxes along lines, on surfaces and in volumina.
- Transient 'loads' are built conveniently by combination of a stationary 'load pattern' with time-dependent functions (see page 58).
- Additional boundary conditions are prescribed temperatures and a surrounding temperature for convectivity elements.

Primary results of a heat transfer analysis are the temperature field and the heat fluxes. In addition,

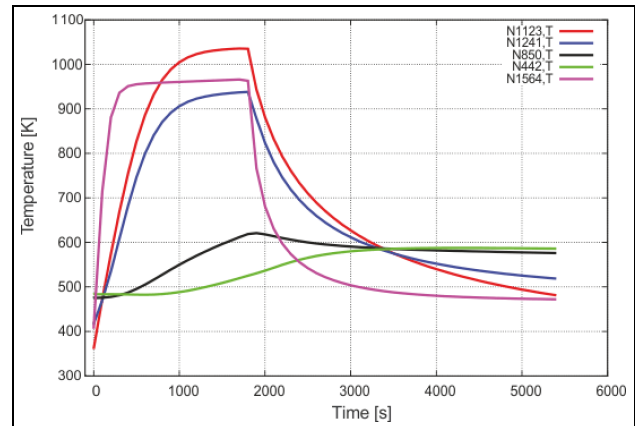


Figure 121: Transient temperatures at nodes over time

the following derived results are available:

- the gradient of the temperature field,
- the heat flux through any internal face,
- arbitrarily composed element sets allow for the output of the heat flux through a part of the surface in absolute or area specific values.

In addition, for transient analyses primary and derived results may be issued for any point in order to generate XY plots.

## PERMAS-NLHT – Nonlinear Heat Transfer

The methods available for nonlinear analyses in PERMAS-HT are complemented by a more advanced set of algorithms to solve higher nonlinear computations for steady-state and transient problems.

In contrary to the methods described in the previous section an automated stepping algorithm is used for both steady-state and transient analysis, which may be complemented by manual selections of explicit points in time or load levels.

For the selection of results, load steps and particular times can be explicitly defined.

### Radiation with heat exchange

Heat transfer by radiation is increasingly important with higher temperatures and for parts with cavities and self-shadowing effects like brakes, combustion

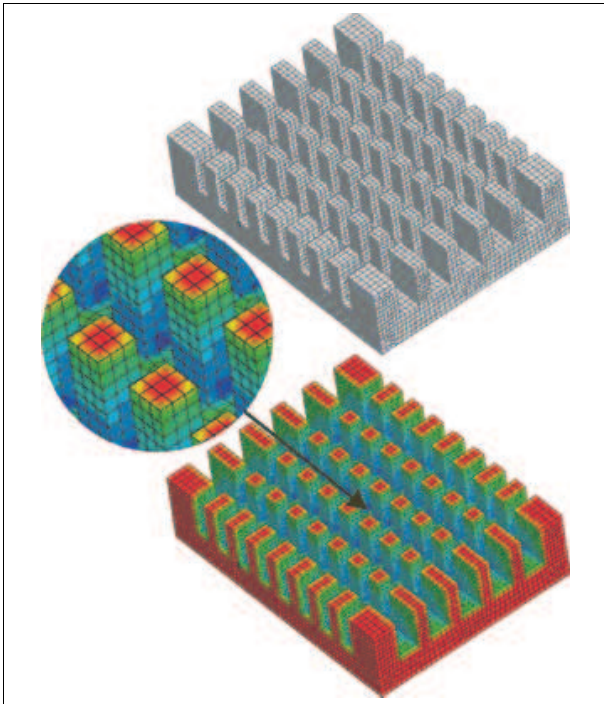


Figure 122: Analysis of cooling element with radiation

engines, and cooling elements. This allows heat transfer analyses with convection and radiation coupled with heat conduction.

The assumptions for this function are heat exchange between surfaces (no radiation from within bodies), radiation of grey bodies (radiation not dependent on wave length), and diffuse emission (radiation not dependent on radiating direction).

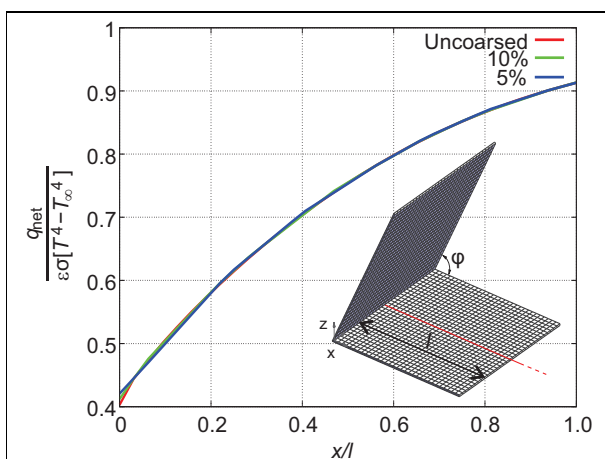


Figure 123: Heat radiation between adjoint plates

The calculation of radiation has the following characteristics:

- The radiation is integrated in the heat transfer

analysis process.

- The convection elements are extended to model also radiating surfaces, i.e. all surface elements where radiation has to be taken into account have to be modeled with convection elements.
- There is a direct integration of the *view factors* over the surface elements instead of averaged view factors.
- In order to accelerate the calculation of viewing factors with a very high number of surface elements an automatic (selective) coarsening procedure is provided to reduce the number of surface elements.
- The computational efficiency is obtained by using *parallelization*.
- The coupled solution of the nonlinear heat transfer equation with radiation boundary conditions is performed in a few iteration steps either for steady-state or transient calculations.

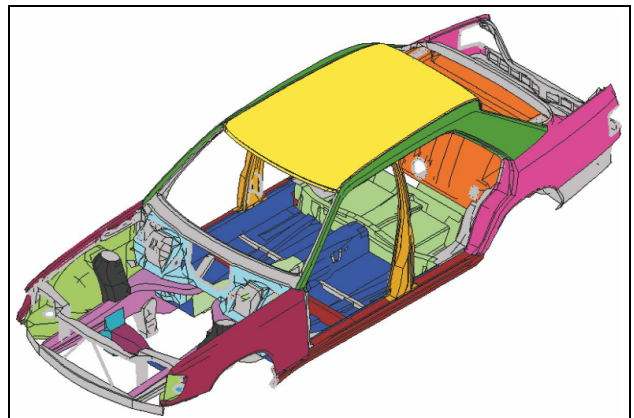


Figure 124: Frequency response optimization of a body-in-white with shape and sizing parameters (see also Fig. 127)

## PERMAS-OPT – Design Optimization

Beside the pure FE modeling, PERMAS also allows the definition of a design model and its automatic optimization.

The following design variables are provided:

- **Sizing:**
  - areas of cross section, inertia moments and general functions between these properties for beam elements,
  - all parameters of standard beam cross sections (see page 56),

- thicknesses/offsets/nonstructural mass of membrane and shell elements,
- stiffness and mass of spring elements,
- mass of mass elements,
- damping parameter of damping elements,
- parameters of control elements,
- convection film coefficients,
- material parameters.
- **Shape optimization:**
  - node coordinates for shape optimization,
  - use of design elements (see page 56),
  - use of incompatible meshes (for *positioning*) without remeshing (see Fig. 77),
  - *bead design* (see Fig. 126).
  - Nonparametric *free-form optimization* (see below on page 85).
- **Design variable linking**

In each optimization constraints shall limit the value range for design variables as well as for the response quantities like:

- displacements, velocities, accelerations,
- element forces,
- reaction forces,
- stresses,
- compliance,
- weight,
- contact gap widths,
- contact pressure,
- contact forces,
- eigenfrequencies,
- sound radiation power density,
- temperatures,
- heat fluxes,
- general constraints as combination or arbitrary function of the above mentioned quantities. Such functions include global criteria like max/min, absmax/absmin, or RMS.
- element quality, where the PERMAS element test is mapped to a continuous variable with values between 0. (i.e. perfect element) and 1. (i.e. erroneous element). This design constraint will help to avoid the failure during optimization due to collapsing elements.
- *Design constraint linking* is available as powerful tool to build constraint equations for multiple result values.

The objective function of an optimization may be the weight or any other specified constraint.

Dependent nodes are also allowed for shape modifications. This allows the use of incompatible meshes to realize larger modifications without the need to remesh a structure (see example in Fig. 125).

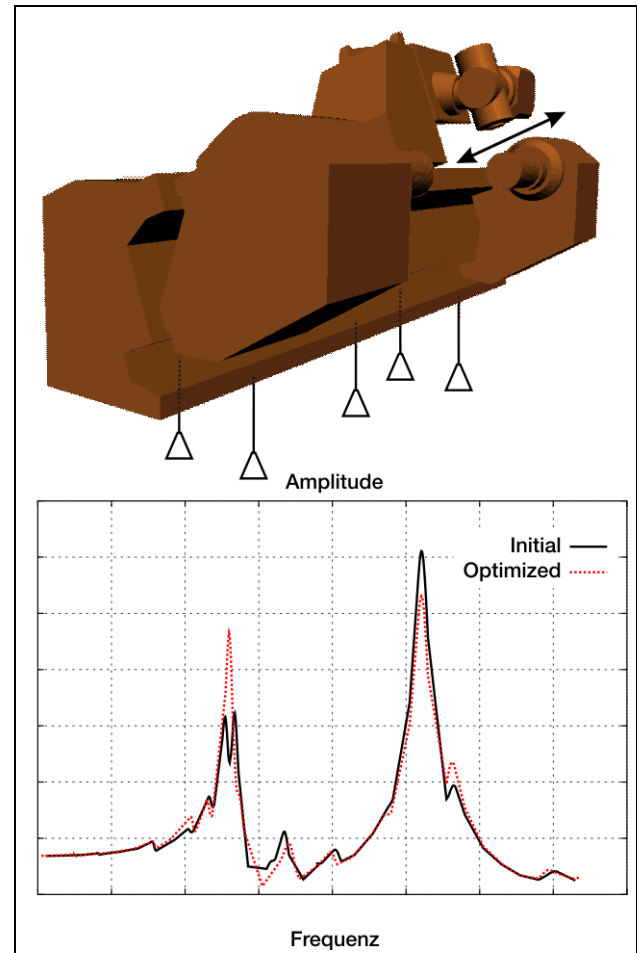


Figure 125: Frequency response optimization for a machine foundation where the position of the base spring elements is optimized which are coupled to the machine structure by interpolating surfaces. Design objective is the maximum flexibility of the tool center point.

The following solvers are available for optimization:

- Linear statics,
- *Inertia relief* (see page 66),
- Eigenvalue analysis,
- Modal frequency response analysis,
- Steady-state heat transfer analysis.

With the aid of module AOS (see page 88) additional solvers are available for Optimization:

- Contact analysis,
- Nonlinear material behavior.
- Complex eigenvalue analysis (including rotat-



ing structures)

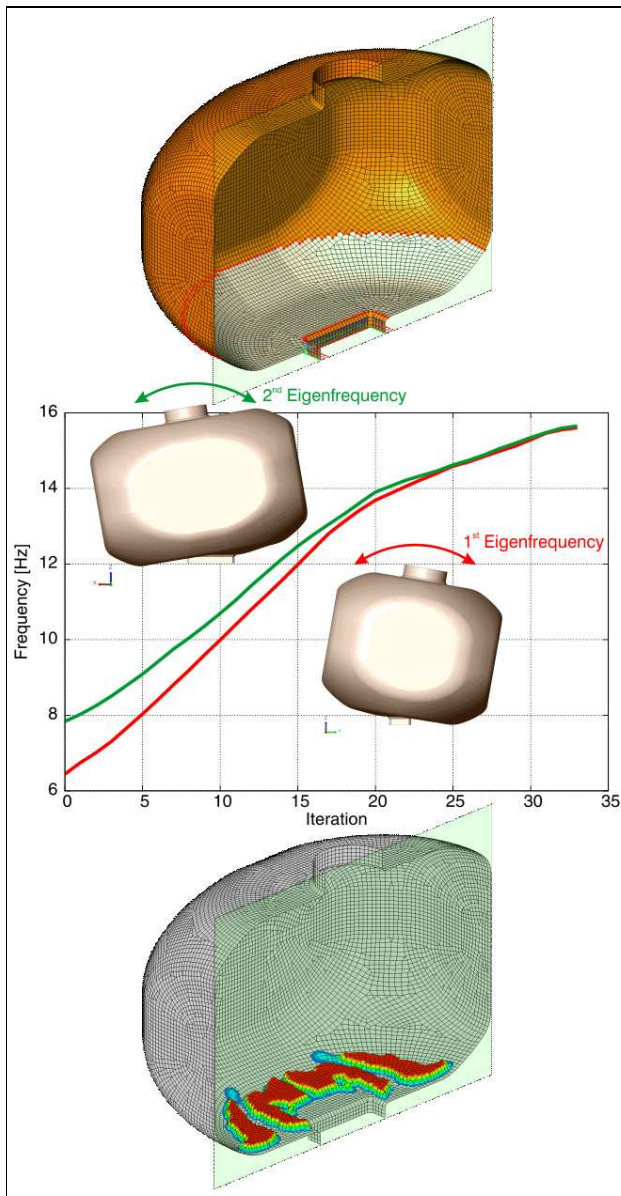


Figure 126: Bead generation for plastic tank

For frequency response optimization amplitudes, phases, real, and imaginary values of the above listed results are available for constraint or objective definition. The limits for the constraints can be made dependent on frequency.

Different solvers can be combined in one optimization task as well as sizing and shape parameters.

The optimization allows taking into account several loading cases as well as different boundary conditions using variant analysis (see page 50.) In addition, dynamic mode frequencies can also be optimized, where a mode tracking during the structural

changes is performed automatically.

If a small part of a structure is optimized, **substructuring** can be used to reduce run time by separating the design space in the top component. So, the reduction of the unmodified parts has to be done only once.

The results of an optimization are the history of the objective function and an overview on the validity of the design after each iteration. In addition, the values of the design variables and the constraints are available as a function of the iterations performed. These functions may easily be viewed as XY plots. The export of sensitivities is also possible.

Moreover, element properties may be prepared for result processing (i.e. thickness distribution) and exported for post-processing.

The results of a shape optimization can be exported as displacements for post-processing with the original model or as new model with identical topology and modified coordinates.

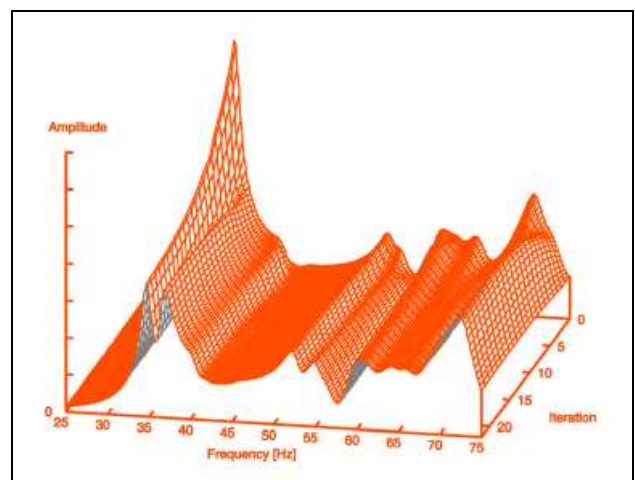


Figure 127: Iteration history of a frequency response optimization

A **non-parametric optimization** feature has been integrated which allows a *free-form shape optimization* of structures for minimizing stresses by homogenization or for limiting stresses at minimum weight (see Fig. 14). This opens the most easy way to define shape optimization of free-form geometries. The set-up of this optimization is supported by a freeshape wizard in VisPER (see page 44).

**Optimization for a robust design** is achieved by additional reliability constraints. Then, the design fulfills all of the above mentioned constraints and it is

also reliable regarding uncertain model parameters (see page 36 for more details).

## PERMAS-TOPO – Layout Optimization

Topology optimization is a method to find in a given limited part of space a finite element structure being optimal relative to a user defined criterion and fulfilling a set of given conditions. For this task a part of the model, the design space, has to be filled with finite elements. Each finite element gets its own design variable, the filling ratio (with values between 0 and 1). It is used for the calculation of a scale factor for the elemental stiffness. If the filling ratio is near zero, so is the stiffness. Then the resp. finite element does not contribute to the mechanical behaviour of the structure and can be neglected. The elements with high filling ratios are the necessary ones to fulfil the given criteria. It is clear that the layout found by this process consists of a subset of the elements of the design space. The finer you discretize the design space the more detailed the layout result can be, but the higher is the computation effort.

This module supports the concept development stage by providing fully integrated topology optimization strategies:

- Design space specification with variable/fixed parts,
- Provide boundary conditions,
- Provide loads,
- Target definition with remaining volume,
- Additional constraints for the optimization, if any.

For the modeling continua elements like membranes, shells, and solids are used and *substructuring* is supported. Additional modeling parameters are:

- **Fixed/free design elements:**
  - filling ratio per design element,
  - design variable limits,
  - design variable modification limits.
- **Manufacturing constraints:**
  - *Release Directions:*  
For casting special constraints have to be applied in order to get producible parts out of a topology optimization. So, release directions can be specified to control this kind of

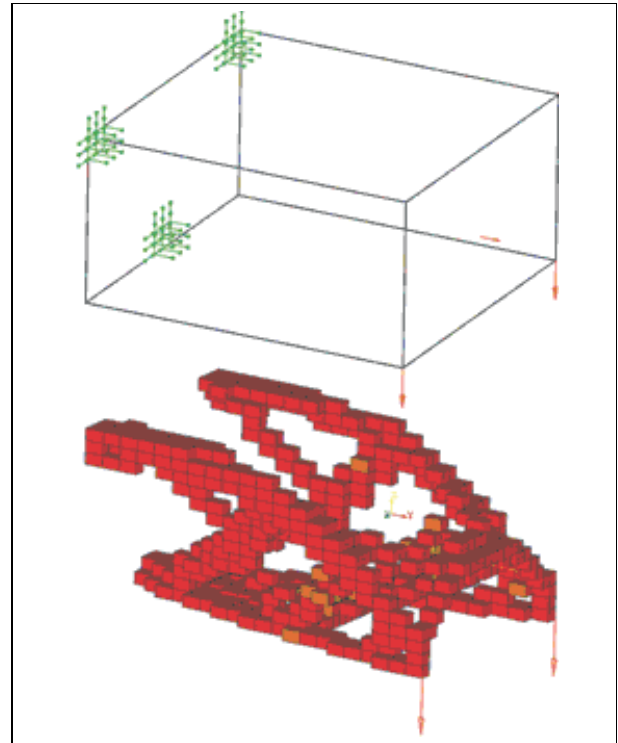


Figure 128: Layout optimization  
Design space with boundary conditions and loading (above) and the optimal material distribution (below)

manufacturing constraint (see Fig. 129). In addition, a minimum residual wall thickness can be specified to obtain a closed structure (like housings). A *fixed mold parting line* can be specified in case of opposing release directions.

- Symmetry conditions:  
Planar, axial, and cyclic symmetry conditions can be specified to determine the final properties of the layout result.
- Repetitive structures: Design element linking is provided to get the same layout for different parts of a structure.
- *Maximum member size:* In combination with release directions a maximum member size option is available to restrict the thickness of remaining structures.
- *Minimum member size:* Minimum member sizes in the remaining structure (i.e. widths and thicknesses) can be controlled by corresponding parameters (so-called checkerboard filter).
- *Frozen regions:* Such regions of the design space may not be changed during the optimization process.

- **Design constraints and design objective:**

- compliance,
- weight,
- reaction forces,
- eigenfrequency (mode range),
- displacements,
- accelerations, velocities,
- stress (in the non-design space),
- element forces (in the non-design space),
- sound radiation power density (in the non-design space).

Each design constraint can also be used as objective function.

Design constraint functions can be used to build more complex constraints out of the above listed basic constraints.

A general objective function facility can be used to set up an objective function dependent on multiple constraint values (like max/min, abs-max/absmin, RMS).

- **Multi-Modeling**

- several load cases simultaneously with different superposition options,
- different design variants.

With solution options in module AOS (see page 88) maximum member size in topology optimization is now exactly formulated.

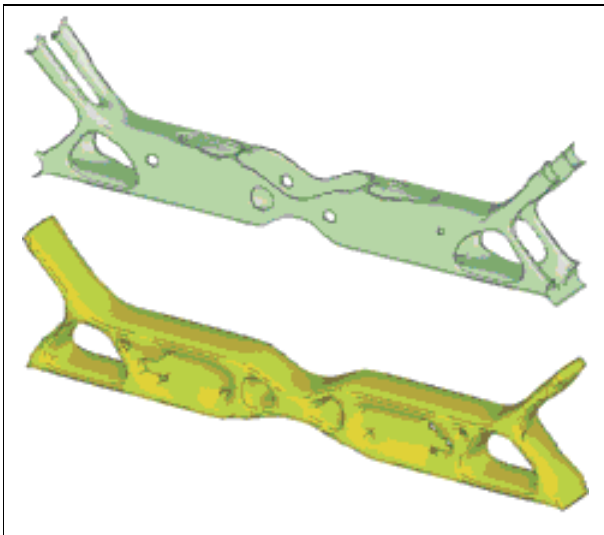


Figure 129: Different results for topology without (above) and with (below) release directions

A number of analysis options are available for the optimization like

- Linear statics,

- Contact analysis,
- Dynamic mode analysis,
- Modal frequency response analysis.

Because eigenfrequencies and mode shapes change a lot during a topology optimization, strategies for the suppression of local modes are available and of particular importance.

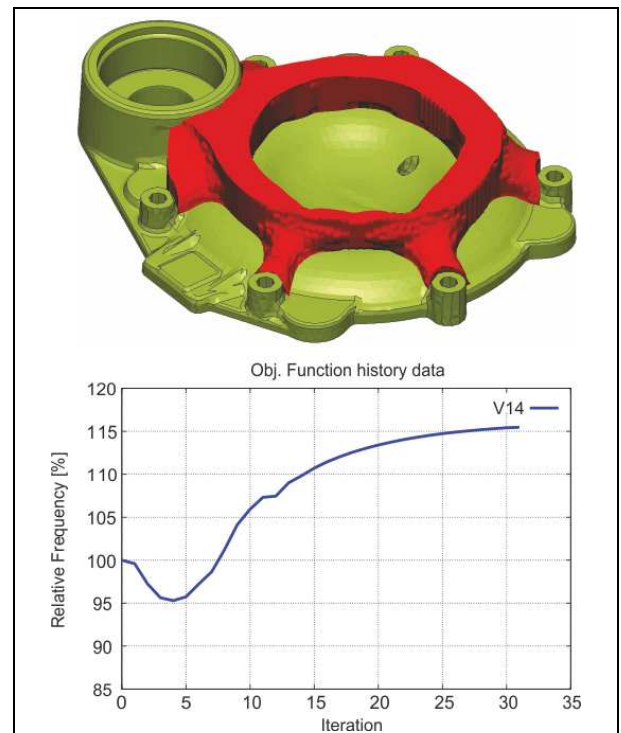


Figure 130: Topology optimization of a gearbox cover

to maximize the second eigenfrequency with release direction (by courtesy of ZF Friedrichshafen AG)

The optimization itself is performed using one of the following algorithms:

- GCA (Global Convex Approximation) for eigenfrequencies and a combination of static and dynamic constraints.
- PD (Primal-Dual Solution)

The iterations of an optimization can be controlled either by the convergence of the objective function or by a maximum number of iterations.

Topology optimization is able to provide a solution with **clear separation** of filled and void elements. So, elements will show filling ratios near 0 or near 1. This feature avoids misinterpretation of topology optimization results and facilitates the use of the result directly for further analysis and design steps (see

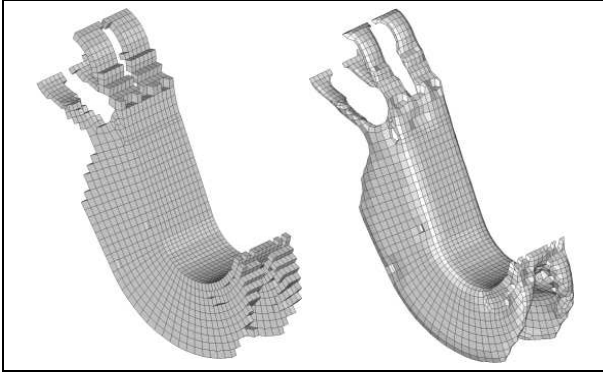


Figure 131: Layout-Optimization of a crane hook  
(on the left side the primary result,  
on the right side the smoothed surface)

Fig. 15).

Beside the history of the objective function, the result of a layout optimization is the element filling ratio. On the basis of the filling ratio, the remaining structure can be visualized in the post-processor easily.

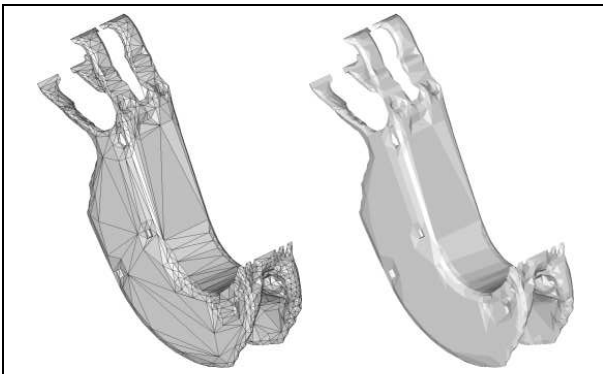


Figure 132: Polygon reduction  
(on the left side the triangle mesh,  
on the right side graphical representation of the surface)

The remaining structure can be further processed by one of the following means:

- **Hull generation:**  
For an automatically determined or prescribed value of the filling ratio the corresponding surface in space is extracted as mesh out of quadrangles and triangles.
- **Smoothing:**  
The hull is smoothed taking into consideration the boundary of parts, loads, and kinematic constraints.
- **Polygon Reduction:**  
The mesh is purged of too small triangles and

quadrangles in order to achieve a surface description as compact as possible.

- **Export:**  
The remaining triangle mesh can be exported as FE mesh for post-processing or as geometry using *STL* format.

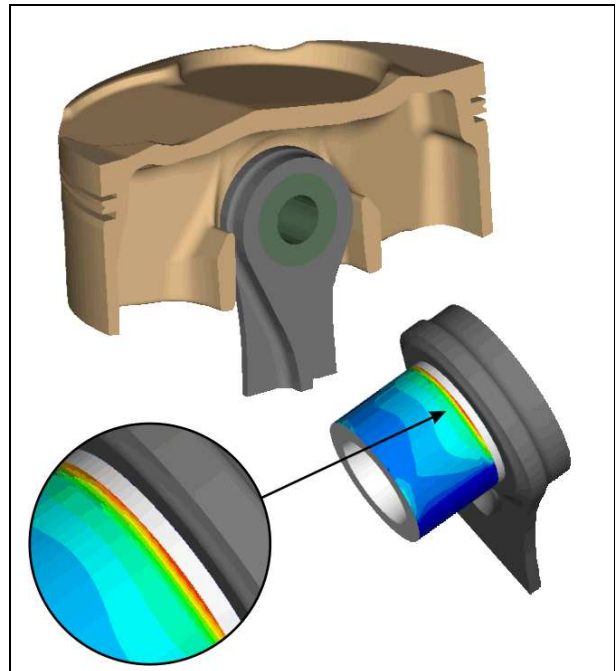


Figure 133: Optimization of contact pressure  
between piston and piston pin (Mahle GmbH, Stuttgart).  
The figure shows the edge pressure between both parts  
(for the optimized result see also next Fig. 134).

## PERMAS-AOS – Advanced Optim. Solvers

This module provides additional optimization solvers which essentially extend the range of applications for the integrated optimization in PERMAS. The extensions are as follows:

- By *Trust Region* method based local methods adaptive stepsize control is facilitated. This extends the previous static modification limit chosen by the user.
- Trust region methods keep track of the best point. They reject points, where no improvement is achieved. This extends the previous methods, where any new point is accepted.
- Methods for derivative-free optimization and global optimization are available.



Using these methods, new application fields for optimization are opened, like contact analysis and non-linear material analysis.

The local methods include the following derivative-based methods:

- **SQP** (Sequential Quadratic Programming):  
This is a damped *Newton* method combined with an *active set strategy* for the optimality equations. It is the best general purpose method (but not necessarily in structural mechanics). Second order information is available by *BFGS* update.
- **SLP** (Sequential Linear Programming):  
This method uses only linear approximation. Usually, it is slower than SQP due to missing 2nd order information. It is sometimes more robust than other gradient based methods (e.g. in the case of steep gradients).
- **SCP** (Sequential Convex Programming):  
Usually, best-of-class method for classical optimization problems arising in structural mechanics. Module OPT uses a method which belongs to SCP class of optimization methods.

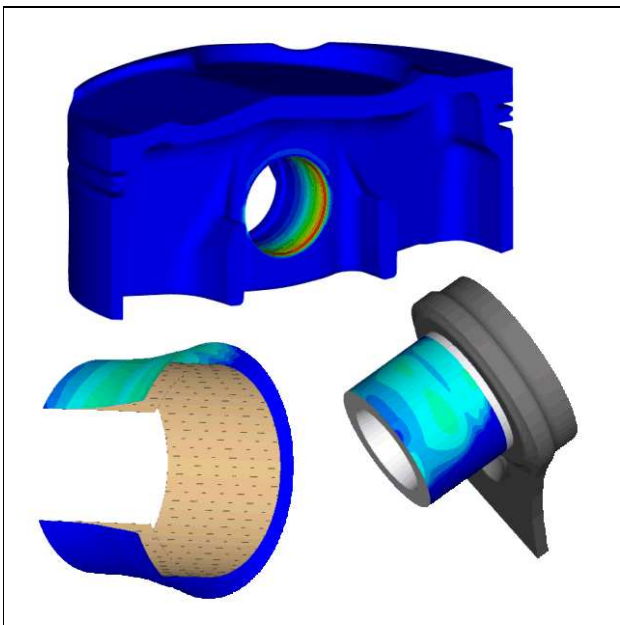


Figure 134: Optimization of contact pressure between piston and piston pin (Mahle GmbH, Stuttgart).  
The figure shows the optimized geometry of the piston and the reduced edge pressure.

When derivatives are not available, e.g. in contact problems or nonlinear material behavior, or when the accuracy of computed derivatives is not sufficiently high (like sometimes in frequency response

analysis), then derivative-free methods can be applied. The new derivative-free (local) methods comprise the following approaches:

- Derivative-based methods using finite differences (with SQP, SLP, SCP). Functions should be smooth enough and the choice of the finite difference parameter for the interval should not be a problem.
- Derivative-free method **WLIN** (Wedge constraint, LINear approximation). There is no need to choose a finite difference parameter. This method can be used for noisy problems.

When global minima have to be found, local methods are not appropriate any more. For such global optimization tasks, the following approaches are available:

- By applying the *Multi-Start method* (**MS**) and using random points derivative-based methods can be used to localize minima. This is combined with keeping track of the best point. This approach can be seen as an automatic trial method. A maximum number of loops is used to terminate the analysis.
- Another method is **LDR** (Locally improved variant of the *Dividing Rectangles* (*DiRect*) algorithm). Fig. 135 shows an example for this approach. This method has been generalized to work with constrained problems. It could be improved by solving local subproblems. It generates a sequence of points that is dense in the design space and hence guarantees to approximate the global solution. Because this method is slow and only useful for small models, a suitable model reduction is highly recommended (see substructuring on page 49).

## PERMAS-RA – Reliability Analysis

In the classical approach to structural analysis a deterministic model is used to predict the behavior of the design under various loading conditions. The results of such calculations are compared to typical limiting constraints such as a maximum stress or deflection under the consideration of safety margins. This is called deterministic approach to the problem of structural safety and the Finite Element Method has become a widespread tool in such procedures.

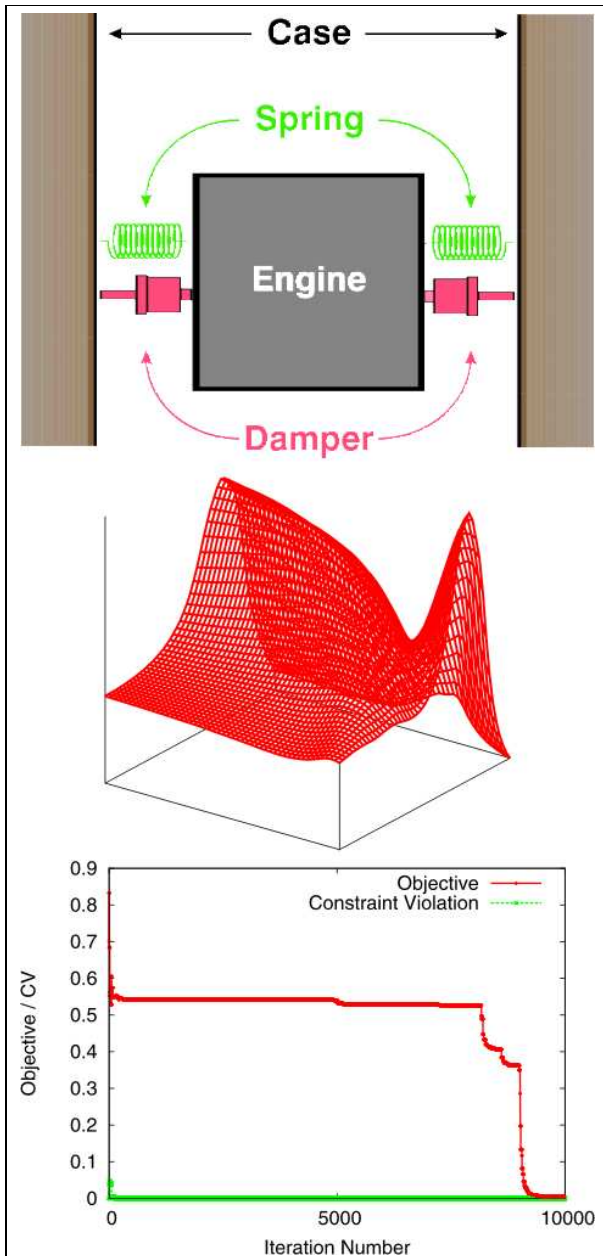


Figure 135: Global optimization

of a spring-damper system with the LDR method. Top: model with 4 design variables, Middle: 2 design variables fixed at the optimum, objective function dependent on the 2 free variables, Bottom: course of objective function and constraint violation during the iterations.

In contrast to this method, the *stochastic analysis* of a design assumes some properties of a structure or the loads to be uncertain knowing only the characteristics of their probability distributions. The limiting constraints on the design will usually be of the same kind as in the deterministic approach. However, the results from the probabilistic analysis will yield the probability of failure with respect to these constraints

and the sensitivity of this probability with respect to the uncertain properties of the model.

This module combines the Finite Element Analysis with the well established COMREL program developed by RCP GmbH, Munich. So, the experience comprised in both software systems could be merged in a single application simplifying the approach to the Stochastic Finite Element Method.

The procedure in reliability analysis comprises the following three steps:

- Definition of uncertain quantities in structural analysis (like geometrical or load parameters) by *basic variables* with an assigned distribution function.
- Definition of *limit state functions* (or *failure functions*) related to result quantities of a structural analysis.
- Calculation of the *probability of failure* for each limit state function.

The following quantities can be used as **basic variables**:

- Design parameters (like geometrical data or coordinates)
- Load factors
- Material parameters
- Parameters of the limit state functions
- Parameters of other basic variables

More than 20 different types of distribution functions are available to describe the basic uncertain variables.

The stochastic analysis performs an assessment of the failure parameters for the following analysis types:

- Linear static analysis
- Contact analysis
- Dynamic eigenvalue analysis
- Frequency response analysis
- Complex eigenvalue analysis (including rotating structures)

For this purpose, a number of methods are available:

- Efficient sensitivity based methods as First/Second Order Reliability Methods (*FORM/SORM*)
- *Response surface methods*
- *Monte Carlo* simulation using adaptive sampling

- Crude Monte Carlo simulation

The reliability analysis allows to take into account several loading cases as well as different boundary conditions using different failure functions.

- The definition of **Failure functions** is made using
  - General functions
  - Dependent on
    - \* results (displacements, stresses, etc.)
    - \* basic variables
    - \* constant values
- The primary **Results** of such an analysis are
  - Probability of failure for each limit state function
  - Parameter sensitivities of the limit state functions
  - Result sensitivities for basic variables (elasticities)
  - Selected data of each iteration for Monte Carlo simulations

---

## PERMAS-EMS – Electro- and Magneto-Statics

This module allows for steady-state electromagnetic analysis. Magnetic analyses may be based on a previously performed calculation of the steady-state current distribution. The analysis uses a scalar potential for the electric field and a vector potential for the magnetic field.

Various load types are supported.

The determination and handling of singularities is analogous to a static analysis (see page 55).

Heat induced by an electrical field can be used for a subsequent thermal analysis (see page 81). From that thermal stresses can be derived performing a subsequent static analysis (see page 65).

Forces induced by a magnetic field can be used in a subsequent static analysis (see page 65).

---

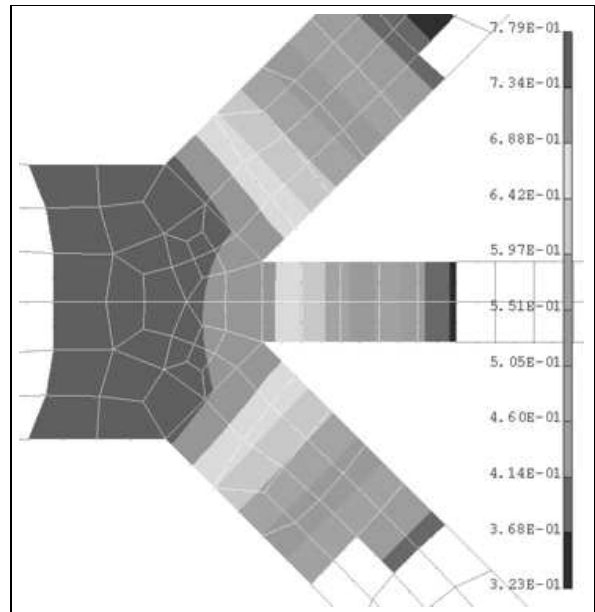


Figure 136: Scalar potential in an electric junction

## PERMAS-EMD – Electrodynamics

A solution of *Maxwell's* equations is available for different problem cases:

- eddy currents
- induction problems
- resonant cavities
- wave propagation
- general electrodynamics

All kinds of loading may be applied in dynamics (like prescribed potential). The specification is made by static loading cases and appropriate time functions like in structural dynamics (see page 76).

---





## PERMAS Special Modules

### PERMAS-LA – Laminate Analysis

The laminate analysis serves for the modeling and analysis of multi-layered *fibre-reinforced composites*. Therefore, PERMAS provides for 3-node and 4-node shell elements.

Out of the pre-processing with MEDINA or I-DEAS the geometry, the boundary conditions and loads as well as the material set-up of the laminate is used. From the number and sequence of layers and their thickness, fibre orientation, and material properties the usual ABD matrices are determined.

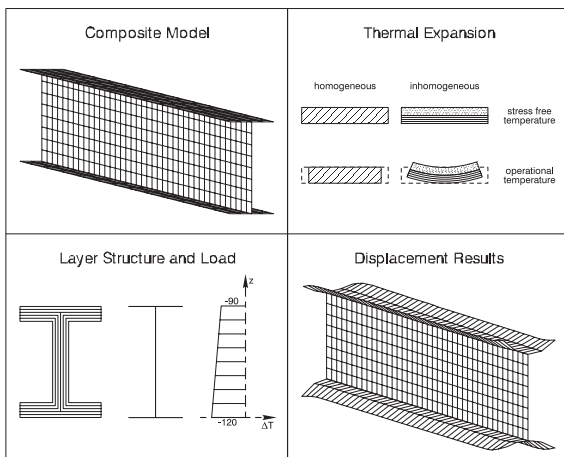


Figure 137: Laminate analysis of a girder

The analysis results in element forces, from which the layer stresses and strains are derived. Using the MEDINA post-processor, failure criteria, failure indices, and safety factors may be evaluated. As usual, all other post-processing features may be used for composite structures in addition.

### PERMAS-WLDS – Refined Weldspot Model

The modeling of weldspot connections is described on page 53. This modeling gives a good representation of the global stiffness.

But along a weld line the weldspot forces can vary very much due to e.g. discretization effects between the incompatibly meshed flanges.

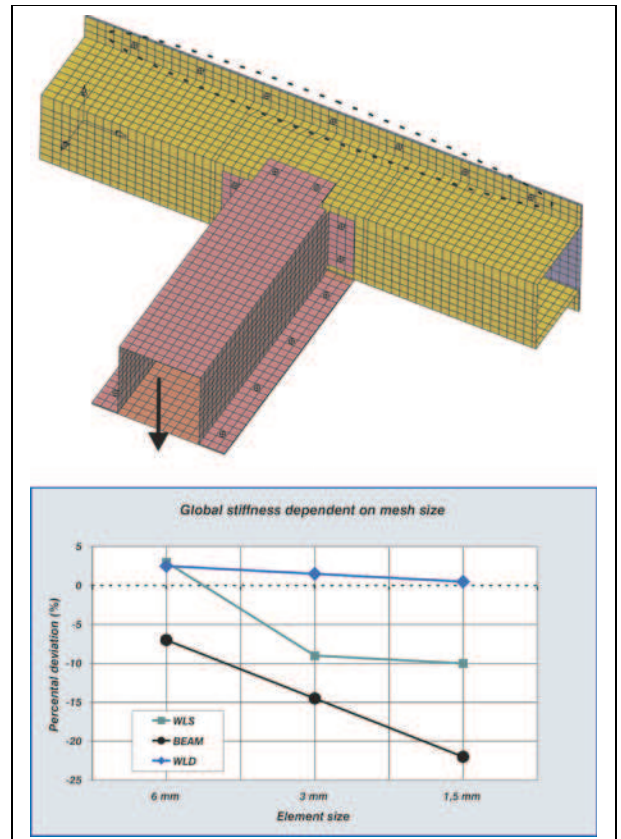


Figure 138: Quality of the refined weldspot model

This module offers a refined weldspot model which is characterized by very low variations of the weldspot forces and by an improved stiffness representation. Among others, this is achieved by an internal calibration of the stiffness using a volume model.

### PERMAS-XPU – GPU accelerator

The module XPU supports Nvidia GPU (Tesla K20c or better with CUDA Version 6) for acceleration of compute-bound analyses by a seamless integration in PERMAS parallelization concept. The functionality is offered together with the S64 versions of PERMAS (see the list of PERMAS operating modes on page 101), since an efficient use of the GPU requires a sufficient core memory of the host machine ( $\geq 32$  GB). The acceleration will be most beneficial for compute-intensive analyses, such as eigenvalue analysis with a high number of modes, a direct fluid-structure coupled response analysis, or large non-

mal contact problems.

---

## Interfaces

### PERMAS-MEDI – MEDINA Door

This interface directly reads the model from the MEDINA data bus (.bif) and writes the results back to the data bus (.bof).

All MEDINA elements and almost all PERMAS *MPCs* are translated (see page 53). Beside Components, different Situations with constraint and load variants may be specified within MEDINA (see page 50).

It is a very special feature in MEDINA that PERMAS element tests have been integrated for element validation during pre-processing. So, if a model is checked in MEDINA, it will likely pass the PERMAS tests.

The part handling by incompatible line/surface coupling is supported.

The interface supports the following analysis types:

- linear and nonlinear statics
- contact analysis
- dynamic mode analysis
- thermal analysis
- fluid-structure acoustics (basic modeling and post-processing)
- electromagnetics (basic modeling and post-processing)

The interface is continuously adapted to new versions of MEDINA and extended to cover new features of PERMAS.

- all kinds of loads incl. inertia loads
- all PERMAS kinematic boundary conditions
- sets
- specifications of substructures
- variant definitions

The interface supports the following analysis types:

- linear and nonlinear statics
- contact analysis
- dynamic mode analysis
- thermal analysis
- fluid-structure acoustics

Even for other types of degrees of freedom like electric potential the model may be prepared within PATRAN.

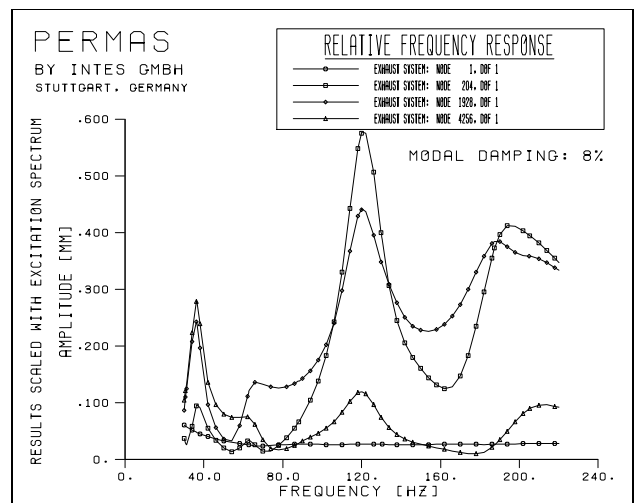


Figure 139: Exhaust System  
Frequency response analysis

### PERMAS-PAT – PATRAN Door

This interface reads the PATRAN database with the model and writes the results onto PATRAN result files.

The modeling is made using custom-made PERMAS preferences. Own solver menus support standard PERMAS solutions from within PATRAN, which may be adapted by the user himself. The following model parts are translated:

- all common elements

### PERMAS-ID – I-DEAS Door

This interface reads the model from an I-DEAS Universal File:

- all common element types
- axisymmetric models
- all kind of loading incl. inertia loads
- many linear kinematic constraints like 'rigid element' and 'coupled dofs'
- laminate material
- sets
- specifications of substructures
- variant definitions

The interface supports the following analysis types:

- linear and nonlinear statics
- contact analysis
- dynamic mode analysis
- thermal analysis

Even for acoustic and electromagnetic analyses the models can be prepared in I-DEAS.

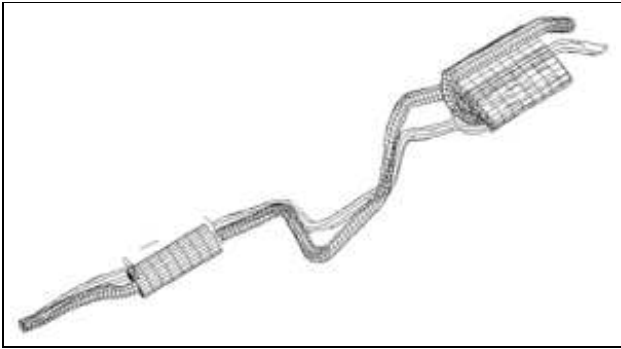


Figure 140: Mode Shape of an Exhaust System

---

## PERMAS-AD – ADAMS Interface

A post-processing interface for the export of model topology and mass-normalized vibration mode shapes. In addition, generalized stiffnesses and masses are exported and, if needed, also static mode shapes.

The joint orthogonalization of static and dynamic modes can also be very efficiently performed within PERMAS (see module DEV, page 73).

In addition, the export of statically or dynamically condensed models to ADAMS is possible. There, the automatic *substructuring* with MLDR may be used, too (see page 76).

The export is made in the format of ADAMS Version 2007 r1 and up.

---

## PERMAS-DADS – DADS Interface

A post-processing interface for the export of model topology and mass-normalized vibration mode shapes. In addition, generalized stiffnesses and masses are exported and, if needed, also static mode shapes.

- Based on single component model.
  - Static mode shapes defined by prescribed degrees of freedom.
  - Static analysis and dynamic eigenvalue analysis.
  - Output based on PERMAS post file format.
- 

## PERMAS-EXCI – EXCITE Interface

Post-processing interface for export of model topology, mass-normalized vibration mode shapes and condensed matrices. This interface was developed for EXCITE Version 7.0

A filtered backtransformation matrix can be exported for further use after EXCITE execution.

The export can be made in zipped format (by *gzip*) in order to reduce the size of exchanged data.

---

## PERMAS-SIM – SIMPACK Interface

Post-processing interface for export of model topology, mass-normalized vibration mode shapes and condensed matrices.

- Based on substructure model.
  - *Guyan's* reduction (*static condensation*) by PERMAS.
  - Optional also with *dynamic condensation*.
  - Or - very interesting - with automatic substructuring using MLDR (see page 76).
  - Output of stiffness/mass/etc. on top component level.
  - Also export of geometrical stiffness matrices.
  - For visualization in SIMPACK the complete model of the uncondensed structure can be exported.
-



## PERMAS-HMS – MotionSolve Interface

Post-processing interface for export of model topology, mass-normalized vibration mode shapes and condensed matrices. This interface was developed for HyperWorks Version 11.

---

## PERMAS-H3D – HYPERVIEW Interface

Post-processing interface for export of model topology and results to HYPERVIEW (from Version 8 onwards). There are formats for HYPERVIEW Version 8 and 11 available.

Supported results are from statics, contact analysis, nonlinear statics, dynamics, acoustics, and heat transfer. The format does not support xy data.

---

## PERMAS-VAO – VAO Interface

Post-processing interface for export of model topology, mass-normalized vibration mode shapes and damping matrices for displacement and pressure degrees of freedom to VAO.

---

## PERMAS-VLAB – Virtual.Lab Interface

Post-processing interface for export of model topology and results to Virtual.Lab:

- Displacements, velocities, accelerations,
- Rigid body mode shapes,
- Assembled loads,
- Reaction forces,
- Stresses,
- Strain energy,
- Kinetic energy,
- Sound radiation power.

The joint orthogonalization of dynamic eigenvectors and static mode shapes can be performed very efficiently in PERMAS (see module DEV, page 73).

---

## PERMAS-MAT – MATLAB Interface

A post-processing interface for the export of model topology and matrices.

Often MATLAB is used for the design of controllers. In case of linear controllers the relevant parameters can directly be used in a PERMAS model when control elements are applied.

Moreover, all controllers (in particular nonlinear controllers) can be used in a dynamic transient analysis by providing an appropriate PERMAS user function, which is linked to the software as C or FORTRAN subroutine.

---

## PERMAS-NAS – NASTRAN Door

The NASTRAN Door reads and checks model files compatible to NASTRAN and translates these input files directly into internal PERMAS data structures.

The **main capabilities** of the NASTRAN-Door are:

- All Bulk Data formats are supported: Small Field, Large Field and Free Field (with all possibilities for card generation).
- Executive and Case Control sections are translated as well as all global usable statements.
- PERMAS Components and Situations are built according to the Bulk Data model and the selections made by physical Case Control requests.
- Take-over of NASTRAN identifiers, e.g. element-, node- and set-IDs survive the interfacing process and will be taken as PERMAS identifiers.
- Automatic label generation with labels compatible to those generated by NASTRAN.
- Fast Bulk Data sorting with machine independent sort sequence (i.e. equivalent sorting on ASCII and EBCDIC computers).
- NASTRAN-like echo of input statements.
- Extensive error tests are performed:
  - All public NASTRAN statements are recognized and analyzed lexically.
  - All supported statements are completely checked for wrong, missing or contradictory arguments.
- Additional features:
  - Calculated results may be referenced under Subcase- or Load-ID.

- Free Field comments within Fix Field statements.
- The INCLUDE statement supports multiple file levels.

A remarkable specialty of the NASTRAN-Door is the ability to utilize the given **control data** as well.

- Control input such as solution-ID, output requests and method selections are converted into task control structures, equivalent to those made by explicit UCI input (User Control Interface).

This so-called **NASTRAN Task** may be started by one special UCI statement.

To execute a NASTRAN-compatible run, only 4 PERMAS UCI-commands are necessary (see figure). Using this simple UCI input, NASTRAN decks can be executed without the necessity of any additional input.

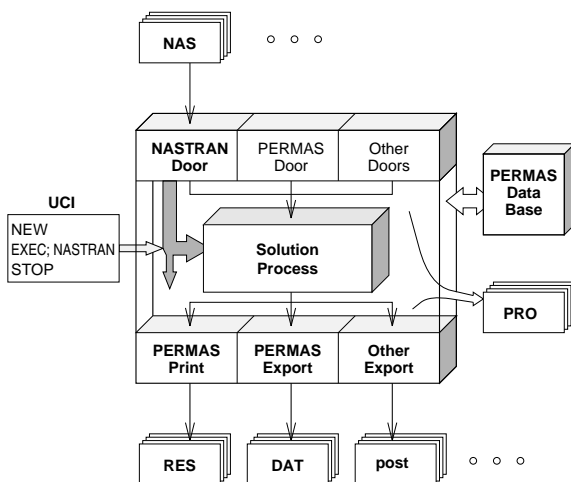


Figure 141: UCI trigger for NASTRAN-Task

Additional features of this NASTRAN Task are:

- Automatic re-translation of its internal task definitions into plain UCI commands echoed to the user.
- The user can choose between the automatic solution control (read from the given NASTRAN model file) and an explicit UCI control.
- The user can extend the NASTRAN Task by additional UCI statements.
- NASTRAN subcases may be selected and executed individually, leaving all remaining subcases untouched.
- Actually the following **solutions** are supported:
  - Linear Static,

- Normal Modes Analysis,
- Direct Frequency Response,
- Direct Transient Response,
- Modal Frequency Response and
- Modal Transient Response.

- Supported parameters, which affect the control flow, are also evaluated.

Finally the application of the NASTRAN Task is not restricted to pure NASTRAN-Door input. Even if model additions were made by other input Doors (e.g. DAT file input) – the user is free to execute the NASTRAN control data, automatically.

The export of limited results to certain OUTPUT2 formats (i.e. as generated for PATRAN and I-DEAS post-processing). The model data comprise elements, nodes, and local coordinate systems. Supported results are displacements, mode shapes, rigid body mode shapes, and reaction forces.

## PERMAS-ABA – ABAQUS-Door

The ABAQUS-Door reads and checks model files compatible to ABAQUS and translates these input files directly into internal PERMAS data structures.

This direct interface between two solvers has the advantage that physical meaning of the Abaqus commands can be interpreted during translation to take the adequate equivalent in PERMAS. The user can customize the translation process by his individual door configurations.

The door supports a large number of model and loading data and provides important information and thematic tables to assure the quality of the translation and analysis process, e.g. an overview on the step history (see Fig. 142).

Step History Overview					
Permas		Abaqus			
Situation [UCI command]	LPAT	Step	Type	Analysis	Supported Step Key
STEP 1 4 [STATIC]	1002	1	General	STATIC	CONTACT INTERFERENCE
	5001,5002	2	General	STATIC	BOUNDARY/LOAD
	5001,5002	3	General	STATIC	BOUNDARY/LOAD
	1	2			CLoad
	2				CLoad
	3	4	General	STATIC	TEMPERATURE

Figure 142: Overview on step history

The following analysis procedures are supported:

- linear and non-linear static analysis,

- Steady-state heat transfer analysis,
- Eigenvalue analysis,
- Modal frequency response analysis.

Following modeling features are supported:

- Mass, beam, truss, spring, damper, 2D, 3D, gasket and connector elements,
- Local systems and transformations,
- Various coupling conditions,
- Anisotropic and temperature-dependent material data,
- Gasket properties and material data,
- Many specifications for contact modeling,
- Heat convection and radiation,
- Static loads and boundary conditions, also centrifugal and inertia loads,
- Complete load history description of supported analysis procedures,
- Dynamic load definitions with real and imaginary parts.

For immediate visualization and inspection, VisPER can be used to read ABAQUS input file. In case of missing or not translated modeling features, VisPER may be also be used to complete the PERMAS model.

---

## PERMAS-CCL – MpCCI Coupling

The integration of the coupling library *MpCCI* allows the coupling of PERMAS with CFD software systems as described in MpCCI section on page 64.

---

## More Interfaces

Beside the above described interfaces a number of additional interfaces for PERMAS are available, which have been developed by other software companies to couple their software to PERMAS. The following list only contains those interfaces which became known to the editor before this Product Description was published. So, the list may be not complete or even not correct. In any case, these interfaces are not part of the PERMAS product and their developers have to be contacted to get more information about the contained functionality.

- **Animator3 (GNS, [www.gns-mbh.com](http://www.gns-mbh.com)):** Post-processing of PERMAS results is done via MEDINA format.
- **ANSA (BETA CAE Systems, [www.beta-cae.gr](http://www.beta-cae.gr)):** ANSA supports the PERMAS input format for the model and the MEDINA format (.bif, .bof) for the result evaluation with the  $\mu$ ETA post-processor.
- **EnSight (CEI, [www.ensight.com](http://www.ensight.com)):** This post-processor takes the results directly from PERMAS files.
- **Evaluator (GNS, [www.gns-mbh.com](http://www.gns-mbh.com)):** This report generator takes the results directly from PERMAS files.
- **FE-Fatigue (nCode, [www.ncode.com](http://www.ncode.com)):** The data transfer is possible using MEDINA formats.
- **FEGraph (von Mach Engineering, [www.vonmach.de](http://www.vonmach.de)):** This software works as a comprehensive post-processor to PERMAS and processes the PERMAS formats.
- **FEMFAT (MAGNA POWERTRAIN, [www.femfat.com](http://www.femfat.com)):** The connection between PERMAS and FEMFAT is possible on the basis of the MEDINA export in PERMAS, because FEMFAT also supports this format.
- **HyperMesh (Altair, [www.altair.com](http://www.altair.com)):** This interface supports the PERMAS input format. It complements the PERMAS module H3D (see page 97) which exports the results for post-processing with HyperView.
- **MAGMALink (MAGMA, [www.magmasoft.de](http://www.magmasoft.de)):** This software is a module for the transfer of casting simulation results out of MAGMASOFT to FE meshes for stress and durability analysis.
- **Material data base MARLIS (M-Base, [www.m-base.de](http://www.m-base.de)):** This material data base contains material data of steel sheets and is capable to issue the material properties as PERMAS material description.
- **SFE CONCEPT (SFE, [www.sfe-berlin.de](http://www.sfe-berlin.de)):** In the concept phase of body-in-white design even automatic optimization with PERMAS can be applied. SFE CONCEPT generates a PERMAS model and reads the relevant results after the analysis run.
- **SimLab (Altair, [www.altair.com](http://www.altair.com)):** This pre-processor provides an interface in PERMAS input format.
- **TOSCA (FE-Design, [www.fe-design.de](http://www.fe-design.de)):** TOSCA is capable to work with PERMAS input and output directly.

All mentioned names of products and companies belong to their holder. The use does not imply that such names are free for general use.

---



## Installation and beyond

### Supported Hardware Platforms

For PERMAS, the following platforms are supported:

Architecture	Operating system
PC x86(_64)	LINUX Debian 6/7, LINUX Ubuntu 12.04/14.04, LINUX RedHat EL5/EL6, LINUX SuSE 11/12
PC x86_64	Windows 7

For VisPER, the following platforms are supported:

Architecture	Operating system
PC x86(_64)	LINUX Debian 6/7, LINUX Ubuntu 12.04/14.04, LINUX RedHat EL5/EL6, LINUX SuSE 11/12
PC x86_64	Windows 7

The supported platforms and the related operating systems are always subject to change due to ongoing development activities and new computers on the market. Therefore, any specific case has to be inquired using the contact address on the last page of this document.

The functions described in this document are usually available on all platforms. Nevertheless, some exceptions are certainly possible (i.e. with *parallelization*, CFD coupling, interfaces to third-party products using program libraries (like H3D, VLAB), etc.). Therefore, it is recommended for any specific case to make an inquiry using the contact address on the last page of this document.

PERMAS fully supports *64 Bit architecture* of modern processors. So, the following execution modes are available:

- 64 bit operating system:
  - D64: Double precision floating point operations on 32 Bit machine words with a memory usage of about 7 to 8 GB.
  - S64: Single precision floating point operations on 64 Bit machine words with a practically unlimited memory usage (recommended minimum is 16 GB).

## Licensing

The following license types are available:

- **Nodelock license:** The execution of the software is provided for one single computing node. This license type is only available as special function of a floating license.
- **Floating license:** The execution of the software is provided for a computer network, where a fixed number of potential executions can be started on different computing nodes.

For both nodelock and floating licenses, the *license manager* is responsible for all bookkeeping of license information. The following server architectures are supported:

- **Single server:** The server is acting independent from other license servers in the network. Each server has its own license equipment.
- **Multi server:** Three servers are needed, and two servers have to be available at least. Each server has the total license equipment.

For the management of the license server a WEB interface is available.

## Maintenance and Porting

PERMAS is regularly maintained and improved. Within a continual improvement process the actual software version is the best one. On a daily basis numerous software tests are performed and their results are verified.

All incoming problem reports are administrated by a special management system (GNATS) and forwarded to the responsible engineer. Each resulting correction leads to a unique version number of the software. If a problem is already solved at INTES and there is no workaround for the user, the actual and improved version of the software will be delivered.

Every month a Technical Newsletter is issued, which reports on software corrections, their reasons, and possible workarounds. There, also frequently asked questions are listed with important problem solutions. The Technical Newsletter can be accessed in a reserved section of the INTES homepage (see next Section). The Technical Newsletter can be sub-

scribed by the users, who will receive it automatically via e-mail.

Usually in a 2 years period a larger development step leads to a new major version of the software, which is shipped to all customers having a rental or maintenance contract. Then, also a new version of the user manuals is delivered. All shipments are made within a short time span.

When new versions of the pertinent pre- and post-processor become available, an adapted version of PERMAS is available in most cases without delay. As soon as changes of the compatibility become known, the users will be notified.

---

## User Support

PERMAS users have access to the following information sources:

- documentation,
- training,
- support platform on INTES internet homepage,
- hotline services via phone and e-mail.

In particular, the support platform on INTES homepage contains up-to-date information on PERMAS and is an increasing source for useful details:

- The Technical Newsletter contains all available information on software problems, known workarounds, and corrections.
  - There is a PERMAS User and Support Forum for the exchange of information between INTES and PERMAS users as well as between all PERMAS users.
  - Downloads.
  - PERMAS Documentation Templates can be used to organize the workflow and standard PERMAS analysis procedures.
  - Administrative information regarding PERMAS licenses, i.e. on how to change the hardware platform for PERMAS.
- 

## MEDINA + PERMAS

Under a partnership agreement with T-Systems INTES offers a package of MEDINA and PERMAS

to its customers.

The package is available for all commonly supported platforms and with Windows and LINUX.

---

## Additional Tools

To support the application of PERMAS, INTES offers some additional tools within the PERMAS Tools which are only available under Linux and Windows operating systems. They comprise:

- the integration of external management tools (like Open Grid Engine) in PERMAS for optimum throughput of all available computers with a minimum of effort for administration. This facilitates job distribution and management, particularly in case of parallel applications with PERMAS.
- the fast and comfortable INTES-EMACS text editor incl. online help, documentation for different file formats, and email service. Among others, this tool comprises the following features:
  - Identical edit environment on almost any of the hardware platforms supported by PERMAS.
  - Utilization of function keys and numeric keypad instead of control- and escape-sequences.
  - Context-sensitive help functions and on-line access to all PERMAS documents.
- the hotline message system OTRS for the reporting of application problems by email.
- The PERMAS Tools comprise a number of tools:
  - **PERMAS control center:** This is a small graphical user interface to submit a PERMAS job and to set the most common job parameters,
  - **PERMASgraph:** A graphical user interface to plot XY data from PERMAS, MEDINA, and PATRAN formats. PERMASgraph supports the following file formats:
    - \* PERMAS native format (.post or .post.gz)
    - \* in MEDINA native format (.bif, .bof)
    - \* in Patran native format
    - \* in CSV format (as exported to PERMAS to spreadsheet format; .csv)
 The diagrams can be saved in different file formats, such as:
    - \* Portable Network Graphics (.png)

- \* Encapsulated PostScript (.eps)
- \* PostScript (.ps)
- \* Portable Documents Format (.pdf)
- \* Microsoft Excel File Format (.xls)

The diagrams can also be printed on a local printer.

Additional tools are available for data manipulation, such as:

- \* Addition/Subtraction of curves
- \* Fourier transformation

A further submenu provides special evaluation tools, such as:

- \* Analyzer for surface-to-surface coupling
- \* Analyzer of potential and kinetic energy of fluid-structure coupled eigenmodes
- \* Analyzer of modal participation factors
- \* Analyzer for strain and kinetic energy (see example in Fig. 111)

- **PERMASmonitor**: This can be used to view the progress of running PERMAS jobs.

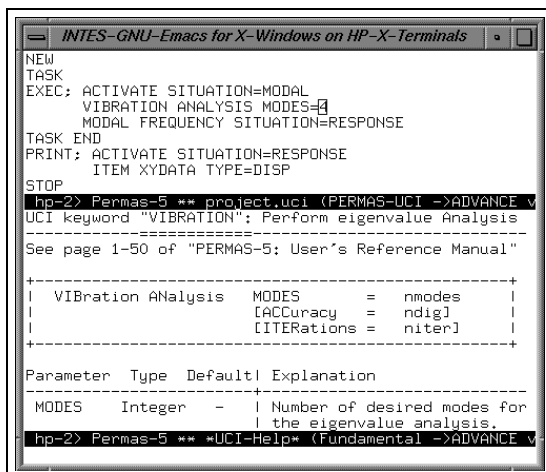


Figure 143: Context-Help for UCI-Files inside Editor

The PERMAS Tools are available for all PERMAS users. In addition, INTES offers a configuration service and an adaptation to the environment at the user's site as well as training.

## Documentation

In parallel to the development of PERMAS the documentation is currently updated:

- Apart from the basic documentation
  - *PERMAS Users Reference Manual*

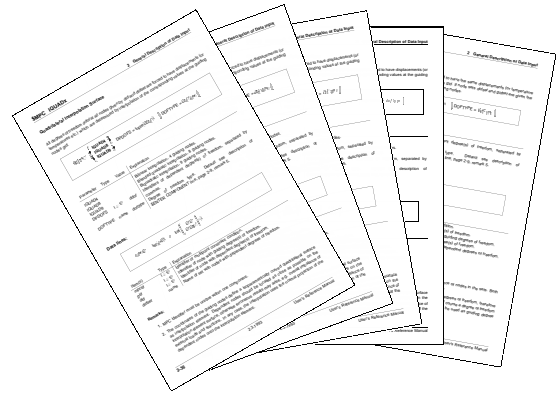


Figure 144: PERMAS documentation

- *PERMAS Examples Manual*
  - *VisPER Users Manual*
  - *PERMAS Programmers Manual*
  - *PERMAS Installation Manual*
- the following documents are provided:
- *ABAQUS Door Manual*,
  - *I-DEAS Door Manual*,
  - *MEDINA Door Manual*,
  - *NASTRAN Door Manual*,
  - *PATRAN Door Manual*.

- Clearly arranged layout, complete index and cross references using page numbers (even among distinct documents) are a matter-of-course.

For online documentation and help, all manuals are available as PDF files with cross references.

## Training

INTES provides training courses for all application fields of PERMAS. Based on a three-day introductory course on linear statics several one- or two-day courses are offered on other topics like contact, heat transfer, dynamics, optimization, etc.

The latest workshop program can be downloaded from the INTES homepage.

## Future Developments

In order to provide the FEA user community with

a powerful analysis tool continuously, the PERMAS and VisPER development team is intensively working on the extension of already existing functional modules and on the development of new modules and functionalities.

The main lines of future software development are as follows:

- **Integration with CAD/CAE**  
This includes e.g. improved and simpler model handling or automatic tools for modeling and result evaluation.
- **More complex simulations**  
This mainly includes functional extensions and coupled analysis features.
- **Higher performance**  
Increasing model sizes requires a continuous improvement of the software efficiency. Beside algorithmic improvements, this includes the adaptation of the software to new hardware developments.
- **Advanced visualization**  
For models and results advanced visualization techniques and tools are developed.

Upon your request we are ready to inform you about current development projects and the current planning status for the next major release of PERMAS and VisPER.

---

## Additional Information

For requesting more information and in case of additional questions please contact:

Marketing:	<b>Reinhard Helfrich</b>
Phone:	<b>+49 (0)711 784 99 - 11</b>
Fax:	<b>+49 711 784 99 - 10</b>
E-mail:	<a href="mailto:info@intes.de">info@intes.de</a>
WWW:	<a href="http://www.intes.de">http://www.intes.de</a>
Address:	<b>INTES GmbH Schulze-Delitzsch-Str. 16 D-70565 Stuttgart</b>



## Index

- $\alpha$ -method, 81
- 10% rule, 79
- active control, 35
- active set strategy, 89
- anisotropy, 57
- architecture
  - 64 bit, 101
- Armstrong-Frederik, 73
- assembled situations, 23, 78
- basic variable, 90
- Bayliss-Turkel, 80
- bead design, 84
- beam elements
  - standard cross sections, 56
- BFGS, 89
- BIW, 21
- body-in-white, 21
- bolt pretension, 67
- brake squeal analysis, 28
- buckling
  - linear, 73
  - nonlinear, 71
- Campbell, 31, 75
- car body analysis, 21
- cascade controller, 32
- casting, 71
- Cauchy, 71
- CB condensation, 74
- CCNG analysis, 25, 69, 71
- COF, 62
- CoMAC, 62
- compensation springs, 10, 13, 68
- complex eigenvalues, 75
- component, 49
- composites, 55, 93
- condensation, 49, 74
  - Craig-Bampton, 74
  - dry, 26
  - dynamic, 26, 49, 74, 76, 96
  - generalized modal, 61, 74
  - Mixed-Boundary Craig-Bampton, 74
  - static, 49, 96
- configuration, 49
- constraint
  - kinematic, 53
- contact analysis, 66
- contact locking, 10, 26, 29, 68
- contact status files, 26, 68
- contact wizard, 46
- control elements, 35
- convectivity elements, 82
- Coriolis matrix, 31
- Coulomb, 67
- Coulomb's law, 67
- coupling
  - analyses, 63
  - CFD, 64
- CQC, 79
- Craig-Bampton, 49, 74
  - mixed-boundary, 49, 74
- critical rotation speed, 31
- cutting forces, 21, 63
- cyclic symmetry, 51
- damping, 78
  - frequency-dependent, 77
  - material, 78
  - matrix, 78
  - modal, 78
  - proportional, 78
  - Rayleigh, 78, 81
  - rotation speed dependent, 31, 77
  - structural, 78
  - viscous, 78
- design constraint linking, 12, 13, 84
- DiRect, 89
- Dividing Rectangles, 89
- Drucker-Prager, 71
- dynamic condensation, 81
- earthquake spectral response, 79
- effective masses, 74
- eigenvalues
  - complex, 75
  - real, 73
- electro-statics, 91
- element generator, 14
- element library, 55
- element stress, 57
  - smoothed, 57
- elements
  - axisymmetric, 56
  - beam, 55
  - contact analysis, 55
  - convectivity, 56, 82
  - discrete, 55
  - flange, 55

- fluid, 55
- fluid-structure coupling, 56
- gasket, 25
- geometry, 56
- load, 56, 58
- membrane, 55
- plate, 55
- plot, 56
- radiation boundary condition (RBC), 56
- rod, 55
- scalar, 56
- semi-infinite, 56
- shell, 55
- solid, 55
- surface waves, 56
- engine analysis, 24
- Engquist-Madya, 80
- error estimator, 56
- error indicator, 57
- failure function, 90
- Fast Fourier transformation, 22
- FFT, 22
- fibre-reinforced composites, 93
- fixed mold parting line, 42, 86
- floating license, 101
- fluid elements, 55
- force-guided contact, 10, 68
- FORM/SORM, 90
- Fourier, 58
- free-form optimization, 43
- frequency response, 77
- friction
  - Coulomb, 67
- frozen regions, 86
- function
  - failure, 90
  - limit state, 90
  - mathematical, 58
- gasket elements, 25, 69, 71
- gnuplot, 59, 62
- green, 2
- GUI, 39
- Guyan, 49, 74, 96
- gyroscopic matrix, 31
- gzip, 96
- hardening, 71
  - isotropic, 71
  - kinematic, 71
  - mixed, 71
  - nonlinear kinematic, 72
- heat transfer, 81
- Hilber-Hughes-Taylor, 77, 80
- hole detection, 44
- hull generation, 42, 44, 88
- incompatible meshing, 21
- inertia relief, 21, 58, 66, 71, 74, 84
- interfaces, 60
- interpolation region, 54
- Kirchhoff, 55
- layout optimization, 86
- LDR, 89
- license manager, 101
- limit state function, 37, 90
- line description, 51
- linear buckling, 73
- MAC, 62, 74
- machine tool, 31
- magneto-statics, 91
- manufacturing constraints, 42, 86
- mapping, 14
- material
  - cast-iron, 72
  - Drucker-Prager, 71
  - Mohr-Coulomb, 71
  - properties, 57
  - Tresca, 71
  - viscoplastic, 71
  - von Mises, 71
- matrix
  - Coriolis, 31
  - gyroscopic, 31
- matrix models, 22, 61
- maximum member size, 86
- Maxwell, 91
- MBCB condensation, 74
- MBS, 21
- message tree, 15, 40
- minimum member size, 86
- MLDR, 23, 76
- modal analysis, 73, 82
- modal participation factors, 73, 78
- model updating, 24
- model verification, 58
- Mohr-Coulomb, 71
- Monte Carlo, 90
- morphing, 43

- MPC, 52 to 54, 66, 95
  - general, 54
  - interpolation region, 54
  - linearization area, 54
  - rigid body, 54
- MpCCI, 17, 64, 99
- MS, 89
- Multi-Body Systems, 21
- Multi-Level Dynamic Reduction, 23
- multi-start method, 89
  
- Newmark, 77, 80, 81
  - $\beta$ , 77, 80
- Newton, 89
- Newton-Raphson, 71, 81
- nodelock license, 101
  
- optimization
  - bead, 84
  - contact pressure, 88, 89
  - derivative-free, 88
  - design elements, 56
  - free-form, 12, 84, 85
  - frequency response, 83
  - global, 88
  - layout, 86
  - non-parametric, 12, 85
  - positioning, 84
  - shape, 84
  - size, 83
  - topology, 86
  - trust region, 88
  
- parallelization, 17, 83, 101
- part connections, 26
- part coupling, 52
- PERMAS Tools, 63
- PERMASgraph, 63, 102
- polygon reduction, 42, 88
- post-buckling, 11
- power spectral density, 79
- pre-buckling behavior, 73
- press fit, 54, 67, 69
- pretension wizard, 46
- probability of failure, 90
- Python, 39
  
- quality assurance, 65
  
- radiation, 82
- random response analysis, 79
- Rayleigh, 78, 81
- RBM Assistant, 10, 14, 40, 68
  
- real eigenvalues, 73
- reduction
  - Guyan, 49, 74, 96
- reference system
  - co-rotating, 31
  - inertial, 31
- refinement indicator, 56
- release directions, 86
- reliability analysis, 89
- response
  - in frequency domain, 77
  - in time domain, 76
  - steady-state, 77
  - transient, 78
- response spectrum, 79
- response surface methods, 90
- restart, 63
- results
  - combination, 62
  - comparison, 62
  - transformation, 62
  - xy data, 62
- robust design, 36, 85
- rolling, 71
- rotating systems, 30
- rotor dynamics, 31
- Rutherford-Boeing, 61
  
- sampling, 10, 65
- sandwich shells, 55
- SBV, 43
- SCP, 89
- SDM, 65
- self contact, 10, 66
- sets
  - element sets, 57
  - node sets, 57
- shape basis vectors, 43
- shape optimization, 43, 84
- shift method, 73
- Simulation Data Management, 65
- situation, 50
- sizing, 83
- sizing optimization, 43
- SLP, 89
- smoothing, 42, 88
- spectral response analysis, 79
- spotweld, 53
- SQP, 89
- static mode shapes, 78, 80
- steady state response
  - rotating systems, 31

steady-state response, 22, 31, 77

stiffness

centrifugal, 73

convective, 73

frequency-dependent, 77

geometric, 73

pressure, 73

rotation speed dependent, 31, 77

spotweld, 53

STL, 88

stochastic analysis, 90

submodeling, 25, 49, 68, 72

subspace iteration, 73

substructure technique, 54, 65, 72, 81

substructure wizard, 47

substructuring, 49, 50, 85, 86, 96

summation rules, 79

surface description, 51

system kernel, 65

task scanner, 65

temperature field, 81

temperature mapping, 25

Thomas, 71

tooltips, 40

topology optimization, 86

transfer function, 22

transient response, 77

Tresca, 71

trust region, 88

user defined material, 71

variant, 50

variant analysis, 50

view factors, 83

visco-plasticity, 71

VisPER, 5, 9, 14, 39

volume-shell transition, 54

von Mises, 71

voxel, 45

wizard, 42, 44, 46

contact, 46

fluid-structure coupling, 44

free-form optimization, 43

pretension, 46

shape optimization, 43

sizing optimization, 43

substructure, 47

topology optimization, 42

WLIN, 89

XML, 65

zero force springs, 10, 13, 68





## Contact:

Phone: +49 (0)711 784 99 - 0

Fax: +49 711 784 99 - 10

E-mail: [info@intes.de](mailto:info@intes.de)WWW: <http://www.intes.de>

Address: INTES GmbH

Schulze-Delitzsch-Str. 16

D-70565 Stuttgart